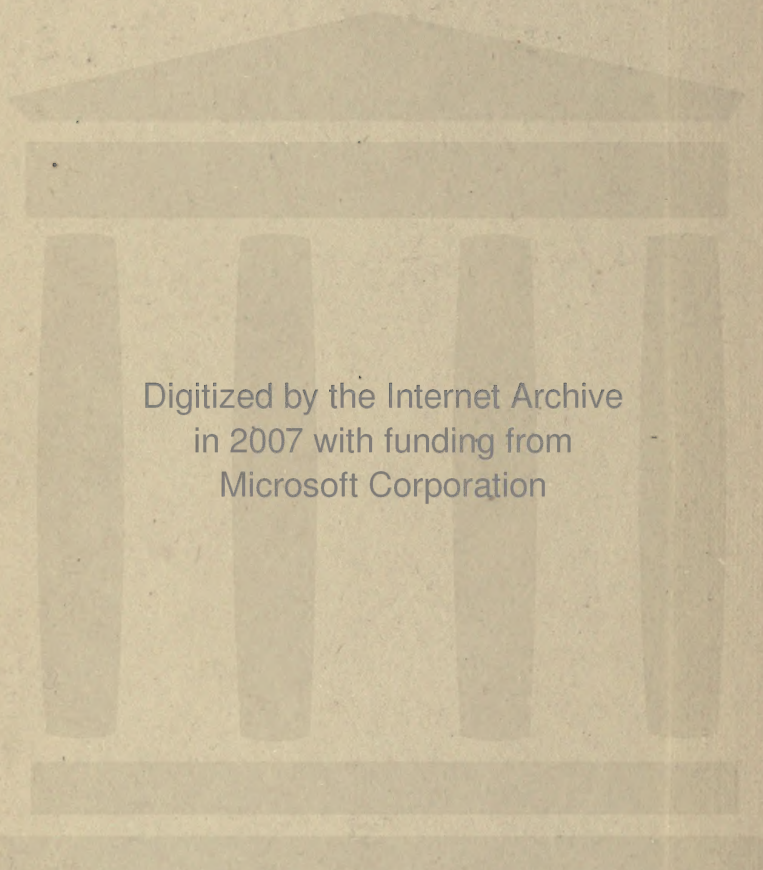




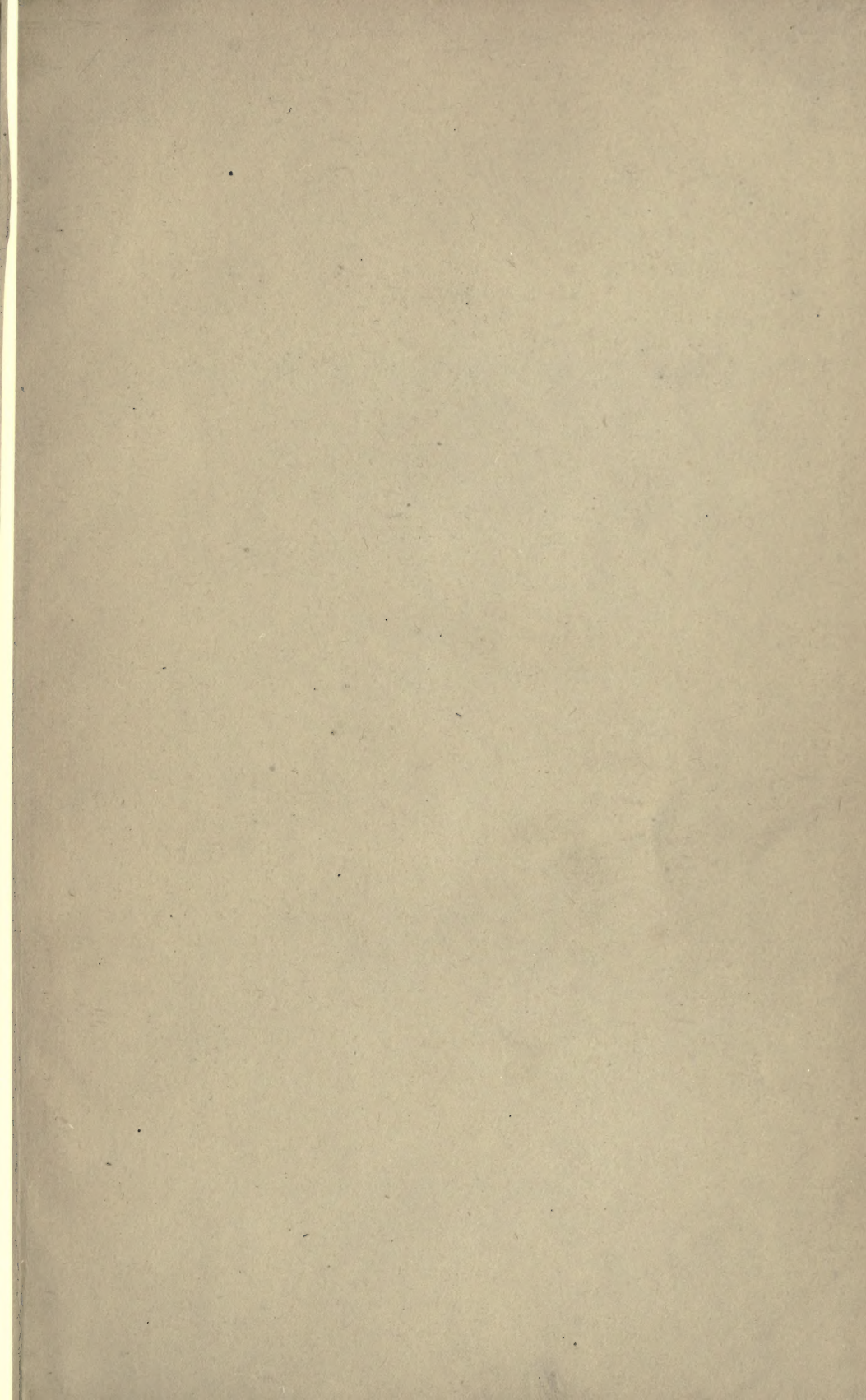
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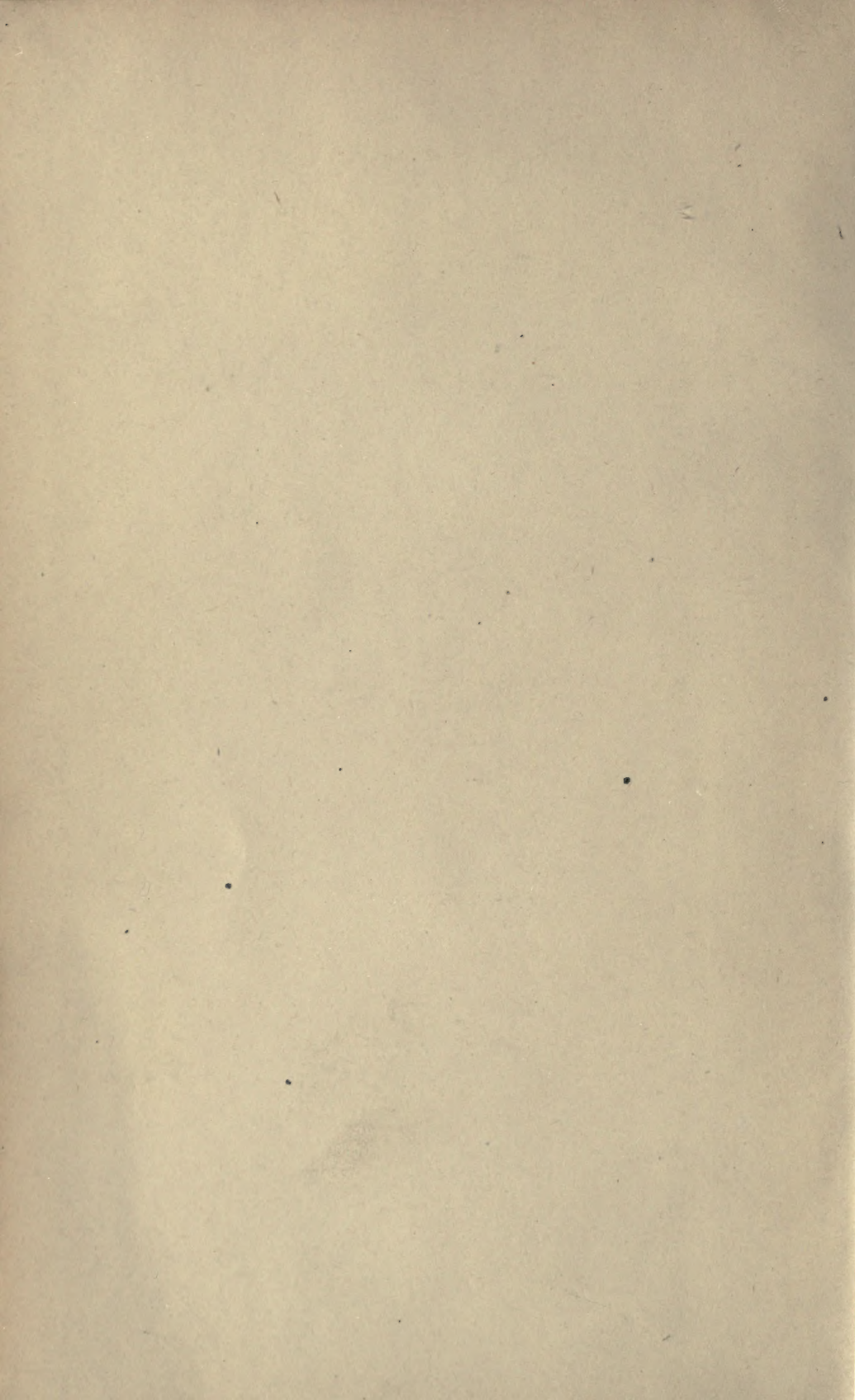
AN INTRODUCTION TO THERMODYNAMICS

MILLS



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AN INTRODUCTION TO THERMODYNAMICS

FOR ENGINEERING STUDENTS

BY

JOHN MILLS

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COLORADO COLLEGE

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PREFACE

This book is the outcome of the writer's desire to put into the hands of a class, studying introductory thermodynamics, in a concise form a statement of those principles and concepts which are essential to the study of the construction and operation of steam engines, steam turbines, and their auxiliaries, and, to a more limited extent, of air compressors and gas engines. The book deals only with the ideal and limiting cases, and aims only at a preparation of the student for reading the more advanced technical works or papers. As such the text makes no claim to originality except, to some extent, in the selection and arrangement of material and in the location of the emphasis.

The material selected is from such sources as the works and books of Clausius, Rankine, Carnot, Kelvin, Stodola, Pellat, Planck, Preston, Edser, Bryan, Boulyin, Swinburne, Peabody, and Buckingham. The arrangement is intended to lead the student from the sophomore physics work into thermodynamics as a continuation and more detailed study of a portion of physical science already somewhat familiar. The emphasis is therefore placed upon the physical concepts, and constant reference is made to the molecular kinetic theory, which is usually familiar to such a student. Although an effort has been made to eliminate as far as practicable the use and solution of differential equations, it is hoped that should the book be used by a student who later takes up the study of pure thermodynamics, he will have little if anything to unlearn and will find the physical interpretation of the mathematical relations rendered easier. The emphasis is also placed upon the solution of numerical problems, many examples of which are given. These in general are arranged to lead from the simpler concepts to the ideal limiting cases of the applications to engineering problems.

In addition to the general statement of sources made above, acknowledgments of details are made in footnotes. It is the writer's pleasure to acknowledge the influence of former study under Professor C. W. Berry of the Massachusetts Institute of Technology. To the inspiration of Professor R. A. Millikan of the University of Chicago, a friend and former teacher, who has read a portion of the proof of this text, a general acknowledgment is gratefully made for most that is good in the writer's point of view and pedagogical method in physics.

JOHN MILLS

COLORADO SPRINGS, COLORADO

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SYMBOLS

The symbols given below are used consistently throughout this text. They have been selected, as far as is possible, in accordance with the following rules: (1) a given symbol is to be used for but one physical magnitude, (2) a lower-case letter is always to represent the value per unit mass of the magnitude to which it refers, (3) the symbols are to be in accordance with the American usage.

$A = 1/J$ = factor for reducing mechanical units of energy to heat units.	λ = total heat contents per unit mass of a mixture (from 32° F.), <i>e.g.</i> $\lambda = q + r + h$, or $\lambda = q + xr$, or $\lambda = q + xp + xApu$.
a = general expression for area.	M = mass (or weight) per second in flow of fluids.
α = volume coefficient of a gas $= \frac{1}{V} \left(\frac{dV}{dt} \right)_p$	m = mass (or weight).
Apu = external latent heat of vaporization in thermal units.	n = general expression for the exponent in the relation $pv^n = \text{constant}$.
<i>B.t.u.</i> = British thermal unit.	p = general expression for pressure (<i>i.e.</i> specific pressure).
β = pressure coefficient of a gas $= \frac{1}{p} \left(\frac{dp}{dt} \right)_v$	ϕ = general expression for entropy.
c = general expression for specific heat.	ϕ_w = general expression for entropy of liquid (per unit mass) above that of water at 32° F.
c_p = specific heat at constant pressure.	ϕ_v = increase in entropy per unit mass due to vaporization.
c_v = specific heat at constant volume.	ϕ_s = increase in entropy per unit mass due to superheating.
E = total intrinsic energy of a substance.	q = heat contents of water per unit mass measured from 32° F.
ϵ = intrinsic energy of a unit mass of a substance.	Ω = general expression for heat transferred to or from an external source.
η = thermal efficiency of a process or cycle.	q = general expression for heat transferred per unit mass.
F = general expression for force.	R = constant in the gas relation $pV = mR\theta$
g = numerical value of the acceleration due to gravity.	r = total latent heat of vaporization per unit mass.
γ = general expression for density.	ρ = internal latent heat of vaporization per unit mass.
h = heat units required to superheat from dry saturated steam, per unit mass.	S = intrinsic energy due to molecular motion.
I = intrinsic energy due to molecular configuration.	s = specific volume of dry saturated steam.
J = factor for reducing thermal units to work units.	σ = specific volume of water = 0.016 cubic foot per pound.
κ = ratio of c_p to c_v .	
l = loss of kinetic energy due to friction, per unit mass of steam.	

- T = temperature measured on the thermodynamic scale.
 t = general expression for temperature.
 τ = general expression for time.
 θ = absolute temperature as measured on a perfect gas scale.
 t_0 = saturation temperature of dry steam before superheating.
 θ_0 = absolute saturation temperature of dry steam before superheating.
 V = general expression for volume.
 v = general expression for the volume of a mixture per unit mass.
 \mathbf{v} = velocity.
- u = increase in specific volume due to vaporization, *i.e.* $u = s - \sigma$.
 W = general expression for external mechanical work.
 w = general expression for external work per unit mass of working substance.
 x = quality of a mixture of x parts steam and $(1 - x)$ parts water.
 y = friction work during flow through a channel per unit mass of the fluid.
 z and dz = distance through which a force acts, *e.g.* $W = \int Fdz$.

AN INTRODUCTION TO THERMODYNAMICS

CHAPTER I

FUNDAMENTAL CONCEPTS AND LAWS

1. Work. Work is defined as the act of overcoming a resistance through space. It is measured by the product of the component of the acting force along the direction of the motion of its point of application, and the distance through which its point of application moves during its action. The scholium to Newton's Third Law of Motion states that "if the action (activity) of an agent be measured by its amount and its velocity conjointly, and if similarly the reaction (counter-activity) of the resistance be measured by the velocities of its several parts and their amounts conjointly, whether these arise from friction, molecular force, weight, or acceleration, action and reaction in all combinations of machines will be equal and opposite." Since velocity is merely the rate at which space is traversed, it is seen that this scholium states that during any given time the work of the acting forces is equal to the work of the resisting forces.

The unit of work depends upon the chosen unit of force. In the absolute C. G. S. system where the unit of force is the dyne (a force which if acting constantly upon one gram will produce a change in its velocity of one centimeter per second every second), the unit of work is the erg, or the work done by one dyne acting through one centimeter. In the gravitational C. G. S. system the unit of force is the gram weight (*i.e.* 980 dynes), and the unit of work is the gram-centimeter. The force unit in the F. P. S. system (gravitational) is the pound, or the force of the earth's attraction upon one pound of matter, and the unit of work is the foot pound.

2. Energy. Energy is defined as the capacity of a body for doing work. It is evident, then, that work and energy are to be measured in the same units. In general the energy possessed by a body or by a system of bodies may be considered as the result of work done upon that body or system by another body or system. The energy of a system may be either its capacity to do work as the result of

work done upon the system in giving to its component parts motion (that is, work done against the resistance of acceleration), in which case it is known as kinetic energy; or, it may be the result of work done upon the system in producing the existing configuration of its component parts (that is, work done against gravitation or other forces), in which case it is known as potential energy.

The sum of the potential and kinetic energy of a body is commonly called its intrinsic energy. The Principle of the Conservation of Energy states that if work is done upon a body the amount of work is equal to the increase in intrinsic energy of the body; that is, to the sum of the increases in the kinetic and potential energy. And conversely, the work done by a body is equal to the decrease in its intrinsic energy. Thus, if dE represents the increase in the intrinsic energy of a body and dW the amount of work done by the body, then $dE + dW = 0$.

The measure of the kinetic energy of a rigid body is the sum of its kinetic energy of translation and its kinetic energy of rotation. The kinetic energy of translation is $\frac{1}{2}mv_0^2$, where m is the mass of the body and v_0 is its absolute velocity of translation. The kinetic energy of rotation is $\frac{1}{2}K\omega_0^2$, where K is the moment of inertia of the body about an axis through its center of mass, and ω_0 is the absolute velocity of rotation about this axis. The potential energy of a body is $\int Fdz$, where the integral represents the work done against the resisting forces in moving the body from a position of zero potential energy to the position under consideration.

From the above statement of the measure of the intrinsic energy of a body it is evident that its energy is the total amount of work which could be derived from it under the most favorable theoretical considerations. For the practical purposes of the universe in which we live the intrinsic energy may be considered in two parts; namely, available and unavailable energy. The availability of the intrinsic energy of a body depends upon external conditions, as will be seen later in various problems. A simple mechanical illustration is as follows. Consider the velocity of a cannon ball of mass m to be v_0^2 , then its energy is $\frac{1}{2}mv_0^2$. If the velocity of the ball after it has struck a ship initially at rest (that is, the velocity communicated to the ship by the impact) be represented by v , then the intrinsic energy of the ball is now $\frac{1}{2}mv^2$. Of the total energy of the ball only the portion $\frac{1}{2}m(v_0^2 - v^2)$ was available for the destruction of the ship.

3. Molecular Kinetic Theory. The molecular kinetic theory is based upon certain assumptions as to the character and actions of molecules which do not pretend to represent the actual character of the molecules but which simplify the problem so that it admits of a mathematical analysis. The molecules of a gas (for which state many of the assumptions would be approximately correct) are considered to be in motion, to act like hard, smooth, small, and perfectly elastic spheres, exerting no forces upon each other except at the instant of collision, and to exist in almost immeasurable numbers in any finite volume of the gas. These assumptions admit of the application of the laws of probability, and of the treatment of collisions as if the molecules were uniform spheres of unit coefficient of restitution, colliding centrally.

Upon these assumptions by a mathematical analysis Maxwell, Clausius, Joule, Boltzmann, and others have developed conclusions¹ as to the behavior of a gas which agree with the observed facts as expressed in the well-known laws of Boyle, Charles, Dalton (and Avogadro).

As expressed by Risteen, one of these conclusions is as follows: "In a molecular mixture there is one physical quality which is the same for every set of molecules; and that is, the *average kinetic energy of translation per molecule*." Now it is an observed fact that whenever two or more substances are brought into intimate contact there is one physical property which is the same for all of the substances, namely, the temperature. Hence, it is natural to assume that equality in the temperature means equality in energy of translation of the molecules.

4. Temperature. The temperature of a body is merely an arbitrary expression of its "hotness" as compared with some states of hotness assumed as standard. Because, however, of the inexactness of the sense of touch as a measure of the "hotness" and because of the experimentally determined fact that some substances expand as they become hotter, the expansion or contraction of a given amount of such a substance is arbitrarily taken as the quantitative measure of a temperature change. For ordinary purposes the substance used is mercury enclosed in glass, and the relative expansion of the mercury is taken as a measure of the change in temperature.

¹ It is desirable at this point to give merely the briefest possible résumé of the kinetic theory (the facts of which are already familiar to the student from his course in general physics), in order that the concept of temperature may be approached from the side of energy. For a more complete statement in non-mathematical language see Risteen, "Molecules and the Molecular Theory," pp. 42-43.

The scale upon which temperature is measured has been chosen in a purely arbitrary manner. On the Centigrade scale one degree¹ of temperature represents a change in temperature equal to that which would produce one one-hundredth of the effect produced in the mercury thermometer by a difference in temperature corresponding to the following easily reproducible states of pure water. The lower point, called zero degrees, is the temperature of pure water and melting ice in an intimate mixture. The upper point, called one hundred degrees, is the temperature of the vapor of pure boiling water in contact with its liquid. In both cases (although it is of primary importance only in the second case) the pressure upon the mixture must be what is known as standard or normal pressure, which is equal to that exerted by a column of mercury 76 centimeters high, when the mercury is at the temperature of melting ice and water. The scale is extended in each direction from these two fixed points by divisions corresponding to a degree as defined above.

The Fahrenheit scale has 180 degrees which are equivalent to 100 degrees on the Centigrade scale. The fixed point corresponding to the temperature of melting ice and water is marked 32 degrees, and the boiling point is 212 degrees.

According to the molecular kinetic theory two bodies are at the same temperature when the average kinetic energy of translation of their molecules is the same. It will be shown later in Section 15 that according to this theory one degree Centigrade is a difference in temperature such that the average kinetic energy of translation of the molecules of the substance considered, is increased (or diminished) by one one-hundredth of the change in this kinetic energy which would be produced by a difference in temperature corresponding to that between zero and one hundred degrees Centigrade. It will also be seen that this is, for a gaseous body, approximately equal to $\frac{1}{2} \frac{1}{\gamma}$ of the average kinetic energy of the molecules when the substance is at zero degrees Centigrade.

5. Thermal Units. When two bodies at different temperatures are placed in proximity, it is found that they ultimately come to a common temperature (unless this difference in temperature is maintained by some external agency). According to the kinetic theory this must mean that a transfer of energy has taken place from one body to the other in order that the final average molecular energy of translation of the two bodies shall be the same. Energy trans-

¹ For more rigorous definition of a degree Centigrade see Section 14.

ferred under these conditions is known as heat. Because it was not recognized in the earlier development of physical science that heat is energy, units were introduced for the measurement of heat.

These thermal units are essentially arbitrary and relative, since they depend upon the unit of temperature difference. Thus the heat (energy) required to change the temperature of one gram of pure water from 15° to 16° Centigrade is the thermal unit in the C. G. S. system and is called the calorie. And similarly the amount of heat (energy) required to raise the temperature of one pound of water from 62° to 63° Fahrenheit is called a British thermal unit, or as abbreviated a B. t. u.

6. First Law of Thermodynamics. The Principle of the Conservation of Energy was not established until the classical work of Joule (and Mayer) performed during the years following 1843. This principle may be stated in a more general form than that given in Section 2 as follows: "Every physical or chemical change of state has a fixed mechanical equivalent expressible in work units." Thus, for example, the amount of heat (or energy) required to produce in water the change in physical state which is produced by a calorie (or a B. t. u.) is to be equated under all conditions to a definite number of work units. This particular application of the Principle of the Conservation of Energy was experimentally shown to be correct by Joule and was really the basis for the adoption of the principle in its broadest form. This application has received the name of the First Law of Thermodynamics. As usually stated it reads: "Heat and mechanical work are mutually convertible,¹ and heat requires for its production and produces by its disappearance a certain definite amount of work for each thermal unit." The value of this mechanical equivalent of heat from the determination made by Rowlands is 778 foot pounds per B. t. u., or 4.19×10^7 ergs per calorie. In symbols this law may be written

$$W = J\Omega \text{ or } \Omega = AW, \quad (1)$$

where Ω is the number of thermal units, W the number of mechanical units which correspond, and J is the (mechanical) equivalent of each thermal unit in work units for the system in which W is to be expressed. The factor A is evidently the reciprocal of J .

7. Specific Heat. The amount of heat required to produce a change of one degree in the temperature of a substance depends

¹ Except in so far as all the heat energy of a body is not available for such a conversion, as is discussed in Section 2 and also in later sections.

upon the nature of the substance, the mass in which it is desired to produce this change, and, as will be seen in Section 18, upon the conditions under which the change is produced. For the same mass of a given substance it is found that the amount of heat required per degree is different at different temperatures; that is, it is a function of the temperature. The thermal capacity per unit mass or "specific heat" of a substance is defined as the number of units of heat required to change the temperature of unit mass of the substance one degree. This is expressed either in calories per gram or in B. t. u. per pound, and the numbers representing the value of the specific heat of a given substance are obviously the same in the two systems. But it would be better to express the specific heat directly in mechanical units, since these belong to an absolute system and since this may readily be done in accordance with the First Law of Thermodynamics. Thus J , the mechanical equivalent of heat, is merely the specific heat of water in absolute units.

Since the specific heat of a substance is slightly different at different temperatures, it may best be defined by a differential notation. Thus if a quantity of heat dq be added to a unit mass of a substance and produce an increase in the temperature of dt , then the specific heat c , at the temperature t , at which the addition was made, or the heat required per degree per unit mass, is $\frac{dq}{dt} = c$. Hence the total heat Ω required to produce in a mass m a change of temperature from t_1 to t_2 is

$$\Omega = m \int_{t_1}^{t_2} c dt. \quad . \quad . \quad . \quad . \quad . \quad . \quad (2)$$

Frequently, it is sufficiently accurate to assume the specific heat constant through the range of temperature t_1 to t_2 and to write

$$\Omega = mc(t_2 - t_1), \quad . \quad . \quad . \quad . \quad . \quad . \quad (3)$$

where c is the average value of the specific heat of the substance for a range of temperature which includes t_1 and t_2 .

8. Component Parts of Specific Heat. When the temperature of any substance is changed by the addition of a quantity of heat, it is possible to consider the change as composed of three parts, and it is frequently possible to assign to each part its proper portion of the heat energy transferred. In general, heat added to a substance may be considered as producing first, a change in the temperature; second, a change in the configuration of the molecules composing the substance; and third, because of the consequent increase (or decrease)

of the volume occupied by the substance, a certain amount of external work. The first portion, which goes to produce a change in the temperature, increases according to the molecular theory the average kinetic energy of translation of the molecules.¹ This portion will be represented by the symbol AdS . The second portion of the heat supplied produces a change in the potential energy of the molecules and will be represented by dI . The third portion, representing an amount of external work, will be symbolized by dW . In general, then, if a small amount of heat dQ is required to produce a change in the temperature of a substance, we may write

$$dQ = A(dS + dI + dW), \dots \dots \dots (4)$$

where dS , dI , and dW are all expressed in mechanical units.

In the case of the change of state from water to steam, to be discussed later, the term dI is large as compared with the other two terms. In the case of a "perfect gas," also to be discussed later, the term dI is zero for all changes.

9. Latent Heat. In the case of a change of state, as that referred to above, where water is converted into a vapor, or *vice versa* (or in the case of the formation of ice from water, or *vice versa*) there is produced no change in temperature until sufficient heat energy has been added, or subtracted, to produce the change in potential energy for the entire mass corresponding to the different configuration of the molecules. During such a change of state the term AdS is then zero.

The term $\frac{AdI}{m}$, representing the amount of heat required per unit mass to produce the change in internal potential energy, is known as the *internal* latent heat for that particular change of state. This internal latent heat is also a function of the temperature at which the change of state occurs, as will be discussed more fully later, in the case of steam. As commonly expressed the latent heat is the sum of the terms $\frac{AdI}{m}$ and $\frac{AdW}{m}$, the latter representing the *external* latent heat or external work which must be done in producing the change.

Thus in C. G. S. units, the latent heat of vaporization of water at 100° Centigrade and under an atmospheric pressure of 76 centi-

¹It is true also that an increase in temperature means in general an increase in the kinetic energy of rotation of the molecule and also an increase in the "sub-molecular" kinetic and potential energy of the component parts of the molecule, but for the further purposes of this discussion all this energy will be considered as included by the term AdS .

meters of mercury is 536.5 calories. Of this the part corresponding to $\frac{AdW}{m}$ may be found as follows. The volume occupied by one gram of water changes from one cubic centimeter to 1649 cubic centimeters, which is the volume occupied by one gram of water vapor under the given condition of temperature. Since this change takes place against a pressure of (76)(13.6)(980) dynes per square centimeter, (76)(13.6)(980)(1649 - 1), or 1.668×10^9 , ergs represent the external work during evaporation. That is, the external latent heat of evaporation is $1.668 \times 10^9 \div 4.19 \times 10^7 = 40$ calories for the conditions of the problem. The internal latent heat, representing the gain in molecular potential energy, is then 496.5 calories or 2.08×10^{10} ergs.

10. Boyle's Law. Boyle's Law states that, other things being equal (that is, the temperature, mass, and molecular constitution of the gas experimented upon remaining unchanged), the product of the pressure at which a given gas is maintained and the volume which it occupies is a constant. Or, in other words, if the pressure is changed, the volume occupied by a given quantity of gas maintained at a constant temperature undergoes a change inversely proportional to the change in pressure. In symbols this is expressed as $pV = \text{a constant}$, or

$$p_1 V_1 = p_2 V_2 = p_3 V_3, \quad \dots \quad (5)$$

where p represents the pressure or force per unit area and V represents the volume (usually measured in the same system of units as is the area).

The value of the constant is evidently for a given mass of some particular gas a function of the temperature. This follows at once from the observable fact that if the temperature is changed, the value of the product pV is also changed. The physical meaning of this fact is evident from the following analysis based upon the assumptions of the kinetic theory.

The pressure upon the bounding surface of a body of gas is due to the impacts of the moving molecules. It is proportional to the number of these impacts per second upon each unit area of the bounding surface, and to the impulse of each molecular impact. The impulse is proportional to the change in momentum of the moving molecule, and this in turn to the velocity and mass of the molecule. The total number of impacts made by a given molecule will be proportional to its velocity. The number of molecular impacts upon each unit of area of the bounding surface will be proportional

to the gas density; that is, for a given mass of gas, inversely proportional to the entire volume. Hence the pressure is directly proportional to the mass of a molecule and to the square of the average molecular velocity, and inversely proportional to the entire volume. That is, the product of the pressure and the volume is directly proportional to the average molecular kinetic energy of translation, and hence is some function of the temperature.¹

This law of Boyle is not rigorously correct, for all gases show departures from it at high pressures. It is followed very closely for pressures under 10 atmospheres. For higher pressures the departures are more marked. Thus for air they range from about $\frac{1}{4}$ per cent at 10 atmospheres to as much as 25 per cent at 600 atmospheres.

11. Joule's Law. Another important law of gases is due to experiments performed by Joule. This law states that "all the work done in compressing a gas is conducted away as heat if the temperature of the gas is kept constant, and conversely, when a gas expands at constant temperature, it must receive from its surroundings a quantity of heat equal to the work done by the expansion." This means that during the expansion of a gas no internal work is done by the molecules either by molecular repulsions or against molecular attractions. For if there are attractive forces between the molecules, then in such an expansion, since the molecules become more widely separated, work must be done to produce this separation, and this must be supplied by the partial conversion of molecular kinetic energy into molecular potential energy. If there is to be no consequent lowering in the temperature, there must be supplied to the molecules of the substance from the outside an amount of heat energy equal to that converted into molecular potential energy. The amount of heat supplied to an expanding gas would then be greater than the amount of external work performed by the gas during expansion by this amount of increase in the molecular potential energy. According to Joule's Law there are then no molecular attractive forces acting between molecules. On the other hand, if repellent forces exist between the molecules, the potential energy term will decrease in value, and there will be a larger amount of external work done than there is heat supplied to the expanding gas. Thus, the possibility of repellent forces between the molecules of a gas is also denied by Joule's Law. In so far, then, as a

¹ See any general physics text, or Millikan, "Mechanics, Molecular Physics, and Heat," Chapters XIV and XVI.

gas obeys Joule's Law the potential energy term dI in the equation $d\Omega = A(dS + dI + dW)$ must be zero. This law is not more rigorous than Boyle's Law, and the departures from Boyle's Law may be explained in part upon the assumption that there are molecular attractions which are not entirely negligible when the density is high and the molecules close together.

Joule's classical experiment was performed with apparatus shown diagrammatically in Figure 1. Two similar metal vessels *A* and *B*,

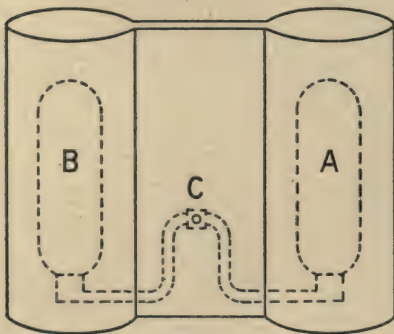


FIG. 1

connecting by a tube fitted with a stopcock *C*, were enclosed in a large calorimeter. The temperature was measured by a thermometer reading to $0^{\circ}.005$ Fahrenheit.¹ The vessel *A* contained air under a pressure of about 22 atmospheres, and the vessel *B* was practically a vacuum. When the stopcock was opened, the air in *A* expanded into *B* without doing external work. It was found im-

possible to detect any change in the temperature of the water as a result of this expansion. From this experiment Joule concluded that "no change in temperature occurs when air is allowed to expand in such a manner as not to develop mechanical work." From this there follows the conclusions stated above as to the intrinsic potential energy of the molecules of a gas.

12. Perfect Gases. Since all gases show departures from the approximate laws of Boyle and Joule, but since these departures are not in general large, it is convenient for the purposes of analysis in problems dealing with gases to treat of purely fictitious so-called "perfect gases." A perfect gas is defined as one for which the laws of Boyle and Joule hold rigorously for all pressures and temperatures. Such a gas would then have no interacting forces between its molecules except at the instant of mutual impact, and according

¹ The more accurate "porous plug experiment" of Kelvin and Joule is described later (see Section 35), but merely as an illustration of an entropy change in an irreversible adiabatic expansion. The conclusions of that experiment which bear upon this discussion are here stated. At ordinary temperatures there is a small but appreciable attraction between the molecules of a gas, and hence a cooling by free expansion through an orifice. The cooling is roughly $0^{\circ}.25$ Centigrade per atmosphere of pressure difference on the two sides of the orifice. The value is less for higher temperatures, and greater for lower temperatures. This phenomenon, successively caused to take place in the same mass of gas, is of course the basis of liquid air machines.

to the kinetic theory (see Section 10, above) the product of the pressure and the volume would be some function of the temperature.

Writing $pV=f(t)$, where $f(t)$ represents some function of the temperature, it is possible to show that for a perfect gas the values of the "pressure coefficient" and of the "volume coefficient" are equal when determined for the same temperature. The pressure coefficient β is defined as the increase in pressure per degree of temperature per unit of pressure. In symbols this is expressed as

$$\beta = \frac{p_1 - p_0}{p_0(t_1 - t_0)},$$

where p_0 is the original pressure exerted by a given mass of gas at a temperature of t_0 , and p_1 and t_1 represent the pressure and temperature of some later condition of the gas while occupying the same volume. In differential notation this is most conveniently expressed by using the symbol $()_v$ to indicate the fact that the quantity enclosed by the parenthesis is found under the conditions of constant volume. Thus

$$\beta = \frac{1}{p} \left(\frac{dp}{dt} \right)_v \dots \dots \dots (6)$$

Similarly, the expansion coefficient α is

$$\alpha = \frac{1}{V} \left(\frac{dV}{dt} \right)_p \dots \dots \dots (7)$$

From the expression of Boyle's Law for a perfect gas as $pV=f(t)$ we have

$$\begin{aligned} \beta &= \frac{1}{p} \left(\frac{df(t)}{Vdt} \right)_v = \frac{f'(t)}{pV} = \frac{f'(t)}{f(t)}, \\ \alpha &= \frac{1}{V} \left(\frac{dV}{dt} \right)_p = \frac{f'(t)}{pV} = \frac{f'(t)}{f(t)}. \end{aligned}$$

The values of α and of β are evidently equal for the same values of the temperature t .

The fact that all gases have the same value of α for any given temperature was discovered as an experimental fact by Charles in 1787, some sixty years before the announcement of Joule's Law, which has been used here in the definition of a perfect gas.

13. Charles's Law. Charles's Law states that all gases show the same fractional or per cent expansions between the limits of two fixed temperatures. Later experiments have shown that this law of Charles (or of Gay-Lussac, as it is sometimes known) is only

approximate. Experiments have also shown that for all gases the values of the pressure coefficient β for the same initial temperature are practically equal, and are also practically equivalent in value to the values of the expansion coefficient α for the same temperature. This is evident from Table 1,¹ where are given for various gases values of α and β which represent their average values for the range of temperature from 0° to 100° Centigrade.

The definition of a perfect gas leads then to the conclusion that for perfect gases Charles's Law holds rigorously, and also the values of α and β are identical. For a perfect gas the value of β will be taken as 0.0036625 at 0° Centigrade, as is explained in Section 15.

14. Standard Hydrogen Thermometer. Because of the fact that the expansion of most solids and liquids is not a linear function of the temperature, while that of a gas is very nearly a linear function in accordance with Charles's Law, the standard thermometric substance is a gas. A gas thermometer might, obviously, with almost equal justification, be constructed to measure temperature by the expansion of the gas under the condition of constant pressure, or by the change in pressure under the condition of constant volume. Because of the experimental difficulty of making the corrections which enter into a determination of temperature with a constant pressure gas thermometer (due to the variations in the exposed length of the column of gas contained in the stem of the thermometer), Regnault adopted the constant volume form. In 1887 the International Committee of Weights and Measures adopted the *constant volume hydrogen thermometer* as the standard. According to this agreement one degree Centigrade is such a temperature as will cause a change in the pressure of a mass of hydrogen (kept at a constant volume) of 0.0036625, or $\frac{1}{273}$, of the pressure which this hydrogen exerts at the temperature of melting ice and pure water.

15. Absolute Temperature. According to the discussion of Section 10 for a constant volume perfect gas thermometer we may write the pressure as proportional to the average molecular kinetic energy of translation. If, further, the pressure coefficient β be taken for a perfect gas as equal to the value 0.0036625, which is given by Chappuis for hydrogen, a gas which most nearly obeys the laws of perfect gases, it may be shown, as follows, that one degree Centigrade change in the temperature of a perfect gas corresponds to a change in the average molecular kinetic energy of $\frac{1}{273}$ part of the

¹ See page 130.

value at zero degrees Centigrade of this average molecular kinetic energy. Thus

$$pV \propto \text{av. K. E.}$$

Hence

$$\left(\frac{dp}{dt}\right)_v \propto \left(\frac{d(\text{av. K. E.})}{dt}\right)_v,$$

and

$$\alpha = \beta = \frac{1}{p} \left(\frac{dp}{dt}\right)_v = \frac{1}{\text{av. K. E.}} \left(\frac{d(\text{av. K. E.})}{dt}\right)_v,$$

or

$$\frac{1}{273} = \frac{1}{\text{av. K. E. at } 0^\circ \text{ C.}} \left(\frac{d(\text{av. K. E.})}{dt}\right)_{v \text{ at } 0^\circ \text{ C.}} \quad \dots \quad (8)$$

where the symbol " $()_v$ at 0° C. " is used to represent the fact that the volume is maintained constant at the value which it has for a temperature of zero degrees Centigrade.

If, then, the temperature of a perfect gas be lowered to -273° Centigrade, it follows that the molecules would be completely deprived of kinetic energy of translation, and hence at rest. The pressure exerted by the gas would then be zero. This temperature of -273° Centigrade is known as the absolute zero; for, if temperatures are measured in degrees absolute, that is in degrees from this absolute zero, the intrinsic energy of a perfect gas is directly proportional to its temperature. For all practical problems the intrinsic energy of the so-called permanent gases may also be taken to be proportional to the absolute temperature.

This absolute zero is not an attainable temperature. All known gases liquefy and even solidify above this temperature. Thus hydrogen liquefies at about -252° C. and solidifies at about -256° C. By evaporating liquid helium a temperature of about -270.5° C. has been attained. These low temperatures are measured with a platinum resistance thermometer which has been found to give results very close to that of a hydrogen thermometer.

For a perfect gas, and for most practical problems in gases, under the conditions of constant volume the pressure may be taken proportional to the absolute temperature. This is expressed in symbols in equation (9). And similarly, for constant pressure, the volume may be taken proportional to the absolute temperature, as is expressed in symbols in equation (10)

$$\left(\frac{p_2}{p_1}\right)_v = \left(\frac{\theta_2}{\theta_1}\right)_v, \quad \dots \quad (9)$$

$$\left(\frac{V_2}{V_1}\right)_p = \left(\frac{\theta_2}{\theta_1}\right)_p, \quad \dots \quad (10)$$

where θ is the absolute temperature as measured on the perfect gas scale. Or, in symbols,

$$\theta_0 = 273. \quad \theta_1 = t_1 + 273 \text{ in degrees Centigrade.} \quad (11)$$

And since 273° Centigrade equals 491.5 Fahrenheit,

$$\theta_0 = 459.5. \quad \theta_1 = t_1 + 459.5 \text{ in degrees Fahrenheit.} \quad (12)$$

16. General Equation for a Perfect Gas. It is evident since

$$pV \propto \text{average molecular K. E.} \propto \theta,$$

that a general expression for the pressure, volume, and absolute temperature of a perfect gas may be written as follows:

$$pV = mR\theta, \quad (13)$$

where p is pressure, V is volume, θ is the absolute temperature, m is the mass of the gas, and R is a constant. The value of R depends upon the kind of gas and has a numerical value which represents in the system of units to which m belongs the value of the expression $\frac{pv}{\theta}$ for unit mass of the gas for any condition for which the corresponding values of p , v , and θ are known.

Thus for air, assuming that it will follow the equation for a perfect gas, and finding R for the C. G. S. system, we have the density of air at the standard conditions of 76 centimeters of mercury pressure and zero degrees Centigrade equal to 0.001293 as given by Regnault for latitude 45° (where g is 980.6). The volume of one gram of air for these conditions is then 773.4 cubic centimeters. The value of p is 1.0134×10^6 dynes per square centimeter. The absolute temperature, θ , is 273. Hence R is 2.867×10^6 .

Similarly for the F. P. S. system, the volume of one pound of air is 12.39 cubic feet at a pressure of 14.7 pounds per square inch and at a temperature of 32° Fahrenheit. Expressing the pressure p as 144×14.7 pounds per square foot and substituting, gives R equal to 53.35.

17. Use of the Term "Specific." As used in physics the meaning of the term "specific" may best be shown by illustration. Thus "specific heat" is the amount of heat required to produce a definite result, namely, unit change of temperature in unit mass. The term "specific" then indicates on the part of the physical magnitude to which it is applied a definite and peculiar quality in which this particular value of the magnitude differs from all other possible values of the same magnitude. In general it limits the magnitude to unit

mass, unit length, unit area, unit volume, or to a unit change in some related magnitude. Thus "specific resistance" is the resistance per unit cube, or the resistance per unit length having unit cross section, as in the case of the mil-foot.

The adjective "specific" is somewhat more widely used in engineering, where the terminology is not otherwise as carefully differentiated as in pure physics. Thus in the rigorous physical usage "pressure" always means "force per unit area," but in engineering usage, broadened and rendered less precise by the influence of the necessity of conveying ideas in popular language, pressure and force are used almost interchangeably. For this reason there has come into common engineering use the term "specific pressure" to mean force per unit area, as for example in the F. P. S. system, where specific pressure is the force in pounds per square foot. Similarly, "density," which to the student of physics means "mass per unit volume," is expressed by the engineer as "specific weight," meaning thereby the weight per cubic foot. "Specific volume" is the volume of one pound in cubic feet. "Specific gravity" is the weight of a given volume of a substance in terms of the weight of the same volume of water, taken as unity.¹

For the purposes of this book the word "pressure" will always be used in its rigorous sense of force per unit area. Also, the following scheme of notation has been adopted to avoid confusion. Whenever volume, quantity of heat, or any other unit which may later be introduced refers to unit mass of the substance under consideration, a lower case letter will be used. Whenever reference is made to the total value of a magnitude, referring to some other mass than unity, a capital letter will be used. It will be noticed that this has been done consistently in the earlier portion of the text; thus, $pV = mR\theta$, but $pV = R\theta$.

18. Specific Heat of Gases. The specific heat of a gas may be determined under the condition either of constant volume or of constant pressure. It is evident from a consideration of equation (4) of Section 8 that when the volume is constant, since the term dW is zero (because there is no external work done in expanding),

¹ This inexactitude of common engineering terminology has frequent illustration. Thus "speed" and "velocity" are used interchangeably. "Acceleration," which is the rate of change of velocity, is popularized into "rate of acceleration," an expression which rigorously is a second derivative of velocity instead of the first. "Electromotive force" and "potential difference" are frequently used as synonymous. Also "energy" and "power" are not carefully distinguished, as is illustrated by the frequent use of expressions like "expenditure of power."

the amount of heat required to produce in unit mass a temperature change of one degree is less than that required to produce this change when the pressure is kept constant and the term dW has a numerical value (representing the work done externally by the expansion of the gas against the constant pressure to which it is subjected). These two values of the specific heat of a gas will in the future be represented by c_p and c_v for the conditions of constant pressure and constant volume, respectively. Or, in symbols,

$$c_p = \left(\frac{dq}{dt} \right)_p \text{ and } c_v = \left(\frac{dq}{dt} \right)_v \quad . \quad . \quad . \quad . \quad (14)$$

The ratio of the specific heats of a gas (written $\kappa = \frac{c_p}{c_v}$) may be shown to depend upon the atomic structure¹ of the gas considered. For monatomic gases this ratio κ has a value of 1.66, for diatomic gases a value of about 1.40; for the diatomic mixture constituting air the value is 1.401, for carbon dioxide it is 1.30 (at 30° C.).

19. Graphical Representation of Gas Transformations. Changes in the values of the magnitudes p , V , and θ for a given mass of gas may best be represented graphically by plotting on a set of coördinate axes values of V as abscissas and values of p as ordinates. Since the temperature θ is a function of p and V , it is implicitly represented upon the same plot.

The possible ways in which a given mass of gas may undergo a continuous transformation, not involving any chemical change, are evidently the following: (a) at constant volume by changing the pressure indirectly through a change in the temperature; (b) at constant pressure by changing the volume indirectly through a change in the temperature; (c) at constant temperature, that is *isothermally*, and hence in accordance with the relation of Boyle's Law; the values of p and V are then related as $pV = p_1V_1 = \text{a constant}$; (d) without the addition or subtraction of heat, that is, as it is called, *adiabatically*; as will be seen in Section 21 the equation expressing this relation is $pV^\kappa = p_1V_1^\kappa = \text{a constant}$; (e) in accordance with the condition that the intrinsic energy shall be constant, that is, an *isoenergetic* transformation; for a perfect gas this condition is evidently equivalent to that of an isothermal change; (f) the change may be under the condition of constancy in what is called the *entropy* of the gas, that is, the change may be *isentropic*; for a reversible transformation, as will be discussed in Section 31, this is equivalent to

¹ See Edser, "Heat for Advanced Students," p. 302.

an adiabatic change; or (*g*) the transformation may not be in accord with any of the above conditions, in which case frequently it may be represented by a relation of the form $pV^n = p_1V_1^n = \text{a constant}$; or, it may be assumed to take place in a series of infinitesimal steps which are in accordance with some of the above conditions.

The plots of Figure 2, lettered *a*, *b*, *c*, *d*, and *g*, represent graphically changes in accordance with the conditions of *a*, *b*, *c*, *d*, and *g*,

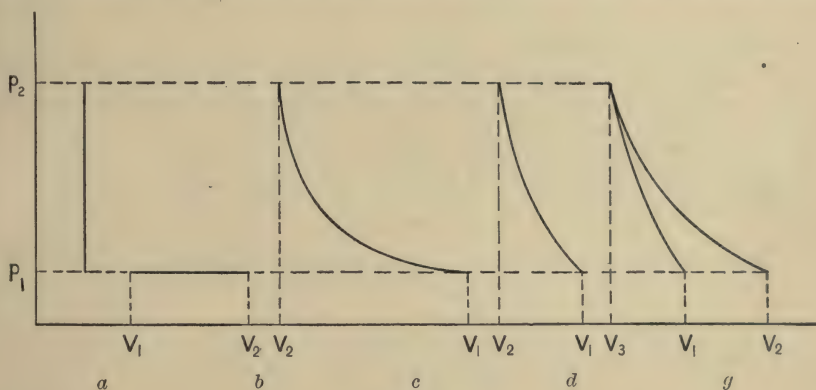


FIG. 2

above, from a volume and pressure of V_1 and p_1 to a volume and pressure of V_2 and p_2 , respectively. The plot (*g*) is drawn to a very much larger scale than the others and represents disproportionately a change as taking place in two imaginary or component infinitesimal steps; namely, from (p_1, V_1) to (p_2, V_3) by an adiabatic compression and from (p_2, V_3) to (p_1, V_2) by an isothermal expansion.

20. Work Calculations for Gas Transformations. The work required to produce a change in the state of a gas, or the external work done by the gas, during some transformation, may be represented graphically upon the pressure-volume plot just described. Thus, consider that a quantity of gas is contained in a vessel fitted with a piston of area a . If the pressure of the gas is p units of force per unit of area, then the total force exerted upon this piston is $F = pa$ units. If the piston is caused to move a small distance dz by this force (or against it), the infinitesimal amount of work dW is $dW = Fdz = padz$. During this motion the volume is changed by an infinitesimal amount dV , which is evidently equal to adz . The expression for the work done in producing this change is then $dW = Fdz = padz = pdV$. This may be considered positive if the gas, or working substance, produces the change and does this amount

of external work, and negative if the work is done by an external agent upon the working substance. The total work required to produce the change from a volume V_1 to a volume V_2 is then

$$W = \int_{V_1}^{V_2} p dV. \quad (15)$$

If the summation is performed between the limits as shown above, it is evident that an expansion will be positive and correspond to work done by the substance, and similarly a compression will be negative and correspond to work done upon the substance.

The integration given above is only possible analytically when p is either constant or some known function of V . In the " p - V plot," however, the work required to produce a given change in volume is always to be represented by the area enclosed between the axis of volumes, the curve $p = f(V)$, and the two ordinates of values V_1 and V_2 .

It is important to note that although the above discussion referred specifically to gases, the integral $\int p dV$ between the proper limits always gives the mechanical work required to produce the change in volume indicated. In the case of the transformations of steam in a steam engine the pressure-volume plot is drawn by a mechanism known as an indicator, which is attached to the cylinder of the engine in such a manner as to record pressures and to the piston of the engine so as to record piston displacements. The work done by the steam is then found approximately by using a planimeter to determine the area.

The cases mentioned in Section 19 and shown in Figure 2, *a*, *b*, *c*, and *d*, will now be discussed. For completeness, in addition to representing the value of the work integral there will be recorded at this time the energy relations and the heat relations for these changes, although some of these relations are not fully developed until Sections 21 and 22. The following symbols will be used: to represent intrinsic energy E ; to represent heat added to the substance from some external source \mathfrak{Q} . The subscripts 1 and 2 will represent states 1 and 2 respectively. It is to be noticed that the relations given below hold only for a perfect gas or one which may be assumed perfect, and in every instance of a change of volume the change is taken as an expansion; that is, V_2 is greater than V_1 .

(*a*) Since the volume is constant, $W = 0$. Because the temperature is increased the intrinsic energy is increased and by an amount

$$E_2 - E_1 = J\mathfrak{Q} = c_v m(t_2 - t_1). \quad (16)$$

(b) Since p is constant,

$$\begin{aligned} W &= p(V_2 - V_1). \quad \text{Also} \\ J\Delta &= c_p m(t_2 - t_1), \\ &= W + E_2 - E_1. \quad \dots \dots \dots (17) \end{aligned}$$

(c) For an isothermal expansion

$E_2 - E_1 = 0$, and $\Delta = AW$, where

$$\begin{aligned} W &= \int_{V_1}^{V_2} p dV = \int_{V_1}^{V_2} p_1 V_1 \frac{dV}{V} = p_1 V_1 \log_e \frac{V_2}{V_1} = p_1 V_1 \log_e \frac{p_1}{p_2} \\ &= mR\theta_1 \log_e \frac{p_1}{p_2}. \quad \dots \dots \dots (18) \end{aligned}$$

(d) For an adiabatic expansion $\Delta = 0$, and $W = E_2 - E_1$, where

$$\begin{aligned} W &= \int_{V_1}^{V_2} p dV = \int_{V_1}^{V_2} p_1 V_1^\kappa \frac{dV}{V^\kappa} = \frac{p_1 V_1}{\kappa - 1} \left[1 - \left(\frac{V_1}{V_2} \right)^{\kappa-1} \right] \\ &= \frac{p_1 V_1}{\kappa - 1} \left[1 - \left(\frac{p_2}{p_1} \right)^{\frac{\kappa-1}{\kappa}} \right] = \frac{p_1 V_1 - p_2 V_2}{\kappa - 1}. \quad \dots \dots \dots (19) \end{aligned}$$

These integrations should be verified by the student. The results will be used in subsequent discussions.

21. Adiabatic Transformation. The equation stated in Section 19 as representing the relation for an adiabatic transformation of a gas will now be deduced. Consider a temperature change of infinitesimal amount taking place in a unit mass of the gas as the result of the addition of an infinitesimal amount of heat energy δq . Imagine the change to take place in two infinitesimal steps as follows, a change in temperature of amount $(\delta\theta)_v$ while the volume remains constant, and a change of $(\delta\theta)_p$ while the pressure remains constant. Since part of this change is at constant volume and part at constant pressure, the amount of heat δq required to produce the change is the sum of the heat required to produce these component changes, namely, $c_v(\delta\theta)_v$ and $c_p(\delta\theta)_p$, where c_v and c_p are the specific heats at constant volume and at constant pressure respectively.

$$\text{Thus} \quad \delta q = c_v(\delta\theta)_v + c_p(\delta\theta)_p. \quad \dots \dots \dots (20)$$

Since, however, for this unit mass of gas p and v are related by the equation $p v = R\theta$, it follows that a change of temperature of amount $(\delta\theta)_v$ corresponds to and is expressible by $(v/R)\delta p$, and similarly a change of temperature $(\delta\theta)_p$ is expressible as $(p/R)\delta v$. Making

this change of variables and writing in differential notation instead of the notation of infinitesimals, gives

$$dq = c_v \frac{v}{R} dp + c_p \frac{p}{R} dv. \quad . \quad . \quad . \quad . \quad . \quad (21)$$

Equation (21) is a general heat equation for a perfect gas and expresses the effect of adding heat energy in terms of the changes of the pressure and the volume of the gas. In Section 38 similar equations will be written expressing this effect in terms of the other possible variables of the perfect gas equation, namely, p and θ , and v and θ .

In the case of an adiabatic transformation, that is by definition one which takes place under conditions which prevent the transfer of heat to or from the substance, obviously the term dq is zero, and equation (21) becomes

$$c_v \frac{v}{R} dp + c_p \frac{p}{R} dv = 0. \quad . \quad . \quad . \quad . \quad . \quad (i)$$

Upon substitution of $\kappa = c_p/c_v$ (see definition of κ in Section 18),

this gives

$$\kappa \frac{dv}{v} = - \frac{dp}{p}. \quad . \quad . \quad . \quad . \quad . \quad (ii)$$

Solving this differential equation by integration between the limits of p_1 and v_1 , which represents the condition of the gas in the first state, and p_2 and v_2 , which represents its second condition, gives

$$\log_e \left(\frac{v_2}{v_1} \right)^\kappa = \log_e \frac{p_1}{p_2}, \quad . \quad . \quad . \quad . \quad . \quad (iii)$$

whence

$$pv^\kappa = p_1 v_1^\kappa = p_2 v_2^\kappa, \quad . \quad . \quad . \quad . \quad . \quad (iv)$$

or using the volume of m units instead of one unit of mass,

$$p_1 V_1^\kappa = p_2 V_2^\kappa. \quad (22)$$

22. Comparison of Adiabatic and Isothermal Transformations. It is evident from equation (22) that the curve representing an adiabatic transformation on a pressure-volume plot is steeper than that representing an isothermal transformation, inasmuch as the tangent of the angle of slope, given by dp/dv is κ times as large for the adiabatic as for the isothermal curve. This is in accordance with the physical ideas involved; for, considering a compression, it is clear that the work done by an external agent would tend to raise the temperature of the gas. If the change is isothermal, this must be prevented by conducting away the heat equivalent of the work

as fast as it is produced. The isothermal change is represented in Figure 3 by ab . In an adiabatic compression, on the other hand, this is not done, and the heat goes into increasing the molecular motion of the gas. The adiabatic is ac in Figure 3. This increased molecular motion means an increased pressure for the same final volume as was attained in the isothermal compression. Similar reasoning applies to the case of an expansion.

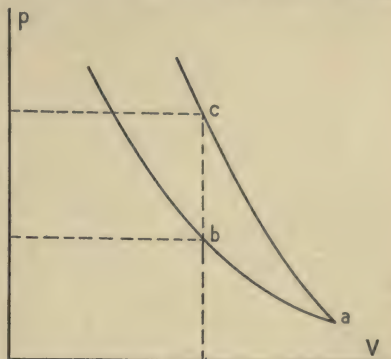


FIG. 3

In general, it is to be noted that if a compression is adiabatic the work done goes into raising the temperature and increasing the intrinsic energy of the gas. If an expansion is adiabatic, part of the intrinsic energy of the gas is transformed into external work and the temperature of the gas is lowered. In both cases the amount of work done is numerically equal to the change in intrinsic energy. On the other hand, if the change is isothermal, heat energy must be given off during a compression and absorbed by the gas during an expansion, and the amount of this heat energy in the case of a perfect gas is equivalent to the work done, as is stated in Joule's Law. Thus, an isothermal transformation for most practical gas problems may be considered an isoenergetic transformation.

23. Second Law of Thermodynamics. It was stated in Section 19 that for a reversible process an adiabatic transformation was also a so-called isentropic transformation. The concept of entropy may best be developed after a discussion of a cycle of changes known as the "Carnot cycle" and of a law known as the "Second Law of Thermodynamics."

The Second Law of Thermodynamics is similar to several laws of physics in that it admits of no direct verification and applies essentially to ideal cases. The first Newtonian law of motion is of this kind, for it states that "every body continues in a state of rest or of uniform motion in a straight line except in so far as compelled by force to change this state." The condition of a body free from the action of forces, either of friction or of gravitational attraction, is one that is not reproducible in nature, and yet the law is accepted as essentially axiomatic, because as nearly as the conditions can be

produced the law is in accord with known facts. The Second Law of Thermodynamics may be accepted in somewhat the same manner, not because it is capable of direct verification but because no deduction from it that is capable of experimental proof has ever failed of verification.

The Second Law of Thermodynamics states that "it is impossible for any self-acting machine, unaided by any external agency, to convey heat from a body at a low temperature to one at a high temperature; or heat cannot of itself (that is, without the performance of work by some external agency) pass from a cold to a warmer body." This law applies only to a type of operations known as "cyclic." Its discussion will be postponed until after a description of the "cycle" of changes suggested by Carnot.

24. Carnot Cycle. Imagine a working substance, as for example air or gas, to be contained in a cylinder fitted with a piston. The

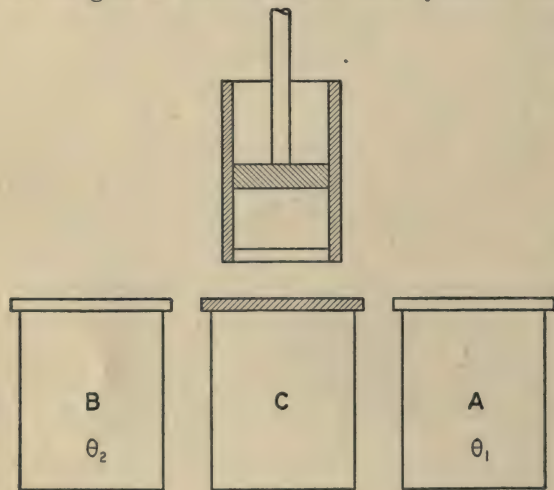


FIG. 4

sides of the cylinder and the piston are supposed to be absolute non-conductors of heat. But the bottom of the cylinder is assumed to be a perfect conductor of heat. Let there be imagined also two sources of heat energy, diagrammed as *A* and *B* in Figure 4, which are maintained at constant temperatures of θ_1 and θ_2 , where θ_1 is higher than θ_2 . Let there also be provided an insulating stand shown at *C*. Let the condition of the gas in the cylinder be represented by the point *a* on the pressure-volume plot of Figure 5, and let this original condition be so chosen that the temperature is θ_1 .

Place the cylinder upon the hot source *A* and allow the gas to expand isothermally from its initial state of p_1, V_1, θ_1 , to a state p_2, V_2, θ_1 , represented by the point *b*. During this expansion the energy necessary to do the external work against the pressure of the piston is derived from the source *A*. Now remove the cylinder

from A and placing it upon the insulating stand C , allow an adiabatic expansion until the temperature has fallen to that of the cold source B , namely, θ_2 . The point c of coördinates p_3 and V_3 represents on the pressure-volume plot the state of the gas. Now place the cylinder upon the source B and compress isothermally until the state of the gas is represented by d of coördinates p_4 and V_4 , and temperature θ_2 . The point d is conditioned by the requirements that it shall lie upon an adiabat through a as well as upon the isothermal through c . Then removing the cylinder, place it upon the stand C and compress the contents adia-

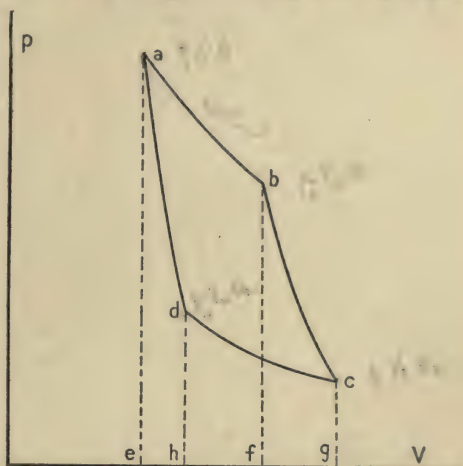


FIG. 5

batically until the temperature has risen to θ_1 , when the state of the gas will again be represented by the point a .

This set of four successive operations is called a cycle for the obvious reason that the working substance undergoes a series of changes which return it to its original condition. The cycle is known as the Carnot Cycle from its proposer, and the scheme here presented is known as a Carnot Engine. In this ideal heat engine the emphasis is placed upon the working substance and its cycle of changes, and not upon the mechanical details by which such an engine could be used practically.

These four operations will now be examined in more detail. During the isothermal expansion from a to b an amount of heat equal to $A \times \text{area } abfe$ has been received from the hot source and converted into work. During the adiabatic expansion from b to c an amount of intrinsic energy equal in heat units to $A \times \text{area } begf$ has been converted into mechanical work. This has been at the expense of the temperature, as noted above. During the isothermal compression from c to d an amount of external work has been converted into $A \times \text{area } cdhg$ heat units and abstracted from the gas by the cold source B . During the adiabatic compression from d to a an amount of heat equal to $A \times \text{area } daeh$ has been given to the working substance as the result of the external work, and the temperature has

consequently risen. The net result of the cycle of operations has been therefore the performance of W units of work by the gas, where

$$W = \text{area } (abfe + becf - cghd - dhea) = \text{area } abcd.$$

Further, since the molecular energy possessed by the working substance is the same at any point on an isoenergetic and hence isothermal line, it follows that the change produced in the intrinsic energy in passing from one isothermal to another is independent of the path followed. Hence, it is considered evident that the loss in intrinsic energy which occurs during the adiabatic expansion from b to c is numerically equal to the gain occasioned by the compression from d to a . Representing by \mathfrak{Q}_1 the entire amount of heat energy taken from the hot source A and equal to $A \times \text{area } abfe$, and similarly representing by \mathfrak{Q}_2 the heat energy rejected to the cold source B , of amount $A \times \text{area } cdhg$, we have, since the working substance returns to its original state, the net amount of heat supplied equal to the net amount of external work performed. Then $\mathfrak{Q}_1 - \mathfrak{Q}_2 = AW$.

The efficiency of the engine is to be written as the ratio of the output AW or $\mathfrak{Q}_1 - \mathfrak{Q}_2$, to the input, which is \mathfrak{Q}_1 . If η represent the efficiency, we have then

$$\eta = \frac{AW}{\mathfrak{Q}_1} = \frac{\mathfrak{Q}_1 - \mathfrak{Q}_2}{\mathfrak{Q}_1} \quad \dots \dots \dots (23)$$

25. Reversible Cycle. The Carnot cycle is reversible, inasmuch as it is possible to perform these operations in the reverse order, and to return the working substance to its initial state. Thus starting with the gas in condition represented by a , allow it to expand adiabatically to the state represented by d in the pressure-volume plot. Then placing it upon the cold source B , allow the gas to expand isothermally from d to c , while receiving an amount of heat \mathfrak{Q}_2 . Then compress adiabatically from c to b , where the temperature is θ_1 . And finally placing the cylinder upon the hot source A , compress isothermally from b to a and reject to the source an amount of heat \mathfrak{Q}_1 . An amount of external work W , equal in value to $J(\mathfrak{Q}_1 - \mathfrak{Q}_2)$, will have been done upon the gas, and an amount of heat \mathfrak{Q}_2 absorbed from the cold source and rejected to the hot source. This transfer of heat from a cold source to a warmer source has been at the expense of work performed by some external agency, and does not therefore constitute a violation of the Second Law;

26. Carnot's Theorem. Upon the assumption of the Second Law a theorem advanced by Carnot may be proved. This theorem states that all reversible heat engines have the same efficiency when absorb-

ing and rejecting heat at the same limiting temperatures. To prove this statement imagine two heat engines, that is, two working substances, operating in the fashion described in Section 24. Let one of these, denoted by *A*, have a greater efficiency than the other engine, which will be represented by *B*. Imagine the mechanical details so arranged that when working with the same hot source and the same cold source the work done by *A* shall equal that done by *B*. Then let *B* pass through its cycle in the reversed direction, in a manner similar to that described in Section 25, and let the mechanical energy necessary to drive it be supplied by *A*. The work done by *A* will then just equal that required to drive *B*, reversed. Let *A* absorb from the hot source an amount of heat \mathfrak{Q}_1 , and reject to the cold source an amount \mathfrak{Q}_2 . Similarly, let *B* absorb from the cold source an amount \mathfrak{Q}_2' , and reject to the hot source an amount \mathfrak{Q}_1' . Then if the heat taken from a source be indicated by a minus sign and that rejected to a source by a plus sign, the net result of the two operations upon the heat sources is shown in the following tabular arrangement :

	HOT SOURCE	COLD SOURCE
Engine <i>A</i>	$-\mathfrak{Q}_1$	$+\mathfrak{Q}_2$
Engine <i>B</i>	$+\mathfrak{Q}_1'$	$-\mathfrak{Q}_2'$
Total heat	$\mathfrak{Q}_1' - \mathfrak{Q}_1$	$\mathfrak{Q}_2 - \mathfrak{Q}_2'$

Now since the work done by *A* is equal to that which *B* would do if passing through its cycle in the positive direction, it follows that

$$\mathfrak{Q}_1 - \mathfrak{Q}_2 = \mathfrak{Q}_1' - \mathfrak{Q}_2'. \quad \dots \dots \dots (i)$$

Also, since by assumption the efficiency of *A* is greater than that of *B*,

$$\frac{\mathfrak{Q}_1 - \mathfrak{Q}_2}{\mathfrak{Q}_1} > \frac{\mathfrak{Q}_1' - \mathfrak{Q}_2'}{\mathfrak{Q}_1'}. \quad \dots \dots \dots (ii)$$

Combining equations (i) and (ii) gives

$$\frac{1}{\mathfrak{Q}_1} > \frac{1}{\mathfrak{Q}_1'}, \quad \dots \dots \dots (iii)$$

and hence

$$\mathfrak{Q}_1' > \mathfrak{Q}_1.$$

Further, by comparing equations (i) and (iii) it is evident that

$$\mathfrak{Q}_2' > \mathfrak{Q}_2.$$

In the tabular arrangement above it appears therefore that an amount of heat is taken from the cold source and added to the hot

source. The net result is that a self-acting machine (or combination of machines) performing a cycle is transferring heat from a cold body to a hot body without the performance of work by some external agency. This is impossible, according to the Second Law. Therefore Ω_1' cannot be greater than Ω_1 and also Ω_2' cannot be greater than Ω_2 .

Hence the efficiency of engine *A* cannot be greater than that of engine *B*. A similar process of reasoning will show that the efficiency of *B* cannot be greater than that of *A*. Hence the efficiencies of any two (or of all) reversible heat engines absorbing heat from the same hot source and rejecting heat to the same cold source must be the same.

27. Discussion of Reversible Processes. Planck says of reversible processes:¹ "A process which can in no way be completely reversed is termed irreversible; all other processes are reversible. That a process may be irreversible it is not sufficient that it cannot be directly reversed. This is the case with many mechanical processes which are not reversible. The full requirement is that it be impossible, even with the assistance of all agents in nature, to restore everywhere the exact initial state when the process has once taken place. The generation of heat by friction, the expansion of a gas without the performance of external work, the absorption of external heat, and the conduction of heat, and so on, are irreversible processes. Since there exists in nature no process entirely free from friction or heat conduction, all processes which actually take place in nature are in reality irreversible; reversible processes form only an ideal limiting case. They are, however, of considerable importance for theoretical demonstration and for application to states of equilibrium."

28. Discussion of the Second Law. In a summary of the foundations of thermodynamics Bryan says:² "Passage of heat from one body to another is usually irreversible and therefore accompanied by a loss of available energy. If we define *A* to be hotter or colder than *B* according as available energy is lost or gained by the transference of heat from *A* to *B*, it follows that heat can, and in general will, pass from hotter to colder bodies, but the reverse change can only be effected by combining it with a compensating transformation. Carnot's cycle reversed is a compensated reversible transformation by which heat can be continuously taken from a colder and given to a hotter body, or *vice versa*, without loss of availability.

¹ Planck, "Thermodynamics," translated by Ogg, p. 85.

² Bryan, "Thermodynamics," Teubner, 1907.

In this case the compensating transformation takes the form of work absorbed or produced."

The conclusion of Carnot's Theorem given in Section 26 is often stated as the Second Law. That theorem, however, holds only for reversible cyclic processes and is not as broad a statement of the Second Law as that quoted in Section 23, which is due chiefly to Clausius. An excellent statement of the essence of the law is quoted from Drude¹ as follows: "Mechanical work can never be continually obtained at the expense of heat if only one source of heat at uniform temperature is at disposal."

From this statement there may easily be derived an alternative proof of Carnot's Theorem. Thus consider the engines *A* and *B* of Section 26, in which engine *A* is assumed to be of greater efficiency than engine *B*, to be so arranged mechanically that they would each take from the same hot source the same amount of heat, but because of greater efficiency *A* would convert more into work and reject less to the cold source than would *B*. Now imagine engine *A* to drive engine *B*, reversed, and let *B* give up to the hot source just as much heat as *A* absorbs from it. Then the hot source may be omitted from consideration and *B* supply *A* directly. Because *A* has a higher efficiency it will reject less heat to the cold source than will *B* with its lower efficiency abstract from it. Further, *A* will do more work than will *B*, and hence there will be a net amount of work available for external purposes. But this would be in violation of the Second Law, for this external work would be done as a result of a net amount of heat absorbed from the cold source; that is, from the single source of heat at disposal. Hence there can be no external work. Hence, further, the efficiencies must be the same.

In this expression of the substance of the Second Law by Drude the emphasis may for the moment be placed upon the word "continually," which is essentially equivalent to the term "in a cyclic process." As an illustration of the fact that the Second Law holds only for cyclic processes, consider the following example. Imagine two non-conducting cylinders, one containing air at a high pressure and a low temperature and the other air at a low pressure and a high temperature. If they are fitted with connecting pistons so that the higher pressure air may expand and compress that in the other cyl-

¹ Drude, "Theory of Optics," translated by Mann and Millikan, pp. 493-494. This statement is equivalent to that of Lord Kelvin; namely, "It is impossible, by means of inanimate material agency, to derive mechanical effect from any portion of matter by cooling it below the temperature of the coldest of surrounding objects."

inder, then, for this adiabatic change the temperature in the high pressure cylinder will fall, while that in the low pressure cylinder will rise. During the process, then, heat has been abstracted from the low temperature side and given to the high temperature side. This does not, however, constitute a violation of the Second Law, since the change is not cyclic.¹

Imagine that the low pressure cylinder is in connection with the atmosphere as in a vacuum pump. The high pressure side then does external work in pushing out the atmosphere contained in the pump, and in so doing the working substance in the high pressure cylinder is cooled still farther below the temperature of the surrounding atmosphere. Heat may then be absorbed from the atmosphere and the original temperature thus regained. But still the working substance is not returned to its original state, and to return it would require as much work as was originally obtained from its expansion. The example cited is then in accordance with the Second Law.

In general, it may be said for the production of mechanical work two sources of heat at different temperatures are required, and the production of work requires the transfer of heat in such a direction as to tend to result in a final equality of temperature of the sources, in which condition all the available energy of the system will have been used up.

29. Thermodynamic Scale of Temperature. For a reversible cycle it has been seen that the efficiency of all heat engines is the same provided only that the temperature limits are the same. Lord Kelvin therefore proposed a thermodynamic scale of temperatures which is independent of the substance used.

Draw any two adiabatic lines for a substance and also two isothermal lines (for reasons evident later the isothermals will be those for temperatures of 0° and 100° C.). Let the area bounded by the two isothermals and the two adiabatics be divided into 100 equal parts by drawing other isothermals. Calling the area between any two adjacent isothermals a , divide the space below the zero degree Centigrade isothermal and between the two adiabatics into areas equal to a by drawing other isothermals. In Figure 6 this has been represented by drawing the isothermals dividing the space between the 0° and the 100° isothermal into four parts, and extending this division below the 0° isothermal as far as is possible in a finite

¹ In part this illustration is derived from Edser, "Heat for Advanced Students," p. 340.

diagram. Let the lowest isothermal corresponding to the condition of the substance when there is no intrinsic molecular energy be

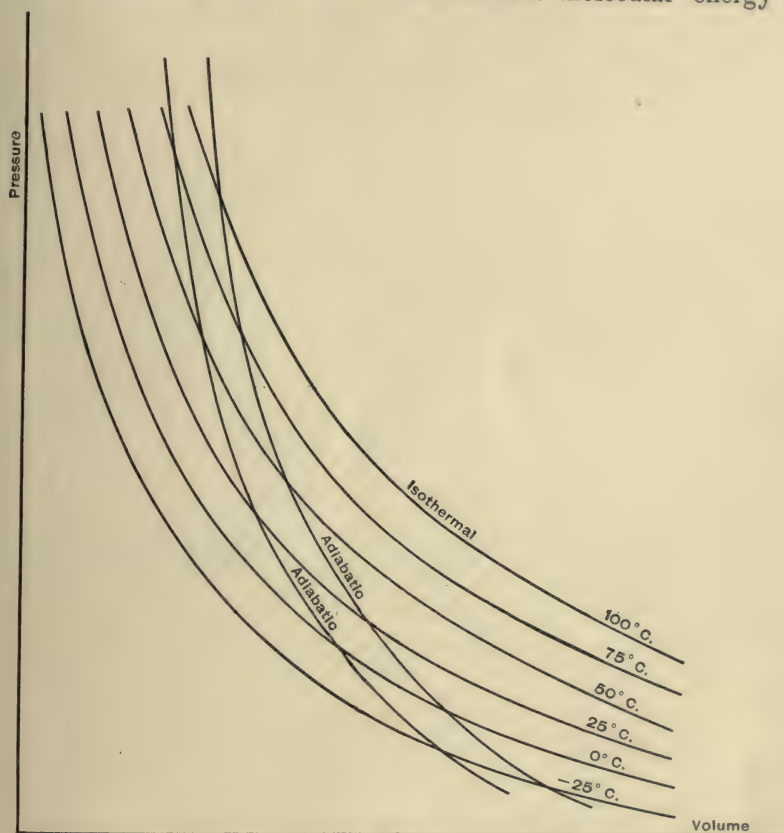


FIG. 6

called the absolute zero of temperature as in Section 15. Number the isothermals from this absolute zero, and let the number of any isothermal be represented by T .

Consider now a Carnot cycle composed of these two adiabatics and any two isothermals as T_1' and T_2' , where T_1' is higher than T_2' .

The efficiency of this cycle is $\frac{\Omega_1' - \Omega_2'}{\Omega_1'}$, where Ω_1' and Ω_2' are measured as in Section 20 by the areas under the isothermal lines. The numerator of this expression is, however, equal to the area included by the two isothermals and the two adiabatics. If now a particular value of zero be given to T_2' , and if T_1' be put equal to T , then the expression for the efficiency of the Carnot cycle between

these limiting isothermals becomes $\frac{\Omega - 0}{\Omega} = \frac{Ta}{\Omega} = 1$. That is, $\Omega = Ta$.

This means merely that in traversing any isothermal between the two limiting adiabatics, the input is $T \times a$ in work units. Hence the efficiency of a Carnot cycle between two isothermals T_1 and T_2 is given by the expression

$$\eta = \frac{\Omega_1 - \Omega_2}{\Omega_1} = \frac{T_1 a - T_2 a}{T_1 a} = \frac{T_1 - T_2}{T_1} \quad \dots \quad (24)$$

If the numbers which have been assigned to the isothermals be considered to be temperatures, it is evident that we have a method for measuring temperature which is independent of the substance chosen. The zero of this thermodynamic scale coincides with the perfect gas thermometer zero, and the size of a degree is approximately the degree Centigrade as measured on a hydrogen thermometer. If now it can be shown that the efficiency of a Carnot cycle with a perfect gas for a working substance is $\frac{\theta_1 - \theta_2}{\theta_1}$, where θ_1 and θ_2 are the absolute temperatures on a perfect gas thermometric scale, it will be evident that the thermodynamic scale of Lord Kelvin is identical with the perfect gas scale.

30. Carnot Cycle for a Perfect Gas.

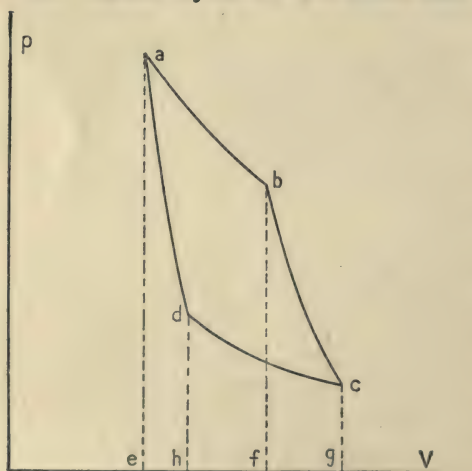


FIG. 7

It has already been shown that the work done by a perfect gas during the adiabatic expansion from b to c of the Carnot cycle shown in Figure 5 (reproduced here as Figure 7) is equal to that done upon the gas during the adiabatic compression from d to a . It remains only to find the values of Ω_1 and Ω_2 in order to arrive at an expression for the efficiency of the Carnot cycle for a perfect gas. If the efficiency can be expressed entirely in terms of the absolute temperatures,

then the comparison suggested in the preceding section can be made.

According to equation (18) the area under the isothermal a to b , which is equal to $J\Omega_1$, is expressible as $mR\theta_1 \log_e (V_b/V_a)$. And

similarly, the heat rejected in passing along the isothermal from c to d is $\mathfrak{Q}_2 = AmR\theta_2 \log_e (V_c/V_d)$. The efficiency is therefore

$$\begin{aligned}\eta &= \frac{\mathfrak{Q}_1 - \mathfrak{Q}_2}{\mathfrak{Q}_1} = \frac{AmR\theta_1 \log_e \frac{V_b}{V_a} - AmR\theta_2 \log_e \frac{V_c}{V_d}}{AmR\theta_1 \log_e \frac{V_b}{V_a}} \\ &= \frac{\theta_1 \log_e \frac{V_b}{V_a} - \theta_2 \log_e \frac{V_c}{V_d}}{\theta_1 \log_e \frac{V_b}{V_a}}.\end{aligned}$$

During the isothermal changes

$$p_a V_a = p_b V_b = mR\theta_1, \text{ and also } p_c V_c = p_d V_d = mR\theta_2.$$

Hence
$$\frac{p_d V_d}{p_a V_a} = \frac{\theta_2}{\theta_1} \text{ and } \frac{p_c V_c}{p_b V_b} = \frac{\theta_2}{\theta_1}.$$

Similarly, for the adiabatic changes $p_b V_b^\kappa = p_c V_c^\kappa$ and $p_d V_d^\kappa = p_a V_a^\kappa$. Substituting from these adiabatic equations gives

$$\frac{\theta_2}{\theta_1} = \left(\frac{V_d}{V_a}\right)^{\kappa-1} \text{ and } \frac{\theta_2}{\theta_1} = \left(\frac{V_c}{V_b}\right)^{\kappa-1},$$

therefore
$$\frac{V_d}{V_a} = \frac{V_c}{V_b} \text{ or } \frac{V_b}{V_a} = \frac{V_c}{V_d}.$$

Substitution of this ratio in the expression for the efficiency gives

$$\eta = \frac{\theta_1 - \theta_2}{\theta_1} \dots \dots \dots (25)$$

A comparison of equations (24) and (25) shows that a perfect gas scale is identical with the absolute thermodynamic scale.

31. Entropy. In equation (23), for the efficiency of a Carnot perfect gas cycle, if the two isothermals are taken at an infinitesimal difference of temperature apart, so that $\theta_1 - \theta_2 = d\theta$, then the corresponding heat $\mathfrak{Q}_1 - \mathfrak{Q}_2$, converted into work, may be represented by the infinitesimal $d\mathfrak{Q}$, and by equation (25) the expression for the efficiency becomes

$$\frac{d\mathfrak{Q}}{\mathfrak{Q}} = \frac{d\theta}{\theta}.$$

The solution of this differential equation gives

$$\log_e \frac{\Sigma_1}{\Sigma_2} = \log_e \frac{\theta_1}{\theta_2}, \text{ or } \frac{\Sigma_1}{\Sigma_2} = \frac{\theta_1}{\theta_2},$$

and hence

$$\frac{\Sigma_1}{\theta_1} = \frac{\Sigma_2}{\theta_2}. \quad . \quad . \quad . \quad . \quad . \quad . \quad . \quad (26)$$

This means that for a reversible process the ratio of the heat absorbed (or rejected) to the absolute temperature of the isothermal traversed in passing between two adiabatics is independent of the isothermal traversed. The quantity which remains constant under these conditions is the *difference in entropy* of the substance in the states represented by the two adiabatics. An *adiabatic line* is then *for a reversible process by definition a line of constant entropy*, or an *isentropic*.

The entropy of any state will be represented by the usual symbol ϕ . The expression $\frac{\Sigma}{\theta}$ is then an expression for the difference in entropy, to be denoted by $\phi_2 - \phi_1$, between the entropy ϕ_1 of the first adiabatic and ϕ_2 , the entropy corresponding to the second adiabatic. This change in entropy between these two isentropics is independent of the path followed, provided only the path is a reversible one, as it was in the case of the isothermal of the Carnot cycle from which this concept was developed.

In general, if a small amount of heat $d\Sigma$ be added to a body at a temperature θ , which is sensibly constant during that addition, then the entropy of the body has been increased by an amount $d\phi$, where

$$d\phi = \frac{d\Sigma}{\theta}.$$

If the change in entropy is produced by a large number of such additions, by processes which are completely reversible, the total change in entropy thereby produced is

$$\phi_2 - \phi_1 = \int_1^2 d\phi = \int_1^2 \frac{d\Sigma}{\theta}. \quad . \quad . \quad . \quad . \quad . \quad . \quad . \quad (27)$$

For the special case where the heat is added along an isothermal this reduces to

$$\phi_2 - \phi_1 = \frac{\Sigma}{\theta}.$$

Further, it is to be noticed that for the Carnot cycle

$$\frac{\Sigma_1}{\theta_1} - \frac{\Sigma_2}{\theta_2} = 0.$$

That is, for a reversible cycle

$$\oint \frac{d\Sigma}{\theta} = 0, \quad (28)$$

where the symbol \oint indicates the fact that the integration is to be performed about a complete cycle. This equation is sometimes stated as a mathematical form of the Second Law, but it is preferable to consider it a deduction from it.

32. Measurement of Entropy. It is not possible to determine the total entropy of a body. That is, the expression $\int \frac{d\Sigma}{\theta}$ when used to find the entropy of a body in a given state will always contain a constant of integration. The integral does, however, admit of the determination of the difference in entropy between two given states. Hence it is that entropy is always expressed as so much more or less than the entropy of some chosen state of the substance under consideration. This is satisfactory for all practical uses of the concept of entropy. Having arbitrarily selected some adiabatic as a reference state to which all other states of the substance are to be compared, it is possible to conceive of the entire pressure-volume plot, for a given mass of the substance, as filled with adiabatic or isentropic lines which represent states of the substance differing by units of entropy, just as the states represented by the isothermal lines differ by units of temperature.

Quoting from Preston,¹ "The entropy of a body being arbitrarily taken as zero in some standard condition *A* defined by some standard temperature and pressure (or volume), the entropy in any other state *B* is the value of the integral $\int \frac{d\Sigma}{\theta}$ taken along any reversible path by which the body may be brought to state *B* from standard state *A*. The path may obviously be an arc *AC* of an isothermal passing through the point *A*, defining the standard state, together with the arc *BC* of the adiabatic line passing through the point *B*."

33. Temperature-Entropy Plot. It is evident from equation (27) that, provided a change is taking place along a reversible path, the

¹ Preston, "Heat," p. 628.

heat added (or rejected) may be expressed as the product of a temperature term and a difference in entropy term, thus

$$\int_1^2 dQ = \int_1^2 \theta d\phi. \quad \dots \quad (29)$$

For this reason and because of the practical interest in steam engineering of following the changes in entropy of steam it is most convenient to plot values of entropy on a temperature-entropy plot.

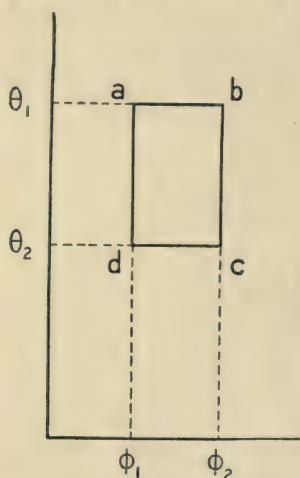


FIG. 8

In the temperature-entropy, or “ θ - ϕ plot,” the ordinates are temperatures and the abscissas are entropies measured from some standard state as described above. The θ - ϕ diagram of the Carnot cycle will now be plotted as an illustration. During the isothermal expansion from a to b of Figure 7, the entropy increases by an amount by $\phi_2 - \phi_1$, and the change is represented as ab in the plot of Figure 8. The reversible adiabatic expansion from b to c is isentropic by definition and is represented on the θ - ϕ plot by the line bc . The isothermal or constant temperature change cd and the reversible adiabatic compression da complete the cycle.

It is to be noticed that for this case, since the isothermal is a reversible process, the heat absorbed at temperature θ_1 is represented by the area under the line ab . Similarly, the heat rejected during the isothermal compression from c to d is given by the area under the line cd . The net heat converted into work in this cycle is then $Q_1 - Q_2 = \theta_1(\phi_2 - \phi_1) - \theta_2(\phi_2 - \phi_1) = (\theta_1 - \theta_2)(\phi_2 - \phi_1)$; that is, it is represented by the area enclosed by the θ - ϕ plot of the cycle.

It is important to note carefully that it is only when the transformation takes place in a reversible manner that the heat absorbed (or rejected) is represented by the area underneath the curve on the θ - ϕ diagram representing the transformation. In all other cases the increase in entropy will be disproportionate to the increase in heat contents as in the case of the porous plug experiment to be described in Section 35. Although this restriction limits this use of a θ - ϕ plot to the ideal cases, such a plot is still of great interest and value in many practical problems.

34. Discussion of Carnot Cycle from θ - ϕ Plot. If any other reversible cycle be plotted upon the θ - ϕ diagram as in Figure 9, it is evident that the work output, which would be proportional to the area enclosed by the plot, is less than for the Carnot cycle, although the same range of temperatures is used. It follows then that no reversible heat engine can have an efficiency greater than that of the Carnot engine. And of course all heat engines with irreversible transformations are less efficient for the same limiting temperatures than the reversible engines. An example of a reversible cycle equal in efficiency to the Carnot cycle is given in Problem 11, on page 51.

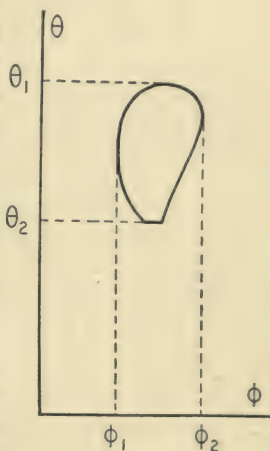


FIG. 9

It is evident from Figures 8 and 9 that the high efficiency of the Carnot cycle is obtained as the result of two conditions, namely, that all the heat absorbed is taken in at the highest temperature, and all the heat rejected is at the lowest temperature. In so far as any heat engine conforms to these requirements (neglecting the fact that its other transformations will not in general be isentropic) will its efficiency approach the limiting and maximum efficiency of a Carnot engine working between the same temperatures.

35. Entropy Changes in Irreversible Processes. Returning again to the concept of entropy, it may be said that entropy is that property of a body which remains constant so long as the body is undergoing adiabatic changes that are reversible. If the change is not reversible, then the entropy in the final state is greater than it would be for a reversible transformation. This is indicated by the classical "porous plug experiment" of Kelvin and Joule. In this experiment, which offers a more accurate method than does that described in Section 11 for testing Joule's Law, it was found that when air expanded from a higher pressure to a lower pressure without doing external work there was a distinct lowering of the temperature, due to the partial conversion of the kinetic energy of the molecules into potential energy of molecular separation. The experiment was performed by allowing air to expand from a high pressure through a plug of cotton wool to atmospheric pressure. The plug was surrounded with a non-conductor and the expansion was adiabatic.

For the purposes of the present discussion the important point of the experiment is as follows. A mass of the gas reaching an opening in the plug, upon the other side of which the pressure is lower, expands adiabatically (a reversible and hence isentropic change), converting a portion of its intrinsic energy into mechanical kinetic energy of the entire mass. This mechanical kinetic energy is immediately converted into heat by friction, viscosity, and the like. The quantity of heat corresponding to this friction work is now returned to the air and causes an increase in entropy, exactly as if it had been added from some outside source. As a result of these two infinitesimal changes, the first isentropic and the second of increasing entropy, the final entropy of the air is obviously increased. The same conclusions as to an increase in entropy may be made for the original experiment of Joule, but it is easier to conceive of the steps in this case. This expansion, although adiabatic,

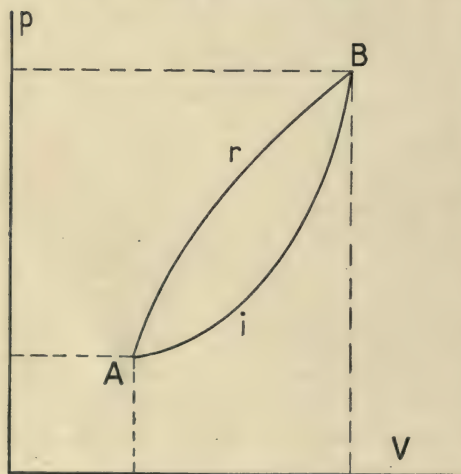


FIG. 10

is obviously irreversible and the entropy is increased. For further discussion it will be assumed to have been shown that all irreversible transformations lead to a final state of greater entropy than would a corresponding reversible transformation.

Consider a change of state for a given substance from A to B by a reversible path as represented on the pressure-volume plot of Figure 10 by the line ArB . This change might be represented on the temperature-entropy plot of Figure 11 by the line ArB . If now the transformation from A to B is along some irreversible path as AiB in Figure 10, it may be represented on the ϕ - θ plot by some line AiB' , since the final entropy is greater than for the reversible change. The integral $\int_A^B \frac{d\phi}{\theta}$ gives the change in entropy along the reversible path, but is less than the true change along the irreversible path. Thus

$$\phi_{B'} - \phi_A > \int_A^B \frac{d\phi}{\theta} \quad \dots \dots \dots (i)$$

Similarly, if the transformation take place from B to A , the entropy along the reversible path is given by the integral $\int_B^A \frac{d\mathfrak{S}}{\theta}$, but the entropy in the final state for the irreversible path is greater than that for the reversible path. That is,

$$\phi_{A'} - \phi_B > \int_B^A \frac{d\mathfrak{S}}{\theta} \quad \dots \quad (ii)$$

For the irreversible change, represented on the p - v plot by $ArBiA$ the total change in entropy is not zero as it would be for an entirely reversible process, but is obviously greater than zero (or at the most equal to it). For, from A to B by a reversible process

$$\phi_B - \phi_A = \int_A^B \frac{d\mathfrak{S}}{\theta}, \quad \dots \quad (iii)$$

and from B to A by an irreversible process

$$\phi_{A'} - \phi_B > \int_B^A \frac{d\mathfrak{S}}{\theta} \quad \dots \quad (iv)$$

Then by the path $ArBiA$

$$\phi_{A'} - \phi_A > \int \frac{d\mathfrak{S}}{\theta} = 0. \quad \dots \quad (v)$$

That is, for the irreversible process just considered, the

expression $\int \frac{d\mathfrak{S}}{\theta}$ taken for the entire change is less than the actual

increase of entropy. But the physical state represented by $\phi_{A'}$ is not the same as the initial physical state of entropy ϕ_A . In order, then, that the process shall be a true cycle, it is necessary to subtract an amount of entropy of value in this case $\phi_{A'} - \phi_A$. This fact, usually stated in the form given below, is due to Clausius.

$$\left(\int \right) \frac{d\mathfrak{S}}{\theta} \leq 0. \quad \dots \quad (30)$$

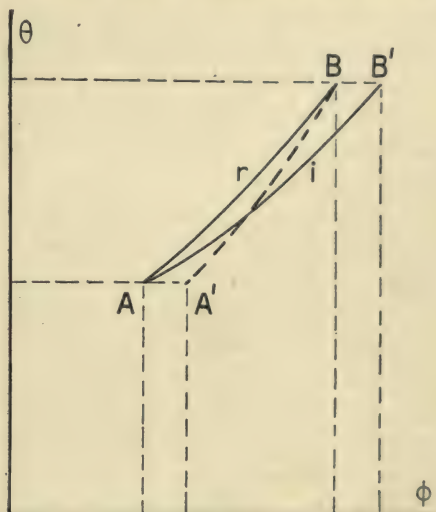


FIG. 11

CHAPTER II

GASES

36. Graphical Representation of State for a Perfect Gas. It is a simple fact of solid analytical geometry that a function of three variables in general represents a surface. The general equation for a given mass m of a perfect gas, namely $pV = mR\theta$, is evidently a function of three variables and may be taken as representing a surface plotted with reference to three rectangular axes, p , V , and θ , corresponding to x , y , and z . This surface is known as the "characteristic surface" of the gas. Since the position of any point on this surface represents a corresponding state or condition of the gas, such a point is usually termed a "state point" of the gas. Now, whatever changes the given mass of gas may undergo, its successive states must be represented by state points on this characteristic surface, corresponding to the successive values of p , V , and θ which the gas assumes during its transformation. The continuous curve passing through these points is a curve in space and has an equation corresponding to the type of transformation which the gas has undergone.

The equation in rectangular coördinates of a curve in space is usually expressed by two equations, each containing two variables only. These two equations represent two cylinders, which are right projecting cylinders of the curve upon the two coördinate planes corresponding to the two sets of variables in which the equations are expressed. The intersection of these two cylinders determines completely the curve in space. In the case of a curve in space representing a gas transformation, since the curve must lie upon a known surface, one equation in addition to the equation for this characteristic surface is sufficient. This equation might be expressed in terms of p and V , p and θ , or V and θ .

The projection on the p and V plane is most serviceable for ordinary purposes, as is evident from the discussion of Sections 19 and 20. It is to be noticed that the expressions there given in p and V coördinates for various type transformations represent equations for a right projecting cylinder of the space curve, perpendicular to the

pressure-volume plane, and also that the third coördinate has already been found from the characteristic equation by substitution of the values of p and V .

In general, a characteristic surface of unit mass of any substance may be represented by the equation stating the relation of any three of the five following magnitudes: specific pressure, specific volume, absolute temperature, intrinsic energy, and entropy. This is true because in general if any two of these magnitudes are known for a given state of the substance, the values of the remaining three for this state may be expressed as functions of the first two. Two exceptions occur, one evidently in the case of perfect gases (and for practical purposes in the case of all gases) inasmuch as the absolute temperature of a gas is directly proportional to its intrinsic energy. (The other exception is in the case of a vapor where the temperature and the pressure do not completely determine the other magnitudes.)

The expression of a transformation in terms of the absolute temperature and the entropy is frequently of value, and the temperature-entropy plot will be used throughout the remainder of this text.¹ Determinations of entropy and intrinsic energy from direct measurements are of course impossible. The three magnitudes that may be directly measured, namely, the temperature, pressure, and volume, are all easily found. For this reason, as well as because of the relation between work and area on a p - v plot (see Section 19), the pressure-volume plot is used almost exclusively in practical work.

37. Equations for Polytropic Transformation of a Perfect Gas. The equation for the most general type of gas transformation has been given in Section 19 as

$$p_1 V_1^n = p_2 V_2^n = \text{a constant.} \quad \dots \dots \dots (i)$$

For n equal to unity this equation expresses the relations of an isothermal change, and for n equal to κ the relations of an adiabatic change.

Since it is frequently desirable to express some special form of such a polytropic transformation in terms of p and θ , or of V and θ , general equations for that purpose will now be derived. Combining equation (i) with the relation for a perfect gas, namely,

$$\frac{p_1 V_1}{\theta_1} = \frac{p_2 V_2}{\theta_2}, \quad \dots \dots \dots (ii)$$

¹ A pressure-entropy plot is sometimes used in superheated steam problems, as is discussed on page 96.

Substituting from (ii) in (33) gives equation (34), which expresses in terms of the temperature, θ , and pressure, p , the same effect as does (33). Thus

$$dq = c_p \frac{\theta}{v} \left[\frac{R}{p} d\theta - \frac{R\theta}{p^2} dp \right] + c_v \frac{\theta}{p} dp = c_p \frac{R\theta}{pv} d\theta - \left(c_p \frac{R\theta^2}{p^2v} - c_v \frac{\theta}{p} \right) dp.$$

Hence

$$dq = c_p d\theta - (c_p - c_v) \theta \frac{dp}{p}. \quad . \quad . \quad . \quad . \quad . \quad (34)$$

Similarly, substituting from (iii) in (33) gives dq in terms of v and θ . Thus

$$dq = c_v d\theta + (c_p - c_v) \theta \frac{dv}{v}. \quad . \quad . \quad . \quad . \quad . \quad (35)$$

39. Entropy Change for Perfect Gas. From the three equations derived above there are easily obtained three expressions for the difference of entropy between two states of a gas provided the change produced may be considered reversible. In case of a reversible transformation

$$\phi_2 - \phi_1 = \int_1^2 \frac{dq}{\theta}.$$

If the coördinates of state 1 be represented by p_1 , v_1 , and θ_1 , and similarly the coördinates of state 2 by p_2 , v_2 , and θ_2 , then we may obtain from equation (33) the following expression for the difference in entropy:

$$\phi_2 - \phi_1 = \int_1^2 \left[c_v \frac{dp}{p} + c_p \frac{dv}{v} \right] = \int_{p_1}^{p_2} c_v \frac{dp}{p} + \int_{v_1}^{v_2} c_p \frac{dv}{v}$$

or

$$\phi_2 - \phi_1 = c_v \log_e \frac{p_2}{p_1} + c_p \log_e \frac{v_2}{v_1}. \quad . \quad . \quad . \quad . \quad . \quad (36)$$

And similarly, equations (34) and (35) lead to equations (37) and (38), as follows:

$$\phi_2 - \phi_1 = c_v \log_e \frac{\theta_2}{\theta_1} + (c_p - c_v) \log_e \frac{v_2}{v_1}. \quad . \quad . \quad . \quad (37)$$

$$\phi_2 - \phi_1 = c_p \log_e \frac{\theta_2}{\theta_1} - (c_p - c_v) \log_e \frac{p_2}{p_1}. \quad . \quad . \quad . \quad (38)$$

Any one of these three equations may be used to calculate the change in entropy. Thus if the change is isothermal, (37) and (38) both reduce to a simple form, since the logarithm of the term involving the temperature reduces to zero in this case. Similarly,

for a constant volume change, (36) or (37) may most easily be used. It is important to notice that, since these equations all involve the specific heat and were derived for unit mass of the gas, the values of the entropy found by their use is in every case *per unit mass* of the substance, and to obtain the total entropy change the value $\phi_2 - \phi_1$ must be multiplied by m . With this understanding it makes no difference whether the ratio of the volumes is v_1/v_2 or V_1/V_2 , since these are equivalent ratios.

40. Relation of the Specific Heats of a Gas. It has already been stated in Section 18 that c_p/c_v is a ratio, κ , that depends upon the

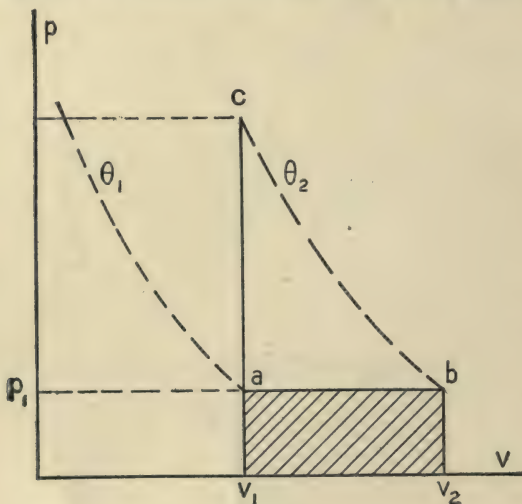


FIG. 12

nature of the gas. It will now be shown that $c_p - c_v = AR$, where R is the constant of the characteristic gas relation.

If unit mass of a perfect gas at a pressure, p , and a volume, v , is heated from θ_1 to θ_2 under the conditions of constant volume, the p - v plot of the change will be as shown in Figure 12 by the line ac . The heat supplied is $c_v(\theta_2 - \theta_1)$. Since no external work is done by the gas, this

expression represents in heat units the increase in intrinsic energy of the gas. If, however, the change from θ_1 to θ_2 be made at constant pressure, as represented by the line ab in the figure, then the heat added is $c_p(\theta_2 - \theta_1)$. Since the final temperatures are the same, the increases in intrinsic energy are the same. During the heating at constant pressure an amount of external work represented by the shaded area of the figure has been done. The value in thermal units of this external work is $Ap(v_2 - v_1)$. It is evident, then, that

$$c_p(\theta_2 - \theta_1) - c_v(\theta_2 - \theta_1) = Ap(v_2 - v_1), \quad \dots \quad (i)$$

or

$$c_p - c_v = \frac{Ap(v_2 - v_1)}{\theta_2 - \theta_1}, \quad \dots \quad (ii)$$

but

$$pv_1 = R\theta_1 \text{ and } pv_2 = R\theta_2. \quad \dots \quad (iii)$$

From (iii) we have

$$p(v_2 - v_1) = R(\theta_2 - \theta_1). \quad \dots \quad (iv)$$

Hence substituting from (iv) in (ii) gives

$$c_p - c_v = AR. \quad \dots \quad (39)$$

41. Changes in Intrinsic Energy of a Gas. For a perfect gas the intrinsic energy consists entirely, as has been said in Section 15, of

molecular kinetic energy and is directly proportional to the absolute temperature. The value of the intrinsic energy of a gas at a given pressure and occupying a given volume may then be expressed in terms of this pressure and volume in accordance with the following reasoning. Imagine the gas to expand adiabatically from the state a , at which the value of the intrinsic energy is desired, until the pressure is zero.

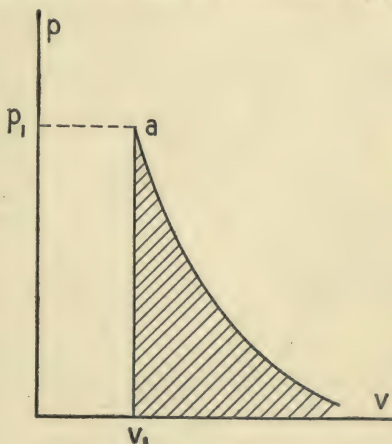


FIG. 13

During this expansion all the intrinsic energy is converted into

external work. Such an expansion is represented diagrammatically in Figure 13. The energy may therefore be measured by the area under the adiabatic drawn through a as in the figure. Denoting the energy in state a as E_1 , we have, according to equation (19) of Section 20,

$$E_1 = \frac{p_1 V_1}{\kappa - 1} \left[1 - \left(\frac{V_1}{V_2} \right)^{\kappa - 1} \right].$$

In this equation the volume V_2 , corresponding to zero pressure, will be infinite and hence the expression for E_1 reduces to

$$E_1 = \frac{p_1 V_1}{\kappa - 1}. \quad \dots \quad (40)$$

This may also be written as

$$E_1 = \frac{m R \theta_1}{\kappa - 1}. \quad \dots \quad (41)$$

From this form it is evident that the value given in equation (40) contains the mass implicitly, and hence the above expressions for intrinsic energy are *for the total energy and not per unit mass of the gas*.

The difference in intrinsic energy between two states 1 and 2 of the same mass of gas is evidently

$$E_2 - E_1 = \frac{p_2 V_2 - p_1 V_1}{\kappa - 1} \quad (42)$$

In order that this energy shall be expressed in foot pounds it is necessary that p be expressed in pounds per square foot and V in cubic feet.

42. Imperfect Gases. As has been previously stated, for ordinary engineering purposes it is sufficient to apply the relations established in this chapter for perfect gases to any of the ordinary so-called "permanent gases" such as hydrogen, oxygen, nitrogen, marsh gas, and mixtures like air (and also, to some extent, to carbon dioxide and the like). When it is desired to analyze the action of any mixture, it may frequently be done by the use of Dalton's Law of partial pressures, which states that if p_1 and p_2 represent the pressures which two chemically inert gases would exert at the same temperature when separately occupying a volume V , then the resultant pressure p for that temperature due to both these gases confined in this volume is the sum of p_1 and p_2 .

Certain constants for the ordinary gases are to be found in Table 2 on page 130.

43. Résumé of Equations for Cases.

The numbers following the equations are the original numbers assigned to them in the preceding chapters.

Temperature.

$$\theta = 459.5 + t \text{ Fahrenheit}; \quad \theta = 273 + t \text{ Centigrade.} \quad (11, 12)$$

External Work.

$$W = \int_{V_1}^{V_2} p dV. \quad (15)$$

Characteristic Equation.

$$pV = mR\theta. \quad (13)$$

Specific Heat.

$$c_p = \left(\frac{dq}{dt} \right)_p \text{ and } c_v = \left(\frac{dq}{dt} \right)_v \quad (14)$$

$$\frac{c_p}{c_v} = \kappa. \quad (\text{Section 18})$$

$$c_p - c_v = AR. \quad (39)$$

General Heat Equations.

$$\Delta = AW. \quad (1)$$

$$d\Delta = \theta d\phi. \quad (29)$$

$$d\Omega = A(dS + dI + dW) = A(dE + dW). \quad (4)$$

$$dq = c_p \theta \frac{dv}{v} + c_v \theta \frac{dp}{p}. \quad (33)$$

$$dq = c_v d\theta + (c_p - c_v) \theta \frac{dv}{v}. \quad (35)$$

$$dq = c_p d\theta - (c_p - c_v) \theta \frac{dp}{p}. \quad (34)$$

Isothermal Transformation.

$$p_1 V_1 = p_2 V_2. \quad (5)$$

$$W = p_1 V_1 \log_e \frac{V_2}{V_1} = p_1 V_1 \log_e \frac{p_1}{p_2} = m R \theta_1 \log_e \frac{V_2}{V_1}. \quad (18)$$

Adiabatic Transformation.

$$p_1 V_1^\kappa = p_2 V_2^\kappa. \quad (22)$$

$$\frac{p_1}{p_2} = \left(\frac{V_2}{V_1} \right)^\kappa = \left(\frac{\theta_1}{\theta_2} \right)^{\kappa-1}. \quad (32)$$

$$\frac{V_1}{V_2} = \left(\frac{p_2}{p_1} \right)^{\frac{1}{\kappa}} = \left(\frac{\theta_2}{\theta_1} \right)^{\frac{1}{\kappa-1}}. \quad (31)$$

$$W = \frac{p_1 V_1}{\kappa - 1} \left[1 - \left(\frac{V_1}{V_2} \right)^{\kappa-1} \right] = \frac{p_1 V_1 - p_2 V_2}{\kappa - 1}. \quad (19)$$

Intrinsic Energy.

$$E_1 = \frac{p_1 V_1}{\kappa - 1}. \quad (40)$$

$$E_2 - E_1 = \frac{p_2 V_2 - p_1 V_1}{\kappa - 1}. \quad (42)$$

Entropy.

$$d\phi = \frac{d\Omega}{\theta}. \quad (27)$$

$$\phi_2 - \phi_1 = c_v \log_e \frac{p_2}{p_1} + c_p \log_e \frac{V_2}{V_1}. \quad (36)$$

$$\phi_2 - \phi_1 = c_v \log_e \frac{\theta_2}{\theta_1} + (c_p - c_v) \log_e \frac{V_2}{V_1}. \quad (37)$$

$$\phi_2 - \phi_1 = c_p \log_e \frac{\theta_2}{\theta_1} - (c_p - c_v) \log_e \frac{p_2}{p_1}. \quad (38)$$

Carnot Cycle.

$$\eta = \frac{\Omega_1 - \Omega_2}{\Omega_1} = \frac{\theta_1 - \theta_2}{\theta_1}. \quad (23, 25)$$

PROBLEMS AND SOLUTIONS: GASES

1. *Use of the characteristic equation.* A gas tank contains 1.5 pounds of oxygen at a temperature of 65° Fahrenheit and a gauge pressure of 50.3 pounds per square inch. The atmospheric pressure is standard atmospheric pressure. Find how much more oxygen the tank can contain and not exceed its safe pressure limit of 250 pounds gauge, if the temperature is liable to rise to 100° .

Solution. A gauge always reads the difference between the pressure which it is to measure and the atmospheric pressure. Normal or standard atmospheric condition is a pressure of 14.7 pounds per square inch.

First, find the volume of the tank from eq. (13) and Table 2,

$$p = (50.3 + 14.7)(144), \quad \theta = 459.5 + 65 = 524.5, \quad m = 1.5, \quad R = 52.56.$$

$$\therefore V = \frac{(1.5)(52.56)(524.5)}{(65)(144)} = 4.42 \text{ cubic feet.}$$

Then, find the weight of oxygen which the tank can contain under the conditions

$$p = (250 + 14.7)(144), \quad \theta = 459.5 + 100 = 559.5,$$

$$m = \frac{(264.7)(144)(4.42)}{(52.56)(559.5)} = 5.73 \text{ pounds.}$$

The added weight is then $5.73 - 1.5 = 4.23$ pounds.

2. *Carnot cycle.* A Carnot engine working between 350° Fahrenheit and 75° does 116,700 foot pounds of work per cycle. Find how much heat is supplied and rejected per cycle.

$$\text{Solution.} \quad \theta_1 = 459.5 + 350 = 809.5, \quad \theta_2 = 459.5 + 75 = 534.5,$$

$$\eta = \frac{\theta_1 - \theta_2}{\theta_1} = \frac{809.5 - 534.5}{809.5} = 0.34.$$

$$W = 116,700 \text{ foot pounds,} \quad AW = \frac{1}{1-\eta} (116,700) = 150 \text{ B. t. u.}$$

$$\eta Q_1 = AW. \quad \therefore Q_1 = \frac{150}{0.34} = 442 \text{ B. t. u.}$$

$$Q_1 - Q_2 = AW. \quad \therefore Q_2 = 442 - 150 = 292 \text{ B. t. u.}$$

3. *Intrinsic energy.* What is the energy of 2 pounds of air at 150 pounds per square inch pressure, absolute, if it occupies 6 cubic feet?

$$\text{Solution.} \quad p = (150)(144), \quad V = 6, \quad m = 2, \quad \kappa = 1.40.$$

$$\text{By eq. (40),} \quad E = \frac{pV}{\kappa - 1}. \quad \therefore E = \frac{(150)(144)(6)}{1.40 - 1} = 324,000 \text{ foot pounds.}$$

Alternative solution. Find the temperature θ , and assuming that the value of c_p holds from the absolute zero to this temperature, calculate the number of thermal units to raise the temperature of 2 pounds of air from zero to θ . Then reduce thermal units to mechanical energy units.

By eq. (13),

$$\theta = \frac{(150)(144)(6)}{(53.35)(2)} = 1215,$$

$$E = mc_v\theta = (2)(0.169)(1215) = 411 \text{ B. t. u.}$$

$$\therefore E = (411)(778) = 320,000 \text{ foot pounds.}^1$$

4. *Constant pressure transformation.* If 2 pounds of air are heated at constant pressure from 60° Fahrenheit to 200° find how much external work is done and how much the intrinsic energy is increased. How much heat is required?

Solution. $c_p = 0.2375$, $c_p - c_v = AR$, $R = 53.35$, $m = 2$, $t_1 = 60$, $t_2 = 200$. The heat required at constant pressure is $c_p m(t_2 - t_1) = (0.2375)(2)(200 - 60) = 66.5 \text{ B. t. u.}$

External work done is $(c_p - c_v)(m)(t_2 - t_1) = AR(2)(200 - 60) = \frac{53.35}{778}(2)(140) = 19.2$.

Increase in intrinsic energy is $\dots\dots\dots 47.3 \text{ B. t. u.}$
Or, expressed in mechanical units, it is $(47.3)(778) = 36,800 \text{ ft. lb.}$

5. *Exponent n.* Given 2 pounds of air at a pressure of 14.7 pounds absolute which occupy a volume of 24.78 cubic feet. If the air expands until the pressure is $0.4 p_1$ and the volume is $V_2 = 2 V_1$, find the exponent n in the equation $p V^n = p_1 V_1^n$, which may be taken to represent the expansion.

Solution. $p_1 V_1^n = p_2 V_2^n$, where $p_2 = 0.4 p_1$ and $V_2 = 2 V_1$.

Taking the logarithm to the base 10, we have

$$\log p_1 + n(\log V_1) = \log p_2 + n(\log V_2).$$

$$\therefore n = \frac{\log p_1 - \log p_2}{\log V_2 - \log V_1} = \frac{\log 2.5}{\log 2} = \frac{0.3979}{0.3010} = 1.32.$$

6. *Entropy changes.* Given 2 pounds of air at 14.7 pounds absolute pressure and 32° Fahrenheit, find and plot the following entropy changes.

(a) constant pressure, temperature increases to 736° absolute, Fahrenheit.

(b) constant volume, temperature increases to 736° absolute.

(c) constant temperature, pressure decreases to $(\frac{2}{3})$ (14.7) pounds.

(d) what is the entropy change in Problem 5?

Solution. By eq. (38),

$$(a) \quad \phi_2 - \phi_1 = c_p \log_e \frac{\theta_2}{\theta_1} = (0.237) \left[2.303 \log \frac{736}{491} \right] = 0.0961.$$

Since this change in entropy is per pound, the total change is 0.192.

$$(b) \text{ By eq. (37), } \phi_2 - \phi_1 = c_v \log_e \frac{\theta_2}{\theta_1} = (0.169) \left[2.303 \log \frac{736}{491} \right] = 0.0684.$$

The total change in two pounds of air is then 0.137.

¹ For all of the problems of this book slide rule results are sufficiently accurate. The slide rule has been used in all numerical work connected with these problems.

(c) By eq. (38), $\phi_2 - \phi_1 = -(c_p - c_v) \log_e \frac{p_2}{p_1} = (0.237 - 0.169)[2.30 \log 1.5] = 0.0276$.

Or the change for 2 pounds is 0.0552.

These changes are referred to in Figure 14 by the marginal letters used above.

(d) By eq. (36), $\phi_2 - \phi_1 = c_p \log_e \frac{V_2}{V_1} + c_v \log \frac{p_2}{p_1} = 2.30 [0.237 \log 2 + 0.169 \log 0.4]$ or 0.0090, and hence for 2 pounds 0.018.

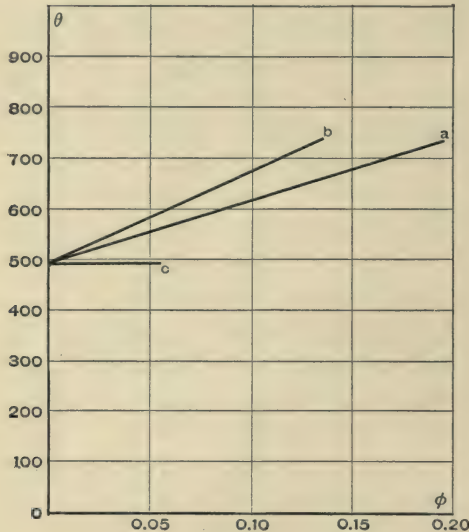


FIG. 14

7. *Isothermal expansion.* Three cubic feet of air at a pressure of 500 pounds absolute and a temperature of 450° Fahrenheit expand isothermally to a volume of 18 cubic feet. Find the heat received. Find the work done per pound of air.

Solution. $p_1 = (500)(144)$, $v_1 = 3$, $\theta_1 = 459.5 + 450 = 910$, $v_2 = 18$.

By eq. (5), $p_2 = \frac{p_1 V_1}{V_2} = \frac{(500)(3)}{(18)} 144 = (83.3)(144)$.¹

By eq. (18), $W = (500)(144)(3) \left[2.30 \log \frac{18}{3} \right] = 387,000$ foot pounds.

$$\mathcal{Q} = AW = \frac{387,000}{778} = 498 \text{ B. t. u.}$$

By eq. (13), $m = \frac{(500)(144)(3)}{(53.35)(910)} = 4.46$ pounds.

Therefore work per pound is 87,000 foot pounds.

8. *Adiabatic expansion.* If the air in Problem 7 after reaching the volume of 18 cubic feet is allowed to expand adiabatically until

¹ It is convenient to solve here for p_2 , which will be used in Problem 9.

the temperature is 200° Fahrenheit, what is the amount of work done per pound of air?

Solution. Designate the state after adiabatic expansion by the subscript 3.

$$p_2 = (83.3)(144), \quad V_2 = 18, \quad \theta_2 = 910, \quad \theta_3 = 459.5 + 200 = 660.$$

$$\text{By eq. (32),} \quad \frac{V_3}{V_2} = \left(\frac{\theta_3}{\theta_2} \right)^{\frac{-1}{\kappa-1}}. \quad \therefore V_3 = \left(\frac{910}{660} \right)^{\frac{1}{0.4}} (18) = 40.2.$$

$$\text{By eq. (19),} \quad W = \frac{p_2 V_2}{\kappa - 1} \left[1 - \left(\frac{V_2}{V_3} \right)^{\kappa-1} \right] = \frac{p_2 V_2}{\kappa - 1} \left[1 - \frac{\theta_3}{\theta_2} \right].$$

$$\text{Therefore} \quad W = 18 \frac{(83.3)(144)}{1.4 - 1} \left[1 - \frac{660}{910} \right] = 148,500 \text{ foot pounds.}$$

$$\text{The work per pound is} \quad \frac{148,500}{4.46} = 33,300 \text{ foot pounds.}$$

9. *Simultaneous equations in p and v.* If the expansions of Problems 7 and 8 are two transformations in a Carnot cycle, find the p and v coordinates of the four intersections of isothermals and adiabatics. What is the efficiency? Plot the cycle.

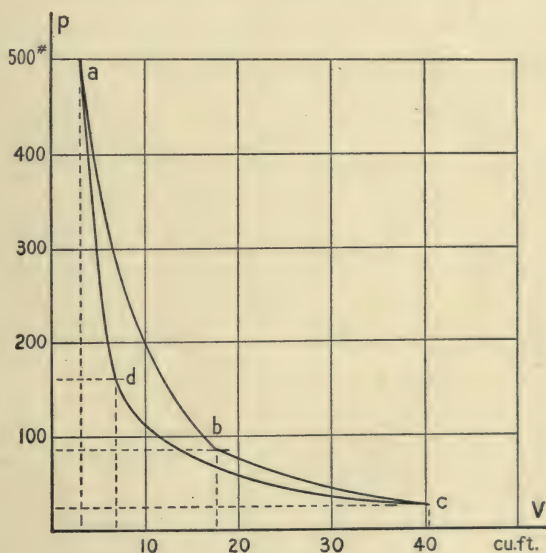


FIG. 15

Solution. From Problems 7 and 8,

$p_1 = 500$	$V_1 = 3$	$\theta_1 = 910$
$p_2 = 83.3$	$V_2 = 18$	$\theta_2 = \theta_1$
	$V_3 = 40.2$	$\theta_3 = 660$

Find p_3 by eq. (22).

$$p_3 = \frac{V_2^\kappa}{V_3^\kappa} p_2 = \left(\frac{18}{40.2} \right)^{1.40} (83.3) = 27.1 \text{ pounds per square inch.}$$

Find V_4 by solving simultaneously the equations for the adiabatic through state point 1 and the isothermal through state point 3, thus

$$p_4 V_4^\kappa = p_1 V_1^\kappa \text{ and } p_3 V_3 = p_4 V_4.$$

Therefore
$$V_4 = \left(\frac{p_1 V_1^\kappa}{p_3 V_3} \right)^{\frac{1}{\kappa-1}} = \left(\frac{(500)(144)(3)^{1.40}}{(27.1)(144)(40.2)} \right)^{\frac{1}{0.4}} = 6.7,$$

and
$$p_4 = \frac{p_3 V_3}{V_4} = \frac{(27.1)(144)(40.2)}{(6.7)} = (163)(144), \text{ or } 163 \text{ pounds per square inch.}$$

The values of p , V , and θ not given in Problems 7 and 8 are then

$$p_3 = 27.8$$

$$p_4 = 163$$

$$V_4 = 6.7$$

$$\theta_4 = \theta_3$$

The efficiency, by eq. (25), is
$$\eta = \frac{910 - 660}{910} = 27.5 \text{ per cent.}$$

The plot of this cycle is seen in Figure 15.

10. Pressure-volume plot. A quantity of air occupies 8 cubic feet at 70° Fahrenheit and under an atmospheric pressure of 15 pounds. It is compressed adiabatically to a volume of 2 cubic feet, cooled at constant volume to its original temperature, and allowed to expand adiabatically to atmospheric pressure. It then heats at constant pressure to its initial temperature. Draw the pressure-volume plot of this cycle.

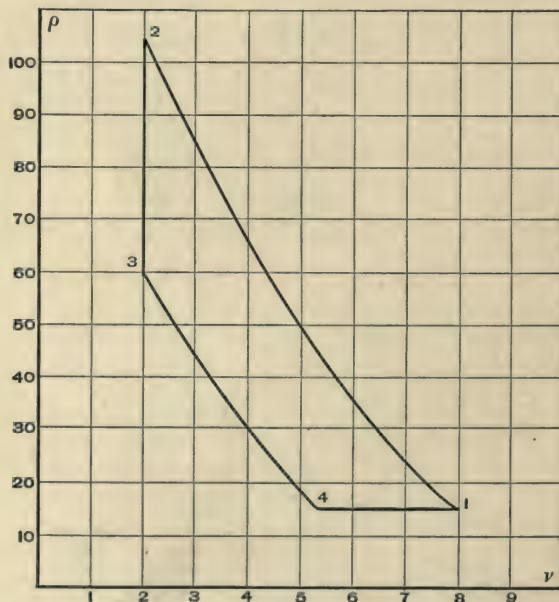


FIG. 16

Solution. The p - v plot is shown in Figure 16, which has been drawn by the use of the values given below.

Given $p_1 V_1^\kappa = p_2 V_2^\kappa$, where $p_1 = (15)(144)$, $V_1 = 8$, $\theta_1 = 460 + 70 = 530$, $V_2 = 2$, $\kappa = 1.40$; then

$$p_2 = \left(\frac{8}{2}\right)^{1.40} (15) = 104.6 \text{ pounds per square inch.}$$

By eq. (31), $\theta_2 = \left(\frac{V_1}{V_2}\right)^{\kappa-1} \theta_1 = \left(\frac{8}{2}\right)^{0.4} 530 = 922$.

Since $\theta_3 = 530$, $p_3 = \frac{\theta_3}{\theta_2} p_2 = \frac{530}{922} 104.6 = 60$.

During adiabatic expansion to atmospheric pressure

$$p V^\kappa = p_3 V_3^\kappa = p_4 V_4^\kappa \text{ and } p_4 = 15.$$

Hence

$$V_4 = \left(\frac{p_3}{p_4}\right)^{\frac{1}{\kappa}} V_3 = \left(\frac{60}{15}\right)^{0.71} (2) = 5.35,$$

and

$$\theta_4 = \left(\frac{2}{5.35}\right)^{0.4} (530) = 358^\circ \text{ absolute, or } -102^\circ \text{ F.}$$

11. *Regenerative cycle, Sterling engine.* The Sterling hot-air engine works upon the following cycle (as shown in the θ - ϕ plot of Fig. 17). From a to b the air in the engine expands isothermally at a temperature θ_1 , from b to c this air is cooled at essentially constant volume to the temperature θ_3 and pressure p_3 , from c to d it is then compressed practically isothermally to the volume v_1 and a pressure p_4 , from d to a the air is heated at constant volume to its initial condition.

This is accomplished by the use of a displacement piston D in conjunction with a working piston W , as is shown in the diagram of Figure 18. The displacement piston, actuated by a rocker arm, moves up its full stroke during da , while the working piston is essentially at rest. It

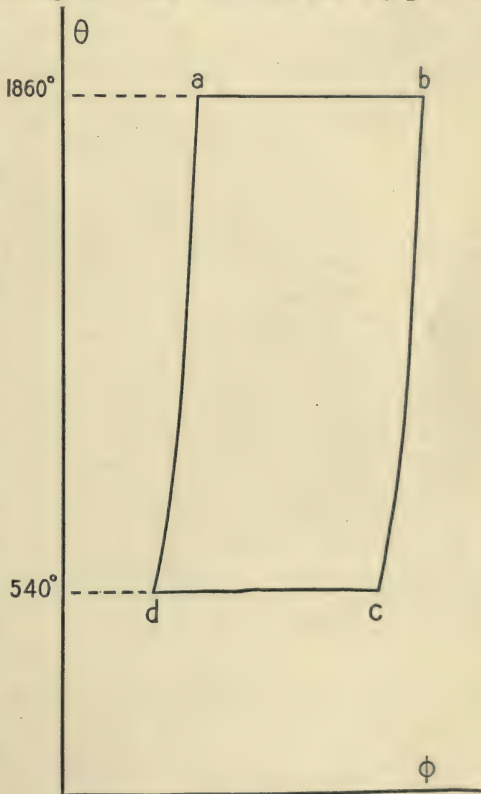


FIG. 17

thus forces all the cool air at temperature θ_3 through the regenerator R into the lower portion beneath which there is a fire F . The re-

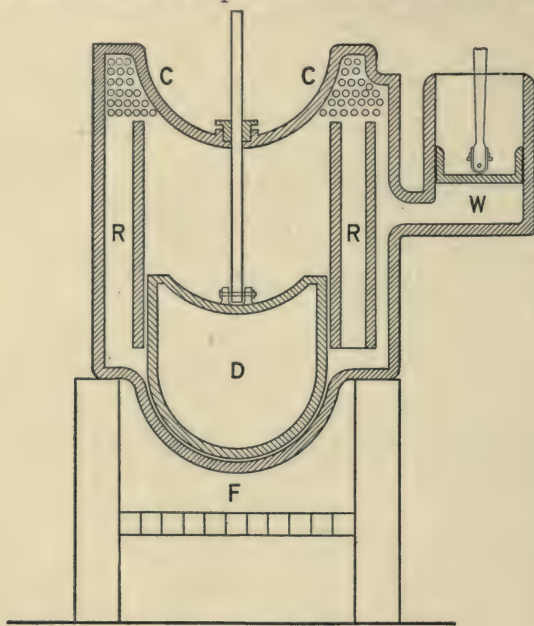


FIG. 18

generator consists merely of thin metal plates or wires which will rapidly absorb or reject heat. The air in contact now with the heating surface expands from a to b and causes a working stroke of the piston W , while the piston D remains at the top of its stroke. When W has reached the end of its stroke, the piston D moves quickly down, thus causing the air below it to pass through the regenerator. The air passing through the regenerator gives up

heat and then passes into the refrigerator C . The pressure is thereby reduced from b to c . The refrigerator consists of a set of pipes through which flows cooling water. The working piston W now returns, compressing essentially isothermally at the temperature of the refrigerator all the air originally contained by both cylinders. This last transformation is represented by cd .

Neglecting clearance, find the efficiency for such an engine working between 1400° and 80° Fahrenheit. Assume the volume of the working cylinder to be 0.9 cubic foot and the entire volume, exclusive of the working cylinder, to be 3 cubic feet. Compare with the Carnot engine in efficiency. Sketch the temperature-entropy plot and compare with the plot for the Carnot cycle. The lowest pressure is 16 pounds per square inch.

Solution. Denote by the subscripts 1, 2, 3, and 4 the state points a , b , c , and d , respectively.

$$\theta_1 = \theta_2 = 460 + 1400 = 1860$$

$$\theta_3 = \theta_4 = 460 + 80 = 540$$

$$V_1 = V_4 = 3$$

$$V_2 = V_3 = 3 + 0.9 = 3.9$$

$$p_3 = 16$$

Then

$$p_4 = \frac{p_3 V_3}{v_4} = \frac{3.9}{3} 16 = 20.8,$$

$$p_1 = \frac{\theta_1}{\theta_4} p_4 = \frac{1860}{540} 20.8 = 71.6,$$

$$p_2 = \frac{\theta_2}{\theta_3} p_3 = \frac{1860}{540} 16 = 55.1.$$

By eq. (18),

$$\mathfrak{Q}_{1-2} = \frac{p_1 V_1}{778} \log_e \frac{V_2}{V_1} = \frac{(71.6)(144)(3)}{(778)} \left[2.30 \log \frac{3.9}{3} \right] = 10.42 \text{ B. t. u.}$$

$\mathfrak{Q}_{2-3} = \mathfrak{Q}_{4-1}$, since they are constant volume changes between the same limiting temperatures.

$$\mathfrak{Q}_{3-4} = \frac{p_3 V_3}{778} \log_e \frac{V_4}{V_3} = \frac{(16)(144)(3.9)}{778} \left[2.30 \log \frac{3}{3.9} \right] = 3.03 \text{ B. t. u.}$$

The efficiency is then

$$\eta = \frac{10.42 - 3.03}{10.42} = 70.8 \text{ per cent.}$$

The Carnot efficiency is

$$\eta = \frac{1860 - 540}{1860} = 70.8 \text{ per cent.}$$

Discussion. This is an illustration of a regenerative cycle where, although the changes are at constant volume instead of isentropic, the efficiency is the same. There are other cycles possible which have the same efficiency as the Carnot for the same range of temperatures, but none has a higher efficiency, as is explained in Section 34.

12. Air compressor with and without clearance. An air compressor with perfect valves receives air at the atmospheric pressure of 15 pounds per square inch and delivers it at a constant pressure of 90 pounds per square inch. The compression follows

the equation pV^n equal to a constant where n is 1.3 due to partial cooling. Draw the p - v plot and find how many foot pounds of

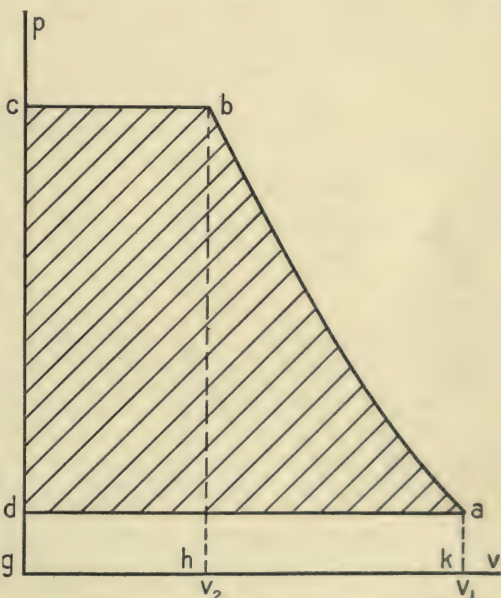


FIG. 19

work are required for each cubic foot of air at the atmospheric conditions under which the condenser is working. (a) Neglect clearance. (b) Allow for a

clearance of 11.1 per cent of the piston displacement.

Solution. Representing the total piston displacement and clearance by V_1 , the volume after compression by V_2 , the clearance volume by V_4 , and the volume occupied by the clearance air at atmospheric pressure by V_3 , Figures 19 and 20 represent respectively the cycle and the work done upon the air for conditions (a) and (b) above. The calculations are tabulated below for these two cases. It is important to notice that the net work done on the air is only that done in compressing it above the 15 pounds at which it is received.

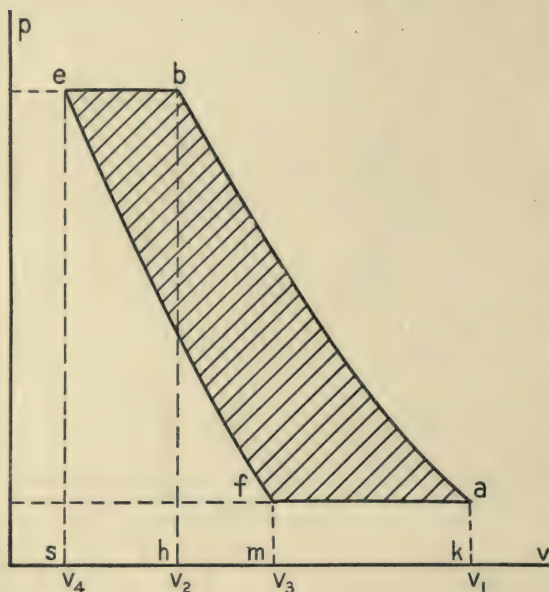


FIG. 20

	(a) Without clearance.	(b) With clearance.
Work of compression. By eq. (19),	area $abhk = 3650 V_1$	area $abhk = 3650 V_1$
Volume expelled at 90 lb. pressure, $V_2 = \left(\frac{15}{90}\right)^{\frac{1}{1.3}} V_1 = 0.251 V_1$		$V_2 - V_4$ $= (0.251 - 0.100) V_1$ $= 0.151 V_1$
Work of expulsion (90) (144) $(V_2 - V_4)$, area $besh = 3250 V_1$		area $besh = 1960 V_1$
Total work,	6900 V_1	5610 V_1
Work done by atmosphere, area $adgk = 2160 V_1$		area $afmk = 1300 V_1$
Work done by clearance air. ($V_3 = 0.397 V_1$), By eq. (15).		area $efms = 1460 V_1$
Therefore net work of compressor,	4740 V_1	2850 V_1
Cubic feet of air (at atmospheric pressure) delivered, V_1		0.603 V_1
Foot pounds of work per cubic foot delivered,	4740	4740

Discussion. The effect of the clearance in a *frictionless* compressor is not to increase the work per cubic foot of air delivered, but to require a larger piston displacement to deliver the same amount of air per stroke.

13. *Otto gas engine cycle.* The Otto, or de Rochas, cycle for a gas engine consists thermodynamically, in the ideal case, of the following processes: (1) adiabatic compression of an explosive mixture

of air and gas; (2) heating at constant volume by combustion; (3) adiabatic expansion of the products of combustion to the original volume occupied by its constituents; (4) cooling at constant volume.

This cycle will be discussed upon the following assumptions: (1) the specific volumes of the constituents will be considered equal to that of the products of combustion under the same conditions of temperature and pressure; (2) the products of combustion are considered to follow the laws of perfect gases and to have specific heats in the same ratio; (3) the density of the gases before and after combustion are considered equal to that of dry air under the same conditions of pressure and temperature.

The p - v plot for such a cycle is shown in Figure 21. The piston makes four strokes for each working stroke or explosion. During the forward stroke ab the piston draws into the cylinder a charge of gas and air. From b to c it compresses. From c to d the heating takes place instantaneously, due to the explosion of the mixture. During the forward stroke de external work is done. The valve then opens at e and the gases exhaust, the pressure and temperature dropping to that of the atmosphere or the "muffler." In effect it is as if the valves had not opened but the gas had been cooled at constant volume as represented by the line eb . The fourth stroke now expels this gas against atmospheric pressure and is represented by ba .

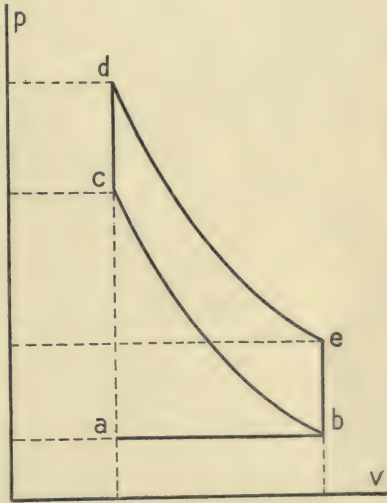


FIG. 21

(a) If the clearance is 25 per cent of the piston displacement, find the theoretical thermal efficiency.

(b) If a mixture of gas and air which will develop 85 B. t. u. by combustion is taken into the cylinder at a pressure of 14 pounds and a temperature of 60° Fahrenheit, find the temperatures at c and d and the pressures at c , d , and e .

(c) Compare the efficiency with the Carnot efficiency for the range of temperatures found from (b).

Solution. (a) All the heat taken in is along cd in Figure 21. Similarly, the heat rejected is along eb . The calculation will be made for one pound of the explosive mixture.

The efficiency is
$$\eta = \frac{q_1 - q_2}{q_1},$$

where

$$q_1 = c_v(\theta_d - \theta_c) \text{ and } q_2 = c_v(\theta_e - \theta_b).$$

Hence

$$\eta = \frac{c_v(\theta_d - \theta_c) - c_v(\theta_e - \theta_b)}{c_v(\theta_d - \theta_c)} = 1 - \frac{\theta_e - \theta_b}{\theta_d - \theta_c}.$$

Since de and cb are adiabatics, they must differ by a constant quantity of entropy. Equating then the difference in entropy found by eq. (37) between c and d to that found between e and b , gives

$$c_v \log \frac{\theta_e}{\theta_b} = c_v \log \frac{\theta_d}{\theta_c}, \text{ hence } \frac{\theta_e}{\theta_b} = \frac{\theta_d}{\theta_c}.$$

Therefore
$$\frac{\theta_e - \theta_b}{\theta_d - \theta_c} = \frac{\theta_b}{\theta_c}.$$

But since b and c lie on the same adiabatic, eq. (32) gives $\frac{\theta_b}{\theta_c} = \left(\frac{v_c}{v_b}\right)^{\kappa-1}$. Now v_c is clearance and v_b is piston displacement plus clearance. Hence

$$\eta = 1 - \frac{\theta_b}{\theta_c} = 1 - \left[\frac{\text{clearance}}{\text{clearance} + \text{piston displacement}} \right]^{\kappa-1}.$$

Substituting for the clearance 25 per cent of the piston displacement gives

$$\eta = 1 - \left[\frac{0.25}{1 + 0.25} \right]^{1.4-1} = 1 - 0.525 = 47.5 \text{ per cent.}$$

(b) $p_a = p_b = 14$. $\theta_b = 60 + 460 = 520$. $q_1 = 85 \text{ B. t. u.}$

$$\theta_c = \theta_b \left(\frac{v_b}{v_c} \right)^{\kappa-1} = 520 \left(\frac{1}{0.525} \right) = 990.$$

$$q_1 = c_v(\theta_d - \theta_c) = 0.169(\theta_d - 990) = 85 \text{ B. t. u.}$$

Therefore $\theta_d = 1490$.

By eq. (32), $p_c = \left(\frac{v_b}{v_c} \right)^{\kappa} p_b = (9.52)(14) = 133$.

Then $\frac{p_d}{p_c} = \frac{\theta_d}{\theta_c}$; therefore $p_d = \frac{1490}{990} 133 = 201$.

By eq. (32), $p_e = \left(\frac{v_d}{v_c} \right)^{\kappa} p_d = \frac{201}{9.52} = 21$.

(c) The Carnot efficiency is

$$\eta = \frac{\theta_d - \theta_b}{\theta_d} = \frac{1490 - 520}{1490} = 65 \text{ per cent.}$$

14. Diesel internal combustion cycle. Gas engines operating upon the Diesel cycle compress air to a high temperature, as shown by bc in Figure 22. The temperature is sufficient to ignite the fuel, which is injected from c to d at a rate such that cd is essentially an isothermal. Considering the initial compression to be adiabatic and from 15 pounds atmospheric pressure and 60° Fahrenheit to 500

pounds gauge, the isothermal to be for 15 per cent of the working stroke, and the remainder of the expansion to be adiabatic, solve for the following: (1) the temperature of the isothermal, (2) the clearance in per cent of the piston displacement, (3) the B.t.u. supplied by the combustion per pound of working substance, (4) the thermal efficiency, and (5) the Carnot efficiency for the same range of temperatures.

Solution.

$$(1) \quad \theta_c = \left(\frac{p_c}{p_b}\right)^{\frac{\kappa-1}{\kappa}} \theta_b = \left(\frac{515}{15}\right)^{\frac{0.40}{1.40}} \quad (520) \\ = 1430^\circ \text{ abs. or } 970^\circ \text{ F.}$$

$$(2) \quad \frac{v_c}{v_b} = \left(\frac{p_b}{p_c}\right)^{\frac{1}{\kappa}}; \text{ therefore } v_c = 0.08 v_b,$$

or $v_c = 0.087 (v_b - v_c)$.

That is, the clearance is 8.7 per cent of the piston displacement.

$$(3) \text{ Since } v_d = 0.15(v_b - v_c) + v_c = 2.73 v_c,$$

the heat supplied along the isothermal is per pound of mixture

$$q_1 = \frac{p_c v_c}{778} \log_e \frac{v_d}{v_c} = \frac{R \theta_c}{778} \log_e \frac{v_d}{v_c} = \frac{(53.35)(1430)}{778} = 98 \text{ B. t. u.}$$

$$\text{Since } \theta_e = \left(\frac{v_d}{v_e}\right)^{\kappa-1} \theta_d = \left[\frac{(2.73)(0.08 v_b)}{v_b}\right]^{0.40} 1430 = 780,$$

the heat rejected at constant volume is

$$q_2 = 0.169(780 - 520) = 44 \text{ B. t. u.}$$

$$(4) \text{ Therefore the efficiency } \eta = \frac{98 - 44}{98} = 55 \text{ per cent.}$$

$$(5) \text{ The Carnot efficiency is } \eta = \frac{1430 - 520}{1430} = 64 \text{ per cent.}$$

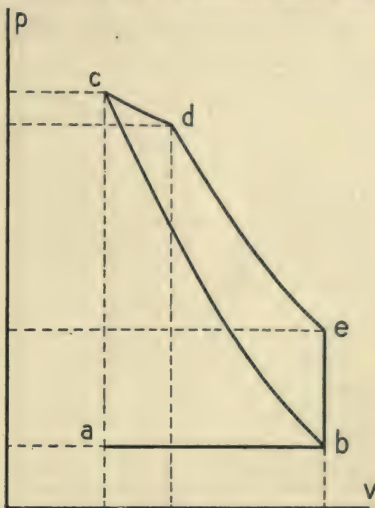


FIG. 22

15. *Air refrigeration.* The fact that air falls in temperature during an adiabatic expansion is made use of in the design of air refrigerating machines. The essential features of such a machine are shown in Figure 23. The air is received by the working cylinder *W* from the room which is to be kept cool. Let its temperature and pressure be θ_a and p_a respectively. The air is compressed adia-

batically as shown in the p - v plot of Figure 24. It is then cooled at constant pressure by passing it through a coil placed in a tank supplied with cooling water.

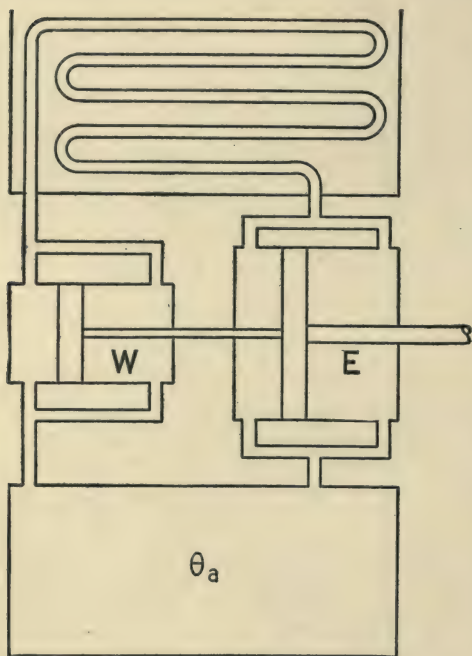


FIG. 23

After this it expands adiabatically along cd as shown in the plot, and does work in the expansion cylinder E , which returns it to the prime mover, which is driving the compressor. The temperature of the air is now considerably below that of the room from which it was received. It then enters the room and is heated at constant pressure.

If p_a is 14.7 pounds absolute, p_b is 14 times p_a , θ_a is to be zero degrees Fahrenheit, how much heat is abstracted from each pound of air while it is in the cooling room? Sketch the p - v plot for each cylinder and the θ - ϕ

plot for the entire cycle. Neglect the clearance. Assume that the cooling water reduces the temperature of the air to 95° Fahrenheit.

Solution.

$p_b = 14 p_a = 206$ pounds per square inch.

Calling v_a equal to v , we find v_b and θ_b by eq. (32) and (19) as follows:

$$v_b = \left(\frac{p_a}{p_b} \right)^{\frac{1}{\kappa}} v_a = \left(\frac{1}{14} \right)^{0.71} v = 0.154 v,$$

$$\theta_b = \left(\frac{p_b}{p_a} \right)^{\frac{\kappa-1}{\kappa}} \theta_a = (14)^{0.20} (460) = 990.$$

These values are plotted in Figure 25, which shows the work of the working cylinder W .

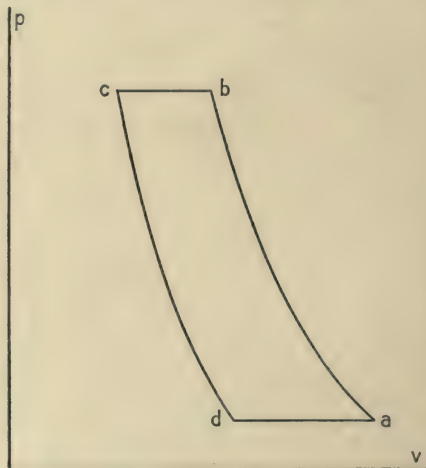


FIG. 24

Similarly, since $\theta_c = 95 + 460 = 555$ and $p_c = p_b$ and $p_d = p_a$,

$$v_c = \frac{\theta_c}{\theta_b} v_b = \frac{555}{990} (0.154 v) = 0.086 v,$$

$$v_d = \left(\frac{p_c}{p_d} \right)^{\frac{1}{\kappa}} v_c = (14)^{0.71} (0.086 v) = 0.562 v,$$

$$\theta_d = \left(\frac{p_d}{p_c} \right)^{\frac{\kappa-1}{\kappa}} \theta_c = \left(\frac{1}{14} \right)^{0.29} (555) = 260.$$

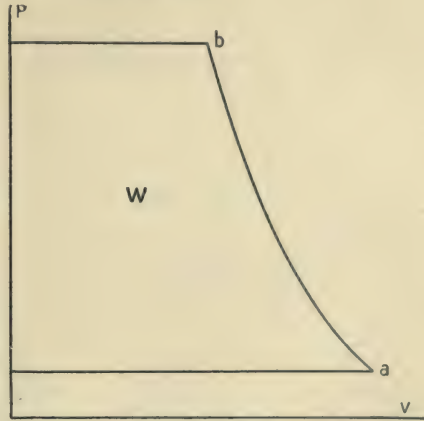


FIG. 25

These values are plotted in Figure 26, which shows the work done in the expansion cylinder *E*.

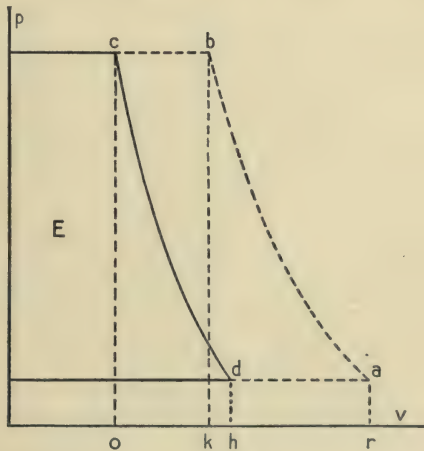


FIG. 26

The net input to the compressor is the dotted area *abcd* in Figure 26. This input will be found by parts as below:

$$\text{area } abkr = \frac{p_b v_b - p_a v_a}{\kappa - 1} = (144)(14.7) \frac{(14)(0.154) - 1}{0.40} v = 6100 v \text{ foot pounds.}$$

$$\text{area } bcok = p_b (v_b - v_c) = (206)(144)(0.068) v = 1990 v \text{ foot pounds.}$$

$$\text{area } cdho = \frac{p_c v_c - p_d v_d}{\kappa - 1} = (144)(14.7) \frac{(14)(0.086) - (0.562)}{0.40} v = 3400 v \text{ foot pounds.}$$

$$\text{area } darh = p_a (v_a - v_d) = (144)(14.7)(0.438) v = 930 v \text{ foot pounds.}$$

The net input is then 3760 foot pounds per cubic foot of entering air.

One pound of air at the admission temperature and pressure has a specific volume of

$$\frac{\theta_a}{491} 12.39 = 11.6 \text{ cubic feet.}$$

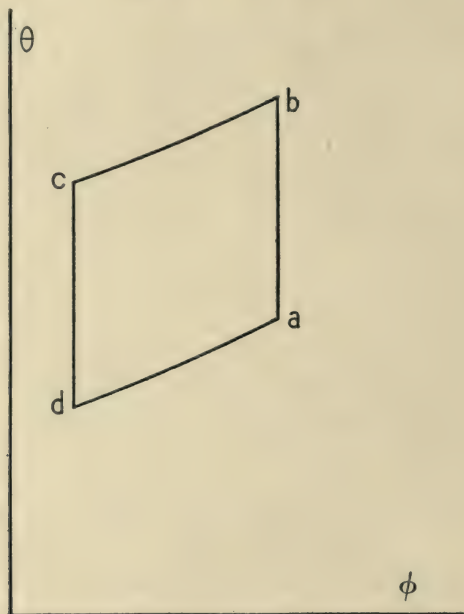


FIG. 27

The work of the compressor per pound of air pumped is then

$$(3760)(11.6) = 43,600 \text{ foot pounds.}$$

The B. t. u. equivalent of this work is

$$\frac{43,600}{778} = 56 \text{ B. t. u.}$$

The heat abstracted per pound by the room it is desired to refrigerate is

$$c_p(\theta_d - \theta_a) = 0.237(460 - 260) = 47.5 \text{ B. t. u.}$$

The cooling water must then remove the sum of these two amounts of heat or 103 B. t. u. As a check calculate the heat abstracted by the cooling water. It is evidently

$$c_p(\theta_b - \theta_c) = 0.237(990 - 555) = 103 \text{ B. t. u.}$$

The temperature-entropy plot is shown in Figure 27.

CHAPTER III

WATER AND ITS SATURATED VAPOR

44. Phenomena of Vaporization. In accordance with the kinetic theory the molecules of a liquid are considered to be in motion in a manner similar to the molecules of a gas, except for the fact that the average free path which a molecule travels between impacts is very much less in a liquid, and except for the fact that the molecules are now within the sphere of each other's influence and exert large mutual attractions. The continual evaporation which takes place from the surface of an exposed liquid demands, however, the generally accepted assumption, that occasional molecules possessing individually kinetic energies higher than the average for the entire body convert a portion of their kinetic energy into potential energy by breaking away from the surface attraction of the neighboring molecules and passing into the space above the liquid. That this free evaporation is facilitated by an increase of temperature, that is, by an increase of molecular kinetic energy, is a fact of common knowledge which supports this theory.

If the space above the liquid is enclosed, some of the molecules, which have escaped, and in that space are moving essentially after the manner of gaseous molecules, will in the course of their motion impinge upon the surface of the liquid and once more become a part of it. When, however, there are as many molecules entering the liquid in this way as leave it in the same time, there is a so-called "state of active equilibrium," and the vapor constituted by the free molecules is said to be saturated. If now the volume above the liquid, which these free molecules occupy, is reduced, there will be momentarily a larger density, and hence a larger number of impacts of the free molecules with the liquid, and hence a condensation of the vapor. The ultimate result will be a reduction of the number of free molecules until the density is reduced to its former value and the state of active equilibrium is again attained. Other things being equal, the density of a vapor in contact with its liquid is independent of the volume which the vapor is free to occupy.

On the other hand, if the temperature of the liquid is increased, the equilibrium will be momentarily disturbed by the escape from the liquid of a larger number of molecules than return to it in the same time. This momentary disturbance lasts until a new stable density of greater value is reached. The density of a saturated vapor in contact with its liquid is then some function of its temperature.

The pressure exerted by the vapor is also a function of the temperature, since by the kinetic theory the pressure depends upon the energy of the molecules and their number per unit volume. The number per unit volume is directly proportional to the density, and hence the pressure increases with a twofold rapidity as the temperature rises. The temperature at which the pressure of the saturated vapor would be equal to the external pressure upon the surface of a given body of liquid is called the boiling temperature of the liquid for that condition of pressure. At this temperature bubbles of the vapor start to form beneath the surface of the liquid, rise to the top, and burst into vapor. After boiling commences no further increase in the temperature (for constant pressure) is possible until all the liquid has been converted into vapor.

Water vapor, or steam, as it is usually called, is colorless and transparent. The cloudy white form it frequently takes is due to the condensation of part or all of the original steam into minute drops of water. In this case the steam is called "wet" because of the mixture with it of water particles. Steam free from moisture, whether due to condensation, entrained water, or other causes, is called "dry."

The temperature and pressure of water vapor are always the same as the temperature and pressure of the water with which it is in contact.¹ If, however, the saturated vapor is removed from contact with its liquid, its temperature may be raised above the original boiling point. The steam is then said to be superheated. If the pressure of the superheated steam is kept constant, its volume increases with the temperature; and also, if its volume is kept constant, the pressure increases with increased temperature. In steam engineering practice the steam is superheated in its passage from the boiler to the engine as the result of passing through coils of pipe in the flue of the boiler or as the result of some similar arrangement. Its volume per pound, or specific volume, then increases, and this increase is cared for at once by the motion of the piston of the engine. The

¹ Except for differences beneath the surface of the liquid due to the hydrostatic pressure.

effect is the same as if the steam were entirely separated from the water from which it is produced.

45. Specific Heat of Water. In order to convert a pound of water into steam, it is necessary, first, to raise the temperature of the water to that corresponding to the boiling point for the given pressure, and, second, to vaporize this water. In each case the necessary heat may be expressed by an equation of the form $d\Omega = AdS + AdI + AdW$, as in Section 8. In raising the temperature of the water there is produced a small increase in its volume. For ordinary engineering purposes this is so small that the terms AdI and AdW may be neglected. This necessary increase in the heat of the liquid will be represented by the symbols $q_2 - q_1$, the significance of which will appear more fully later in this section.

This increase in the heat of the liquid may be found approximately by assuming the specific heat to be constant and of average value unity. Then the heat, $q_2 - q_1$, required to change one pound of water from a temperature of t_1 to the boiling temperature t_2 is expressed as $q_2 - q_1 = t_2 - t_1$. It is in general desirable to know this necessary heat more accurately, and this requires a knowledge of the specific heat of water at various temperatures. The curve of Figure 28 shows the variations of the value of the specific heat of

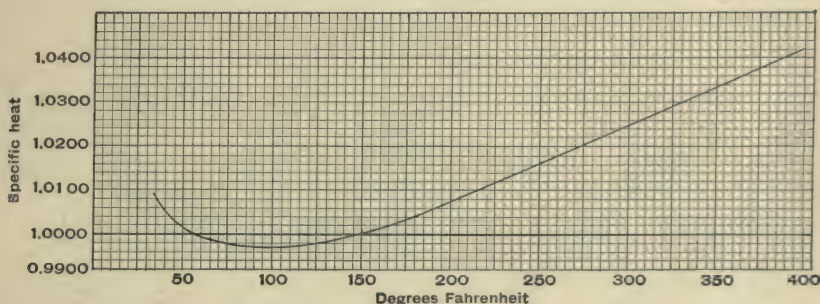


FIG. 28

water, taking the specific heat as unity for the change from 62° to 63° Fahrenheit. From such a curve, accurately constructed on a large scale, it would be possible to obtain the value of the specific heat for each degree and then by a summation of the form $\sum c dt$ to obtain $q_2 - q_1$.

It is more convenient, however, to use some set of tables, in which are tabulated the results of such a summation from 32° Fahrenheit, as a lower limit, to t° as an upper limit, where t has for values all

the whole degrees within the ordinary range of engineering work. The heat which is required to raise one pound of water from 32° to t_1° will be represented by q_1 . Similarly, the heat in B. t. u. required to raise one pound of water from 32° to t_2° is q_2 . The heat required to raise one pound of water from t_1 to t_2 is then $q_2 - q_1$ B. t. u. Such a table of values for q is illustrated in Table 3, on page 130, for a few chosen values. Certain other constants for steam, which will be discussed later, enter into this table.

The plot of Figure 29 shows the general nature of the relation between pressure and temperature for water vapor. It gives, in

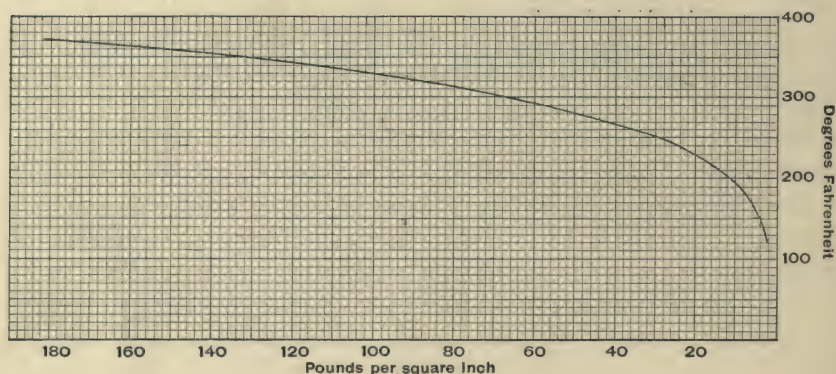


FIG. 29

other words, the boiling point temperature corresponding to a given pressure. There is to be noticed at the higher temperatures the rapid increase of pressure mentioned in Section 44. In Table 3, the value of p corresponding to any given temperature t is the pressure in pounds per square inch for which t is the boiling point.

46. Heat of Vaporization. The heat required to vaporize one pound of water may be considered in three parts. The first of these, namely, $q_2 - q_1$, which is required to raise the temperature of the water to the boiling point, has already been discussed. The remaining two have been considered partially in the illustration given on page 7. They are ρ , the internal latent heat required to separate the molecules against their mutual attractions, and Apu , the external latent heat or heat equivalent of the external work which must be done against the pressure p to cause the increase, u , in the volume. This increase u is $s - \sigma$, where s represents the specific volume of one pound of dry steam at the pressure p , and σ represents the volume of one pound of water.

The total heat required to raise one pound of water from 32° Fahrenheit to a temperature t and there to vaporize all of it is then

$$\lambda = q + \rho + A\mu, \quad (43)$$

where

$$u = s - \sigma. \quad (44)$$

The letter λ will be used hereafter as a general symbol for the total heat contents of a pound of water (in whatever form it may be) as measured by the heat in B. t. u. required to change the water from 32° Fahrenheit to the condition for which λ represents the heat contents. Also the letter r will be used to denote the total latent heat of vaporization. Thus

$$r = \rho + A\mu. \quad (45)$$

In Figure 30 is shown the general relation of q , ρ , $A\mu$, λ , and s for steam at various temperatures. It is to be noticed that the term $A\mu$, which is represented in the figure by the portion of the ordinate between the curve of ρ , the "internal latent heat," and r , the total "latent heat," is almost constant. Since the specific volume of a pound of water is always approximately 0.016 cubic foot, it may for many purposes be entirely neglected as compared with the specific volume s of dry steam. Hence the product $A\mu$ is practically equal to Aps . This relation between p and s has been expressed in empirical formulas for calculating the specific volume of dry steam for any pressure. Thus by Zeuner as

$$ps^{1.0646} = 479, \quad (46)$$

and by Rankine as

$$ps^{\frac{1}{3}} = 482. \quad (47)$$

These equations obviously make allowance for the fact that $A\mu$ is not exactly constant but increases somewhat at higher pressures.

The values for p , r , ρ , $A\mu$, s , and γ , the density or mass per cubic foot, for selected values of t are given in Table 3 on page 130.¹ It is to be remembered that these values of r , ρ , $A\mu$, as well as q , are all in B. t. u. per pound. The specific volume s is in cubic feet per pound. The pressure p is in pounds weight per square foot, but since pressures are usually measured by gauges graduated in pounds

¹ A convenient set of steam tables is the work of Professor Peabody of the Massachusetts Institute of Technology. It is published under the title "Steam and Entropy Tables" by Wiley and Sons. The values used in this text and included in Table 3 are taken from the 8th edition by the kind permission of the author and the publishers. The symbols used in Table 3 are those of Professor Peabody except in the case of entropy, which is considered later in Sections 49 and 50. The tables compiled by Professors Marks and Davis of Harvard University also embody in a most convenient form the results of the recent investigations (Longmans, 1908).

per square inch, the values of the table in the column headed by p are so expressed and must be multiplied by a factor of 144 before being used in any numerical calculations.

It is to be noticed also from Figure 30 that ρ , the internal latent heat of vaporization, decreases with the volume. This is in accord-

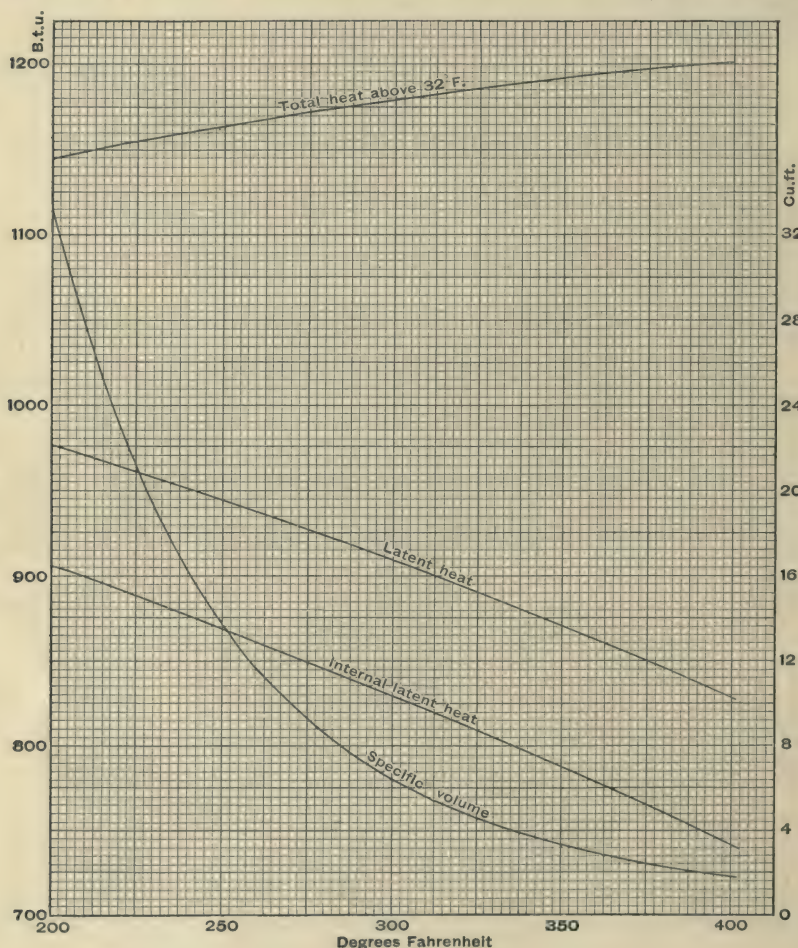


FIG. 30

ance with the kinetic theory, for when occupying a smaller volume the molecules are closer together and less work has been done in producing vaporization against their mutual attractions.

47. Quality of Steam. Consider a pound of water at 32° Fahrenheit, and the results of raising its temperature to t° and there evap-

orating against the corresponding pressure of p pounds, not the entire pound of water as the problem has previously been considered, but a fraction of a pound, say x . Then obviously the fundamental equations given below on the left hand must be restated as shown on the right hand.

$$\lambda = q + r = q + \rho + A p u. \dots (43) \quad \lambda = q + x r = q + x \rho + x A p u. \dots (48)$$

$$v = s = u + \sigma. \dots (44) \quad v = x s + (1 - x) \sigma = x u + \sigma. \dots (49)$$

In general, if from any cause the steam considered is not dry but has mixed with it water, and the mixture is x parts dry steam and $(1 - x)$ parts water, then x is defined as the "quality," and the modified equations given above must be used. The quality x is usually expressed in per cent, but obviously it must enter into these equations as a decimal. The amount of water present, $(1 - x)$ parts, is expressed as a per cent and called the "priming."

48. Intrinsic Energy of Steam. In the formation of one pound of dry steam at a pressure p , it is to be noted carefully that the molecules of the forming steam act as direct agents by which a certain amount of mechanical work, $p u$, is done at the expense of $A p u$ heat units supplied by the fire. At no time do these $p u$ units constitute part of the intrinsic energy of the steam. Therefore, in the equation $\lambda = q + \rho + A p u$, which is in the form of $\mathfrak{Q} = A(dS + dI + dW)$, the term $q + \rho$ represents the intrinsic energy of the steam and corresponds to $A dS + A dI$. Representing by ϵ the intrinsic energy of a pound of steam and water mixture, we have, if x is unity,

$$A \epsilon = q + \rho. \dots (50)$$

Or, more generally, for the intrinsic energy of one pound of a mixture of quality x ,

$$A \epsilon = q + x \rho. \dots (51)$$

49. Entropy of the Liquid. By definition, the entropy of a body in state 2 is the integral $\int_1^2 \frac{dq}{\theta}$, taken along any reversible path from state point 1, arbitrarily chosen as a zero of entropy. By convention for steam the point 1 is taken as the entropy of a pound of water at 32° Fahrenheit. The heating of water is a reversible process, and the integral representing the entropy of one pound, at any temperature t° Fahrenheit or θ° Fahrenheit on the absolute scale, may then be written as

$$\phi_w = \int_{\theta=491.5}^{\theta} \frac{cd\theta}{\theta}. \dots (52)$$

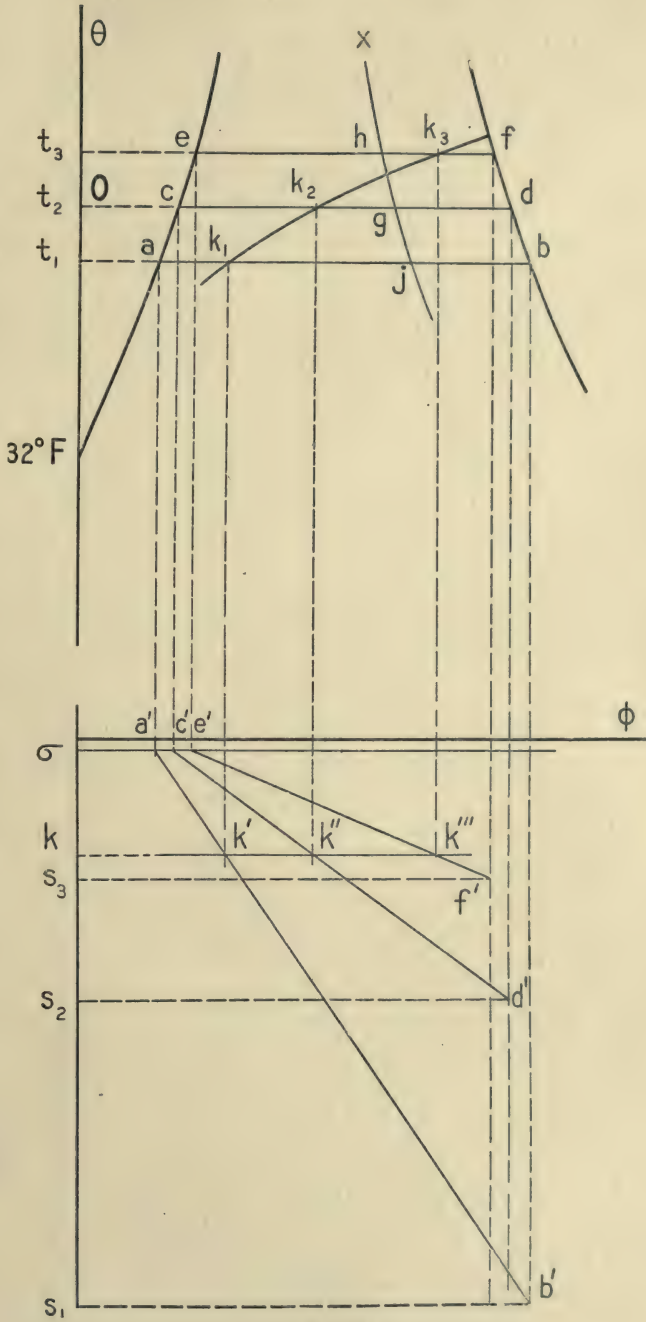


FIG. 31

of lesser value, in the study of the steam engine. Such a temperature-entropy chart is given in part on page 86.

The curve *Face* of Figure 31 is called the "water line," and the locus of the points like *b*, *d*, and *f*, is the "dry steam line" or "line of constant steam weight."

Since the entropy of a mixture is $\phi_v + x\phi_v$ and the length of the lines *ab*, *cd*, and *ef* represents values of ϕ_v , the entropy of any mixture may be easily determined from this chart. Thus if the entropy of a mixture of quality *x* at a temperature t_2 is desired, we proceed as follows. Divide the line *cd* into two parts by a point *g* so located that *cg/cd* shall equal *x*. Then obviously the length of the line *cg* is $x\phi_v$ and the total entropy of the mixture is represented by the line *Og*. For convenience in making such interpolations on the chart a series of lines like *hgg'* may be drawn, indicating all the points of the same quality for values of *x* differing by 10 per cent, as is done in Figure 32. (In the chart on page 86 such lines have been drawn for differences of one per cent.) These lines of constant quality are usually called "*x* lines."

Since the temperature is constant during vaporization, if the pressure is constant, the lines *ab*, *cd*, and so on represent constant pressure lines, the values of which correspond to the temperatures t_1 , t_2 , and so on.

Lines of constant volume showing the entropy of all states of a mixture of steam and water which occupy per pound the same volume may be drawn as follows. Continue the axis of temperatures below the entropy axis and let it represent to a convenient scale an axis of volumes. Draw a line parallel to the entropy axis and below it by an amount equal to σ , the specific volume of water. In Figure 31 the distance σ is made disproportionately large for clearness. The letters *a'*, *c'*, and *e'* refer to the intersections of the dotted vertical projection lines with this constant volume line through σ . Then for any pressure and temperature as t_1 project normal to the entropy axis the line *aa'* to intersect the constant volume line σ in *a'*, and the line *bb'* to intersect the line s_1b' , which is drawn through s_1 , the point on the axis of volumes which corresponds to the specific volume of dry steam at the temperature t_1 . The line *a'b'* shows then the increase of entropy with increasing volume due to vaporization at this temperature. In a similar manner, other lines, showing the relations of volume and entropy for the temperatures t_2 , t_3 , and so on, may be drawn. These are represented by *c'd'* and *e'f'*. For any volume as *k* the line of constant volume *kk'''* intersects the

lines $a'b'$, $e'd'$, and $e'f'$ in points k' , k'' , and k''' , the abscissas of which represent the entropy of mixtures of constant volume k but of different temperatures. The corresponding temperatures may be found by projecting the points k' , k'' , and k''' back onto the lines ab , cd , and ef , respectively. The ordinates of these projections, namely k_1 , k_2 , and k_3 , represent the temperatures for mixtures of volume k . The line k_1k_3 is then on the temperature-entropy plot a line of constant volume, and similarly, the lines $a'b'$, $e'd'$, and $e'f'$ on the volume-entropy plot are lines of constant pressure and temperature. On the chart of page 86 several constant volume lines are represented.

A family of curves for constant total heat contents might also be constructed through all the points for which $\lambda = q + xr = \text{constant}$, for different values of the parameter λ . These

lines of constant heat contents are of especial value in the solution by the use of a θ - ϕ chart of problems on the flow of steam through nozzles. There is unfortunately no graphical method of constructing constant heat content curves, and the calculation and plotting of separate points is necessarily tedious. Constant heat curves are shown in the chart on page 86. Similarly, curves for constant intrinsic energy might

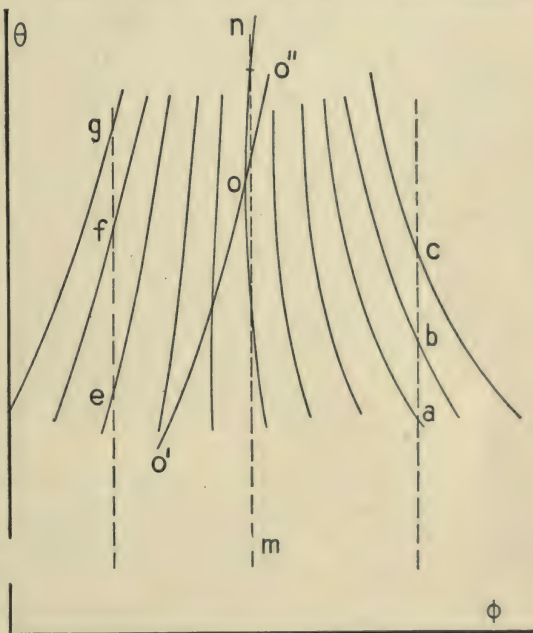


FIG. 32

be plotted in accordance with the relation $\epsilon = q + xp = \text{constant}$. The general appearance of these curves is similar to those for constant heat contents, but slightly different in slope.¹

52. Isentropic Transformation of a Mixture. From a study of the temperature-entropy diagram of Figure 32 it is evident that if an

¹ For a method of determining graphically the intrinsic energy corresponding to a desired point on the θ - ϕ chart, the reader is referred to Berry, "Temperature Entropy Diagram," 1st edition, pp. 28-29. For the converse problem there is no graphical solution.

isentropic compression abc be made, starting with a mixture the state of which is represented by a , as successive lines of smaller volume are intersected (volume lines are not shown in the figure), so also are met lines of greater quality or higher values of x . That is, an adiabatic compression of steam near its saturation point causes reëvaporation, the heat for which is obtained at the expense of the external work done during compression. When the point c is reached, further isentropic compression results in superheating the steam, as was first announced by Cazin. Conversely, starting from c , an isentropic expansion results in a condensation or lowering of the quality x , as was first stated by Hirn. That is, the external work is now done by the expanding steam at the expense of the heat of vaporization of a part of it.

From the same figure it will be noticed that the x lines for those qualities less than 0.5 in general slope to the left, while the slope is toward the right for those qualities greater than about 0.5. Hence for the isentropic line efg , when the mixture is near its liquid condition, it is evident that compression results in condensation, and, *vice versa*, expansion results in reëvaporation, as was stated in Section 44.

An isentropic line as mn of Figure 32 if produced will in general cut an x line in two points. Hence for each x line an isentropic may be drawn tangent to it at some point, as o . If the line $o'o''$ be drawn through all these possible points of tangency, its general appearance will be as shown in the figure. Such a line is called the "zero line" (because it represents states for which the specific heat may be shown to be zero).¹

The changes in the state of a mixture undergoing an isentropic compression may then be expressed in a more rigorous form by saying that for values of x greater than the zero line values there is a reëvaporation, and for values less than the zero line values there is a condensation. Conversely, for isentropic expansion there will be condensation (or reëvaporation) for values of x greater than (or less than) the zero line values.

Since for an isentropic change the external work performed by the steam is at the expense of the intrinsic energy, the work w per pound of a steam mixture undergoing a reversible adiabatic expansion from state 1 to state 2 is

$$w = \epsilon_1 - \epsilon_2 = q_1 + x_1\rho_1 - q_2 - x_2\rho_2. \quad . \quad . \quad . \quad (56)$$

¹ Berry, "Temperature Entropy Diagram," pp. 25-26.

53. Specific Heat of Steam. The specific heat of *dry* steam may be found for any of the following conditions: (*a*) superheating under constant pressure, (*b*) superheating at a constant volume, (*c*) for changes of both pressure and volume so related that the whole mass remains dry saturated steam, that is, for changes such that the temperature and entropy are related as in the dry steam line *fb* of Figure 31, page 69. For constant pressure the value of 0.48 may be used for temperatures near that of saturation; for higher temperatures, values are given in the Table on page 93. For constant volume, Regnault gave 0.346, but this determination is not entirely reliable.

For condition (*c*) the specific heat is negative for all ordinary temperatures and pressures. The physical significance of this fact may best be shown by an example. Thus, suppose it is desired to raise the temperature of one pound of steam from 212° to 213° Fahrenheit in accordance with condition (*c*) above. This may be done by compressing the steam under a pressure which increases according to the relation plotted in Figure 29 of page 64. The specific volume decreases with this increased temperature and pressure. The external work done may be taken as approximately equal to the product of the average pressure and the decrease in volume. This is work done upon the steam by some external source. The change in total heat contents represents the amount of heat which must be added from some external source. The values of *p*, *q*, *s*, and *r* for 212° and for 213° Fahrenheit are given below.

212°	213°	212°	213°
$q = 180.3$	$q = 181.3$	$p = 14.70$	$p = 14.99$
$r = 969.7$	$r = 969.1$	$s = 26.78$	$s = 26.29$
$\lambda = 1150.0$	$\lambda = 1150.4$		

Hence, 0.4 B. t. u. must be supplied from some external source to change from 212° to 213°. But an amount of external work

$$\frac{14.70 + 14.99}{2} 144(26.78 - 26.29) = 1020 \text{ ft. lb.} = 1.35 \text{ B. t. u.}$$

has been done upon the steam. Now, to maintain the required conditions, external work to the value of 1.35 B. t. u. has been done, but only 0.4 B. t. u. are needed to change the steam from 212° to 213°. Hence, unless this excess of (1.35 - 0.4) obtained from the external work is to go into superheating, it must be subtracted.

This amount of heat per pound or 0.95 B. t. u. must therefore be conducted away, and hence the specific heat is negative.¹

54. Résumé of Equations for a Mixture of Steam and Water. The following equations refer to a pound of mixture, and are numbered as in the preceding sections.

Total Heat Contents.

$$\begin{aligned}\lambda &= q + r = q + \rho + Apu = A\epsilon + Apu \\ \text{or} \quad \lambda &= q + xr = q + xp + xApu = A\epsilon + xApu \quad . \quad . \quad (48)\end{aligned}$$

Total Entropy.

$$\phi = \phi_w + \phi_v = \int_{\theta=491.5}^{\theta} \frac{cd\theta}{\theta} + \frac{r}{\theta}, \quad \text{or } \phi = \phi_w + x\phi_v. \quad . \quad . \quad (55)$$

Total Intrinsic Energy.

$$A\epsilon = q + \rho, \quad \text{or } A\epsilon = q + xp. \quad . \quad . \quad . \quad (51)$$

Specific Volume. For water $\sigma = 0.016$. For steam (dry) s .

$$v = xs + (1 - x)\sigma = xu + \sigma. \quad . \quad . \quad . \quad (49)$$

Increase of Volume. $u = s - \sigma$, or xu (44)

Constant Heat Transformation.

$$\lambda_1 = \lambda_2, \quad \text{or } q_1 + x_1r_1 = q_2 + x_2r_2. \quad . \quad . \quad . \quad (48)$$

Constant Energy Transformation.

$$\epsilon_1 = \epsilon_2, \quad \text{or } q_1 + x_1\rho_1 = q_2 + x_2\rho_2. \quad . \quad . \quad . \quad (51)$$

Isothermal Transformation. } q , θ , and p are constant.
Constant Pressure. }

$$w = p(v_2 - v_1) = pu(x_2 - x_1).$$

Isentropic Transformation. }
Reversible Adiabatic. }

$$\phi_1 = \phi_2 = \phi_{w_1} + x_1\phi_{v_1} = \phi_{w_2} + x_2\phi_{v_2}. \quad . \quad . \quad . \quad (55)$$

$$w = \epsilon_2 - \epsilon_1 = \frac{1}{A}(q_2 - q_1 + x_2\rho_2 - x_1\rho_1). \quad . \quad . \quad . \quad (56)$$

Irreversible Adiabatic Change.

$$dq = 0, \text{ but } d\phi \neq 0.$$

¹ This method is of course unsatisfactory for the purpose of determining this specific heat because of the large errors introduced in a difference between two quantities where the difference is of a magnitude not much greater than the probable error in the quantities subtracted. For the method used by Clausius see Preston, "Heat," pp. 655-658. For 212° Fahrenheit Clausius gives 1.13. The method used above is selected because it is free from differential notation, and emphasizes the physical relations involved.

PROBLEMS AND SOLUTIONS: SATURATED WATER VAPOR

16. Quality. How much heat has been required to change one pound of water at 60° Fahrenheit into a mixture of steam and water occupying 2 cubic feet at a pressure of 130 pounds per square inch absolute?

Solution. From Table 3,

For 130 pounds

For 60° F.

$$q_2 = 318.6$$

$$q_1 = 28.1$$

$$v_2 = 872.1$$

$$s_2 = 3.451$$

$$\text{By eq. (49), } v_2 = 2 = x_2 s_2 + (1 - x_2) \sigma \text{ or } x_2 = \frac{2 - 0.016}{3.450 - 0.016} = 0.577.$$

$$\text{By eq. (48), } \lambda_2 - \lambda_1 = q_2 + x_2 r_2 - q_1 = 318.6 + (0.577)(872.1) - 28.1 = 793 \text{ B. t. u.}$$

17. Intrinsic energy. Find the intrinsic energy, volume, and entropy of a mixture of 5 pounds of steam and water which is 80 per cent steam, if the pressure is 120 pounds absolute per square inch. How much heat must be added to make the quality unity at this pressure?

Solution. From Table 3, for 120 pounds,

$$q = 312.3, \quad r = 876.9, \quad \rho = 794.2, \quad \phi_w = 0.4922, \quad \phi_v = 1.0951, \quad s = 3.723.$$

Given $x = 0.80$.

$$\text{By eq. (51), energy per pound} = 778 [312 + (0.80)(794.2)] = 736,000 \text{ foot pounds,}$$

$$\text{or } 3,680,000 \text{ foot pounds for 5 pounds.}$$

$$\text{By eq. (55), entropy per pound} = 0.4922 + (0.80)(1.0951) = 1.3682,$$

$$\text{or } 6.841 \text{ for 5 pounds.}$$

$$\text{By eq. (48), total heat per pound} = 312.3 + (0.80)(876.9) = 1013 \text{ B. t. u.,}$$

$$\text{or } 5065 \text{ B. t. u. for 5 pounds.}$$

$$\text{The total heat for } x = 1.00 \text{ is } 312.3 + 876.9 = 1189.2 \text{ B. t. u. per pound,}$$

$$\text{or } 5946 \text{ B. t. u. for 5 pounds.}$$

$$\text{Therefore the heat which must be added is } 5946 - 5065 = 881 \text{ B. t. u.}$$

By eq. (49),

$$\text{the volume per pound} = (0.80)(3.723) + (1 - 0.80)(0.016) = 2.99 \text{ cubic feet.}$$

18. Isoenergetic transformation. If three pounds of a mixture of steam and water at a pressure of 120 pounds absolute and a quality of 0.90 undergoes an isoenergetic change until the pressure is 150 pounds absolute, find the quality and the volume in the final condition.

Solution. From Table 3,

For 120 pounds

$$q_1 = 312.3$$

$$\rho_1 = 794.2$$

For 150 pounds

$$q_2 = 330.0$$

$$\rho_2 = 779.3$$

$$s_2 = 3.014$$

Given $x_1 = 0.90$.

By eq. (51),

$$q_1 + x_1\rho_1 = q_2 + x_2\rho_2.$$

Therefore

$$x_2 = \frac{312.3 + (0.90)(794.2) - 330.0}{779.3} = 0.893.$$

By eq. (49),

$$v_2 = x_2s_2 + (1 - x_2)\sigma.$$

Therefore

$$v_2 = (0.893)(3.014) + (1 - 0.893)\sigma \text{ per pound} \\ = 2.70 \text{ cubic feet per pound,}$$

or for 3 pounds the volume is 8.10 cubic feet.

19. Isothermal transformation. Three pounds of steam and water at 140 pounds absolute occupy 6 cubic feet. If the volume is increased at constant pressure to 9 cubic feet, find the heat added, the external work done, the final quality, and the increase in entropy.

Solution. The volume per pound is 2 cubic feet in state 1 and 3 cubic feet in state 2.

From Table 3, for 140 pounds,

$$\rho = 784.0, \quad A p u = 83.4, \quad \phi_v = 1.0675, \quad s_1 = 3.220.$$

By eq. (49),

$$v_1 = x_1s + (1 - x_1)\sigma.$$

Therefore

$$x_1 = \frac{2 - 0.016}{3.220 - 0.016} = 0.620.$$

Similarly,

$$x_2 = \frac{3 - 0.016}{3.220 - 0.016} = 0.932.$$

The heat added is

$$\lambda_2 - \lambda_1 = x_2r - x_1r = (x_2 - x_1)(\rho + A p u),$$

or

$$(0.932 - 0.620)(784 + 83.4) = 245 + 26 = 271 \text{ B. t. u. per pound.}$$

Of this added heat 26 B. t. u. represents the external work. For 3 pounds the added heat is 813 B. t. u. and the external work is

$$(3)(778)(26) = 60,400 \text{ foot pounds.}$$

The increase in entropy is $(x_2 - x_1)\phi_v = (0.932 - 0.620)(1.0675) = 0.333$ per pound, or 1.0 unit of entropy for 3 pounds.

20. Adiabatic transformation. Five pounds of steam of quality 0.80 expand adiabatically from 130 pounds to 30 pounds absolute pressure. Find the quality, the change in volume, and the external work done.

Solution.

For 130 pounds

$$q_1 = 318.6$$

$$\rho_1 = 789.0$$

$$\phi_{w_1} = 0.5000$$

$$\phi_{v_1} = 1.0808$$

$$s_1 = 3.451$$

For 30 pounds

$$q_2 = 219.1$$

$$\rho_2 = 868.2$$

$$s_2 = 13.74$$

$$\phi_{w_2} = 0.3687$$

$$\phi_{v_2} = 1.3305$$

$$s_2 = 13.74$$

Given $x_1 = 0.80$.

By eq. (55), for an adiabatic expansion,

$$\phi_{w_1} + x_1\phi_{v_1} = \phi_{w_2} + x_2\phi_{v_2}.$$

Therefore

$$x_2 = \frac{0.5000 + (0.80)(1.0808) - 0.3687}{1.3305} = 0.75.$$

By eq. (56), the work per pound is

$$\epsilon_1 - \epsilon_2 = \frac{1}{A}(q_1 + x_1\rho_1 - q_2 - x_2\rho_2),$$

or

$778 [318.6 + (0.80)(789) - 219.1 - (0.75)(868.2)] = 778(79.5) = 62,000$ ft. lb.,
or 310,000 foot pounds for 5 pounds of mixture.

By eq. (49),

$$v_1 = (0.80)(3.450) + (1 - 0.80)\sigma,$$

and

$$v_2 = (0.75)(13.74) + (1 - 0.75)\sigma.$$

The increase of volume per pound is (neglecting σ),

$$(0.75)(13.74) - (0.80)(3.45) = 7.63 \text{ cubic feet, or } 38.15 \text{ for 5 pounds.}$$

21. Constant volume transformation. A closed tank containing 10 pounds of dry steam at 130 pounds absolute is cooled until 90 per cent of the steam has been condensed. What is the final temperature and how much heat has been removed?

Solution. Since the change takes place at constant volume, no external work is done, and the heat removed is that which is required to reduce the intrinsic energy from state 1 to state 2.

For 130 pounds

$$q_1 = 318.6$$

$$\rho_1 = 789.0$$

$$s_1 = 3.451$$

Given $x_1 = 1.00$ and $x_2 = 0.10$. Solving for 1 pound,

By eq. (49),

$$v_1 = s_1 \text{ and } v_2 = x_2 s_2 + (1 - x_2)\sigma = 0.10 s_2 + (0.90)(0.016).$$

For constant volume,

$$v_1 = v_2,$$

hence

$$3.451 = 0.10 s_2 + 0.014, \quad \text{or } s_2 = 34.37.$$

From tables similar to Table 3, $s_2 = 34.37$ is found to be the specific volume of steam at 198.9° Fahrenheit and 11.3 pounds pressure.

For these conditions, $q_2 = 167.1$, $\rho_2 = 905.6$.

The heat removed is $\epsilon_1 - \epsilon_2$, where, by eq. (51),

$$\epsilon_1 = 318.6 + 789.0 \text{ and } \epsilon_2 = 167.1 + (0.10)(905.6);$$

that is, 850 B. t. u. per pound, or 8500 B. t. u. per 10 pounds.

22. Boiler explosion. How much energy is liberated if a boiler containing two tons of water at 353° Fahrenheit explodes? Assume the atmospheric pressure to be 14.7 pounds. Draw the θ - ϕ dia-

gram. What volume would the steam thus produced occupy if there was no condensation?

Solution.

At 353° F.
at 14.7 pounds,

$p = 140$ pounds,
 $q_2 = 180.3$.

$q_1 = 324.4$;

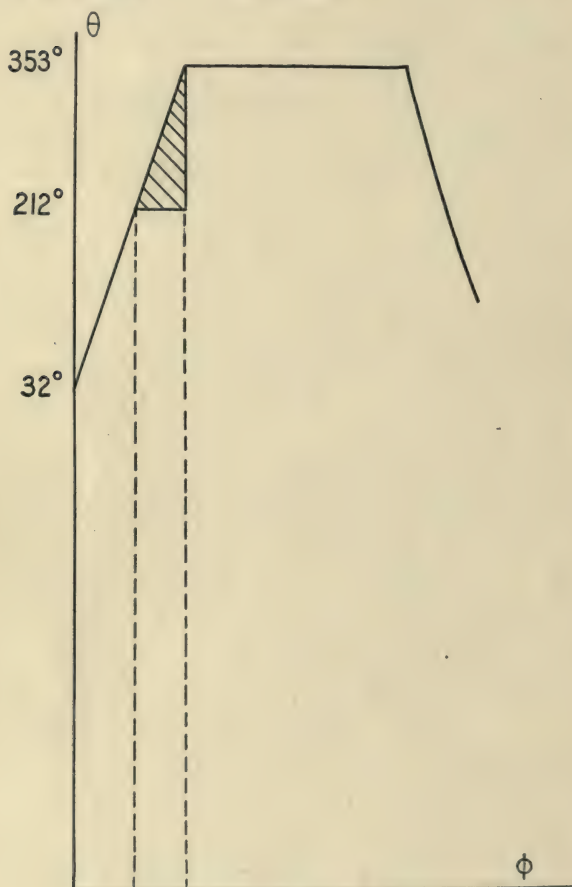


FIG. 33

sent not actual conditions but theoretical conditions which are useful only in giving some ideas of the magnitude of the energy liberated.

Hence the heat energy liberated per pound is $q_1 - q_2$, or 144 B. t. u.

For 2 tons this is 576,000 B. t. u., that is 448,000,000 foot pounds.

The θ - ϕ plot is shown to scale in Figure 33, where the heat liberated is represented by the shaded area.

If this energy is used entirely to produce re-evaporation at a pressure of 14.7 pounds, the number of pounds of steam produced is found as follows. From Table 3, r for 14.7 pounds is 969.7. For each pound of water there is then evaporated $x = \frac{144}{969.7} = 0.149$ pound. The volume occupied by this steam is 0.149 s or $0.149 (26.78) = 3.98$ cubic feet. The steam produced by the explosion would fill 4000 (3.98) or 15,920 cubic feet at the atmospheric pressure. Actually the steam would in part be formed at higher pressure, there would be destructive expansion, and the values calculated above repre-

23. *Equivalent evaporation and boiler horse power.* What is the equivalent evaporation from and at 212° Fahrenheit per pound of coal in a boiler where 19,000 pounds of feed water at a temperature of 70° Fahrenheit are converted into steam at 80.3 pounds gauge by the combustion of one ton of coal?

If a boiler horse power is defined as the capacity of a boiler which evaporates 30 pounds of water per hour from a temperature of 100° Fahrenheit against a pressure of 70 pounds gauge, find the equivalent evaporation from and at 212° Fahrenheit, which is equal to a boiler horse power. If the above evaporation took place in two hours, what is the boiler horse power?

Solution. From Table 3,

For 80.3 + 14.7 lb.	For 70° F.	For 70 + 14.7 lb.	For 100° F.	For 212° F.
$q = 294.6$	$q = 38.1$	$q = 286.4$	$q = 68$	$q = 180.3$
$r = 890.5$		$r = 896.7$		$r = 969.7$

Per pound of coal, the given boiler evaporates $\frac{19,000}{2000} = 9.5$ pounds of water from 70° Fahrenheit to steam at 95 pounds absolute. The B. t. u. required are evidently $294.6 + 890.5 - 38.1 = 1147.0$ per pound.

Per pound of water from and at 212° the B. t. u. required are 969.7. The equivalent evaporation of the given boiler is evidently $9.5 \frac{1147.0}{969.7} = 11.25$ pounds of water from and at 212° per pound of coal.

Now a boiler horse power is equivalent to $30(286.4 + 896.7 - 68) = 30(1115.0)$ B. t. u. per hour. That is, a boiler horse power is equal to an equivalent evaporation of $30 \frac{1115.0}{969.7} = 34.5$ pounds per hour.

The actual boiler had an equivalent evaporation per hour of $\frac{2000}{2} 11.25 = 11,250$ pounds. Hence, a boiler horse power of $\frac{11,250}{34.5} = 325$.

24. Surface condenser. A surface condenser receives steam at a pressure of 4 pounds absolute. Twenty pounds of water at 60° Fahrenheit flow through the pipes of the condenser for every pound of steam that is condensed. What is the temperature of the feed water leaving the condenser? The condenser water flows to a Worthington cooling tower, where it trickles down through tiles that cause it to spread over a large surface. If five per cent is evaporated and carried away by a current of air from a fan passing through the tower, other things being equal, what would be the final temperature of the remaining 95 per cent?

Solution. For 4 pounds pressure, $q = 121.0$ and $r = 1005.5$, or $\lambda_1 = 1126.5$ B. t. u. For 60° Fahrenheit temperature, $q = 28.1 = \lambda_2$.

Let q_x represent the heat of the liquid per pound for the final common temperature of the steam and condenser water.

Then

$$\lambda_1 - q_x = 20(q_x - \lambda_2),$$

or

$$21 q_x = \lambda_1 + 20 \lambda_2,$$

or

$$q_x = \frac{1126.5 + 562}{21} = 80.2.$$

Hence the temperature is 112° Fahrenheit.

To evaporate 5 per cent of a pound of water at 112° Fahrenheit requires $0.05 r$ B. t. u. For 112° Fahrenheit, r is 1029. Hence it requires 51.5 B. t. u.

The abstraction of 51.5 B. t. u. from the remaining 95 per cent of the pound to re-evaporate the 5 per cent reduces the heat of the liquid for the water to $[(0.95)(80.2) - 51.5]$ B. t. u. for each 95 per cent of a pound. Hence per pound of remaining water q is $\frac{(0.95)(80.2) - 51.5}{0.95} = 26$ B. t. u. The corresponding temperature is $26 + 32$, or 58° Fahrenheit. (Actually the temperature of the cooling water and the condensed steam are not exactly the same.)

25. Jet condenser. A jet condenser, which injects cooling water into the steam which is to be condensed, receives steam at 4 pounds absolute. The water of the jet is 60° Fahrenheit. If the feed is 120° Fahrenheit, how much condensing water is used per pound of steam condensed?

Solution. Using the values of λ from Problem 24 and q for 120° Fahrenheit as 88 B. t. u., if we represent by m the mass of the condensing water used per pound of steam condensed, then

$$(1126.5 - 88) = m(88 - 28.1),$$

or

$$m = \frac{1126.5 - 88}{88 - 28.1} = 17.3 \text{ pounds.}$$

26. Rankine cycle. The operations of a steam engine are, essentially, the admission of steam, its expansion, and its final expulsion. The quantity of the working fluid in the engine is not always the same, and for this reason the p - v plot of the engine does not represent a closed cycle in the sense in which the cycle of the Carnot engine of page 22 is a closed cycle. Considering the boiler, the engine, and the condenser as a unit, the steam does pass through a closed cycle. The assumptions of the theoretical and ideal Rankine cycle are as follows: (1) Water is raised from the temperature at which it is fed into the boiler to the temperature and pressure at which it is to be admitted to the engine cylinder. (2) When the admission valve of the engine opens, an isothermal expansion occurs in the boiler and engine cylinder while the piston is moved forward by this expansion until cut-off occurs. (3) The steam then expands adiabatically in the engine cylinder until its pressure is that of the condenser. (4) It is then expelled by the back stroke of the piston and condensed at the constant temperature and pressure of the condenser. This condensed water is then fed back into the boiler.

Draw the p - v and θ - ϕ plots of such a Rankine cycle for one pound of water received by the engine and expelled at a condenser pressure such that the vacuum gauge reads 24.4 inches of mercury. The boiler pressure is 80 pounds gauge, the atmospheric pressure is 15 pounds. Calculate the efficiency.

Solution. The p - v plot is shown in Figure 35, which is drawn to scale to emphasize the volume relations. The θ - ϕ plot is also drawn to scale and is shown in Figure 34. The work done may most easily be found from the θ - ϕ plot. The shaded area $abcd$ represents the work done in heat units. The efficiency is evidently

$$\eta = \frac{\text{area } abcd}{\text{area } abceg} = \frac{\text{areas } (Oobf + bcef - Ooag - adeg)}{\text{areas } (Oobf + bcef - Ooag)}$$

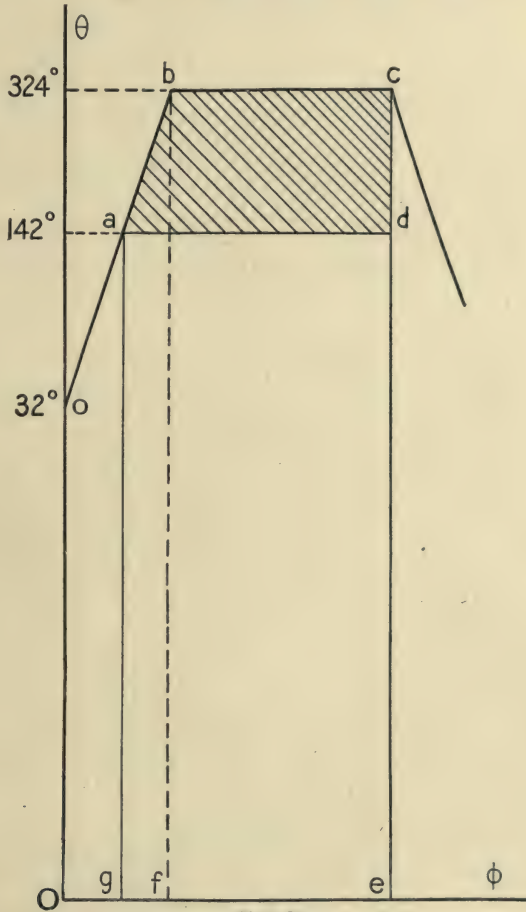
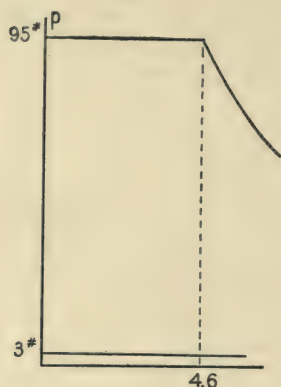


FIG. 34

Replacing these areas by their heat equivalents, we have

$$\eta = \frac{q_1 + r_1 - q_2 - x_2 r_2}{q_1 + r_1 - q_2} = 1 - \frac{x_2 r_2}{q_1 + r_1 - q_2}$$

The boiler pressure is $80 + 15 = 95$ pounds absolute. Since 30 inches of mercury correspond to a pressure of 14.7 pounds, 24.4 inches correspond to a condenser vacuum of $\frac{14.7}{30} 24.4 = 12$ pounds. Hence the condenser pressure is 3 pounds absolute.



From Table 3,

For 95 pounds pressure

$$q_1 = 294.6$$

$$r_1 = 890.5$$

$$\phi_w = 0.4699$$

$$\phi_v = 1.1363$$

$$s_1 = 4.644$$

For 3 pounds pressure

$$q_2 = 109.6$$

$$r_2 = 1012.2$$

$$\phi_w = 0.2011$$

$$\phi_{v_2} = 1.6841$$

$$s_2 = 118.4$$

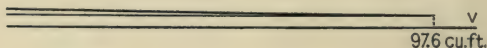


FIG. 35

For the problem under solution the value of x_2 must be found by using equation (55), the condition for an isentropic expansion. Thus

$$\phi_{w_1} + \phi_{v_1} = \phi_{w_2} + x_2 \phi_{v_2},$$

$$\text{or } x_2 = \frac{0.4699 + 1.1363 - 0.2011}{1.6841} = 0.834.$$

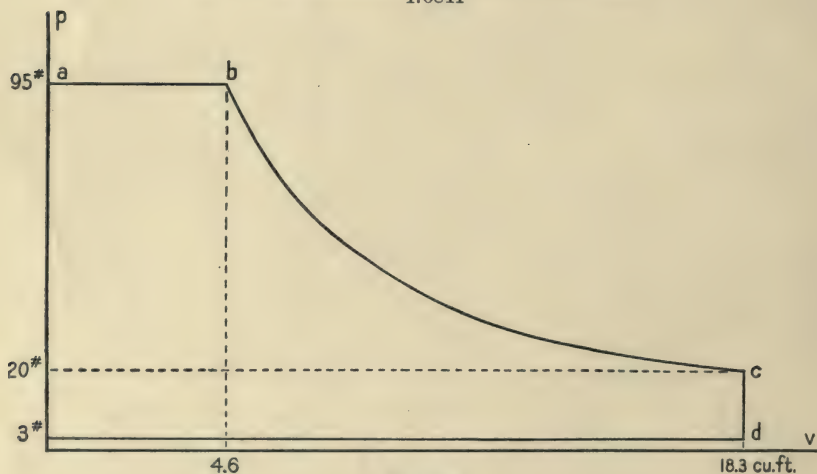


FIG. 36

Then substitution in the equation above for the "Rankine efficiency" gives

$$\eta = 1 - \frac{(0.834)(1012.2)}{294.6 + 890.5 - 109.6} = 1 - 0.786, \text{ or } 21.4 \text{ per cent.}$$

27. *Incomplete expansion.* Because of the large piston displacement required for a complete expansion down to the condenser pressure, the resulting small force exerted upon the piston and the

larger amount of friction, better mechanical efficiency can be obtained by sacrificing some theoretical thermal efficiency and stopping the expansion at some pressure higher than the condenser back pressure. Such a cycle may be represented as in Figure 36. From c to d the steam in an actual engine flows through the exhaust valve until the pressure has fallen to the condenser back pressure. From d to e the remainder of the steam is expelled by the back stroke of the engine. For all thermodynamic purposes in this discussion the effect is the same as if the steam were cooled at constant volume from c to d , and compressed isothermally from d to e .

If the initial pressure is 95 pounds and the condenser back pressure 3 pounds absolute as in Problem 26, find the thermal efficiency if release occurs at 20 pounds absolute. Draw the θ - ϕ plot. What proportion of the total piston displacement is the cut-off in this case?

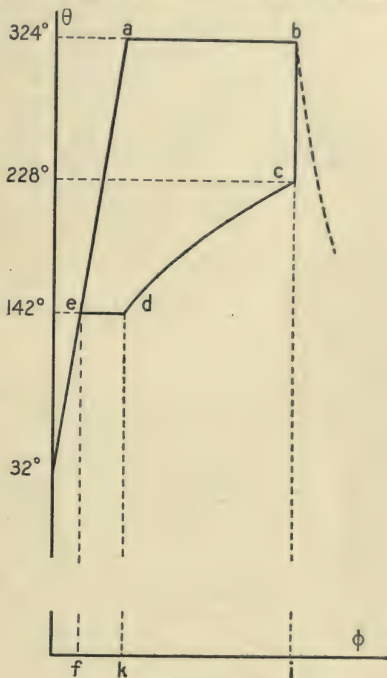


FIG. 37

Solution. The p - v and θ - ϕ plots are shown to scale in Figure 36 and Figure 37. The efficiency may be found from the θ - ϕ plot by the use of a planimeter. The line cd represents the constant volume of v_c . The efficiency may also be obtained by calculations based on the p - v plot as follows. Denote by 1 the state for 95 pounds pressure, by 2 that for 20 pounds, and by 3 that for 3 pounds.

Then

$$\eta = \frac{\text{areas}(abmo + bcnm - dnoe)}{J(q_1 + r_1 - q_3)},$$

where

area $abmo = p_a v_a = p_1 s_1$ per pound of entering steam,

area $bcnm = J(q_b + p_b - q_c - x_c p_c) = J(q_1 + p_1 - q_2 - x_2 p_2)$,

area $dnoe = p_d v_d = p_3 v_c = p_3(x_2 u_2 + \sigma)$.

From Table 3,

For $p_1 = 95$

$q_1 = 294.6$

$r_1 = 890.5$

$s_1 = 4.644$

$\phi_{w_1} = 0.4699$

$\phi_{v_1} = 1.1363$

$p_1 = 808.8$

For $p_2 = 20$

$q_2 = 196.4$

$r_2 = 959.4$

$s_2 = 20.09$

$u_2 = 20.07$

$\phi_{w_2} = 0.3362$

$\phi_{v_2} = 1.3957$

$p_2 = 885.1$

For $p_3 = 3$

$q_3 = 109.6$

For an adiabatic expansion to a pressure of 20 pounds and quality x_2 we have

$$\phi_{w_1} + \phi_{v_1} = \phi_{w_2} + x_2 \phi_{v_2},$$

or

$$x_2 = \frac{0.4699 + 1.1363 - 0.3362}{1.3957} = 0.909.$$

Hence from the relations established from Figure 36

Work during admission is $(95)(144)(4.644) = 63,200$ ft. lb. per lb. of steam,

Work during expansion is by eq. (36)

$$778[294.6 + 808.8 - 196.4 - (0.909)(885.1)] = 79,400 \text{ ft. lb.}$$

The combined work of admission and expansion is $\frac{142,600 \text{ ft. lb.}}{}$

Work during expulsion is $(3)(144)[0.909(20.07)] = \frac{7800 \text{ ft. lb.}}{}$

Therefore external work done by the engine is $\frac{134,800 \text{ ft. lb. per lb. of steam.}}{}$

The input is $778(294.6 + 890.5 - 109.6) = 832,000$ ft. lb. per lb. of steam.

Hence the efficiency is $\frac{1348}{8320} = 16.1$ per cent.

The volume at release per pound is $v_e = (0.909)(20.07) = 18.3$ cubic feet.

The volume at cut-off per pound is $v_b = 4.64$.

The cut-off is then $\frac{4.6}{18.3} = 25$ per cent.

28. Efficiency and boiler pressure. Assuming a non-conducting engine working non-expansively and non-conducting, draw the p - v and θ - ϕ plot for a boiler pressure of 90 pounds absolute and an atmospheric pressure of 14.7 pounds. The feed water is 60° Fahrenheit. Calculate the efficiency for pressures of 50, 70, 90, 110, and 130 pounds absolute and plot a curve showing the relation between boiler pressure efficiency for this case.

Solution. The efficiency for a pressure of 90 pounds is calculated as an illustration. The p - v and θ - ϕ plots are shown in Figures 38 and 39

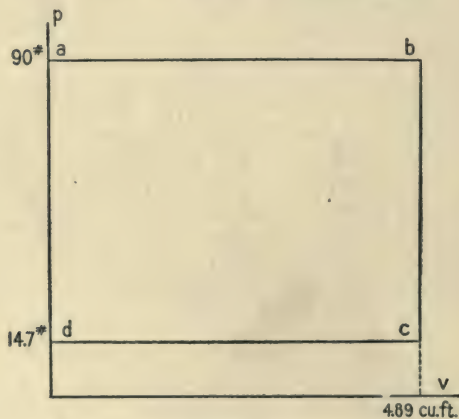


FIG. 38

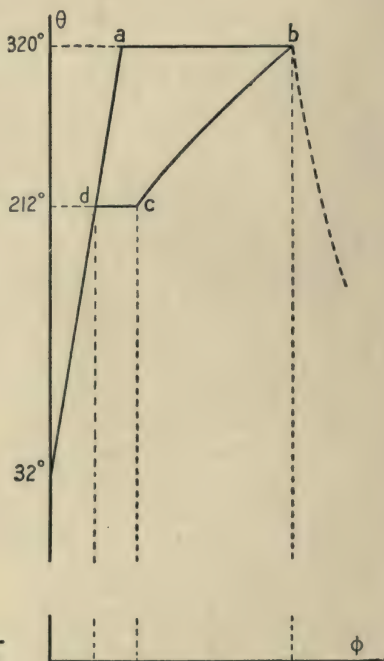


FIG. 39

respectively. Table 3 gives for 90 pounds $q_1 = 290.7$, $r_1 = 893.5$, and $s_1 = 4.886$. For 60° Fahrenheit $q_3 = 28.1$.

The input is $q_1 + r_1 - q_3 = 290.7 + 893.5 - 28.1 = 1156.1$ B.t.u.

The output is most conveniently found from the p - v plot. It is $(p_a - p_d)v_b = (p_1 - p_2)s_1$.

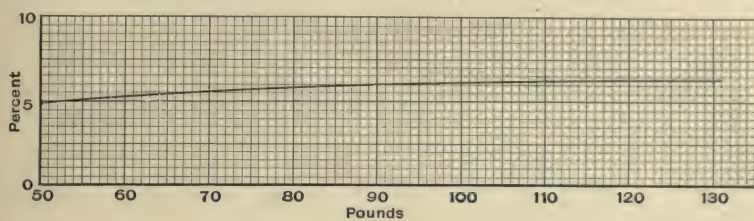


FIG. 40

Or substituting the indicated values, the output is

$$(144)(90 - 14.7)4.886 = 52,900 \text{ foot pounds} = 68 \text{ B. t. u.}$$

The efficiency is then $\frac{68}{1156} = 5.9$ per cent.

The curve of efficiency and boiler pressure is shown in Figure 40.

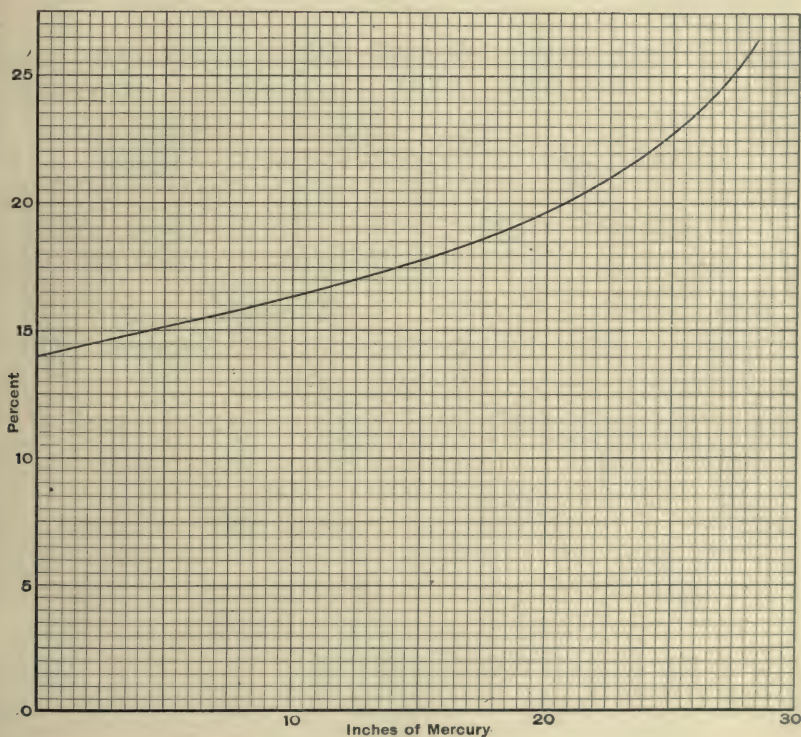


FIG. 41

29. *Condensing engine.* Find for the Rankine cycle the theoretical efficiency of an engine supplied with steam at 100 pounds absolute for the following values of the condenser back pressure: 1, 2, 3, 4, 6, 10, and 14.7 pounds absolute. Plot efficiency against inches of mercury indicated by a vacuum gauge from an atmospheric pressure of 14.7 pounds.

Solution. The solution of this problem is identical with that of Problem 26. The plot of Figure 41 shows the variation of the efficiency with the condenser vacuum.

30. *Use of the temperature entropy chart.* Assume a frictionless non-conducting engine working on a cycle of incomplete expansion as in Problem 27. If the clearance is negligible, the cut-off 40 per cent of the piston displacement of 5 cubic feet, the admission steam at 100 pounds absolute, and the condenser back pressure 3 pounds absolute, draw the p - v and θ - ϕ plots.

Solution. One pound of steam at 100 pounds pressure has a volume of 4.432 cubic feet. Therefore at cut-off the engine cylinder contains $\frac{40}{100} \frac{5.00}{4.432} = 0.451$ pounds.

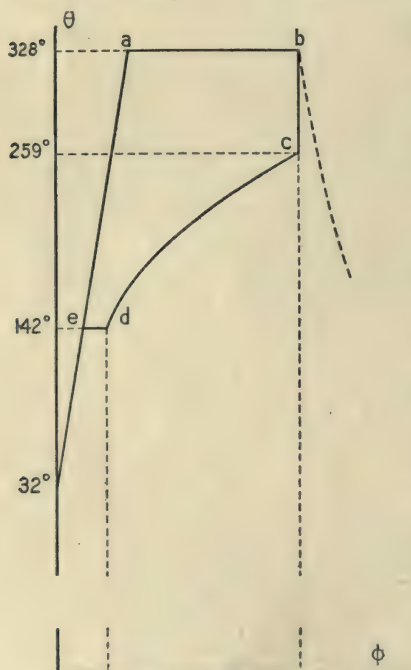


FIG. 43

Since the θ - ϕ chart and the tables for steam given on page 130 are both per pound of steam, it is most convenient to reduce the engine under consideration to one containing a pound of steam (or mixture if the steam is wet). Thus, if the cylinder volume is imagined to be increased in the ratio $\frac{1}{0.451}$, that is, to $\frac{5.00}{0.451} = 11.1$ cubic feet, there will be one pound of steam in the cylinder at cut-off.

From the θ - ϕ chart of Figure 42 it is found that the entropy of one pound of dry steam at 100 pounds is 1.602. Where the line of constant entropy of this value intersects the line of constant volume for 11 cubic feet gives the point c on the θ - ϕ plot of Figure 43, corresponding to the point c, which may now be plotted on the p - v plot of Figure 44. If intervening points on the p - v plot of the expansion are desired, they may be found by noting the intersections of the line of constant entropy, equal to 1.602, with various lines of constant volume and then finding the pressures corresponding to those intersections. The constant volume line of value

11 cubic feet is followed until it intersects the constant pressure line for 3 pounds. This gives the point d on the θ - ϕ plot.

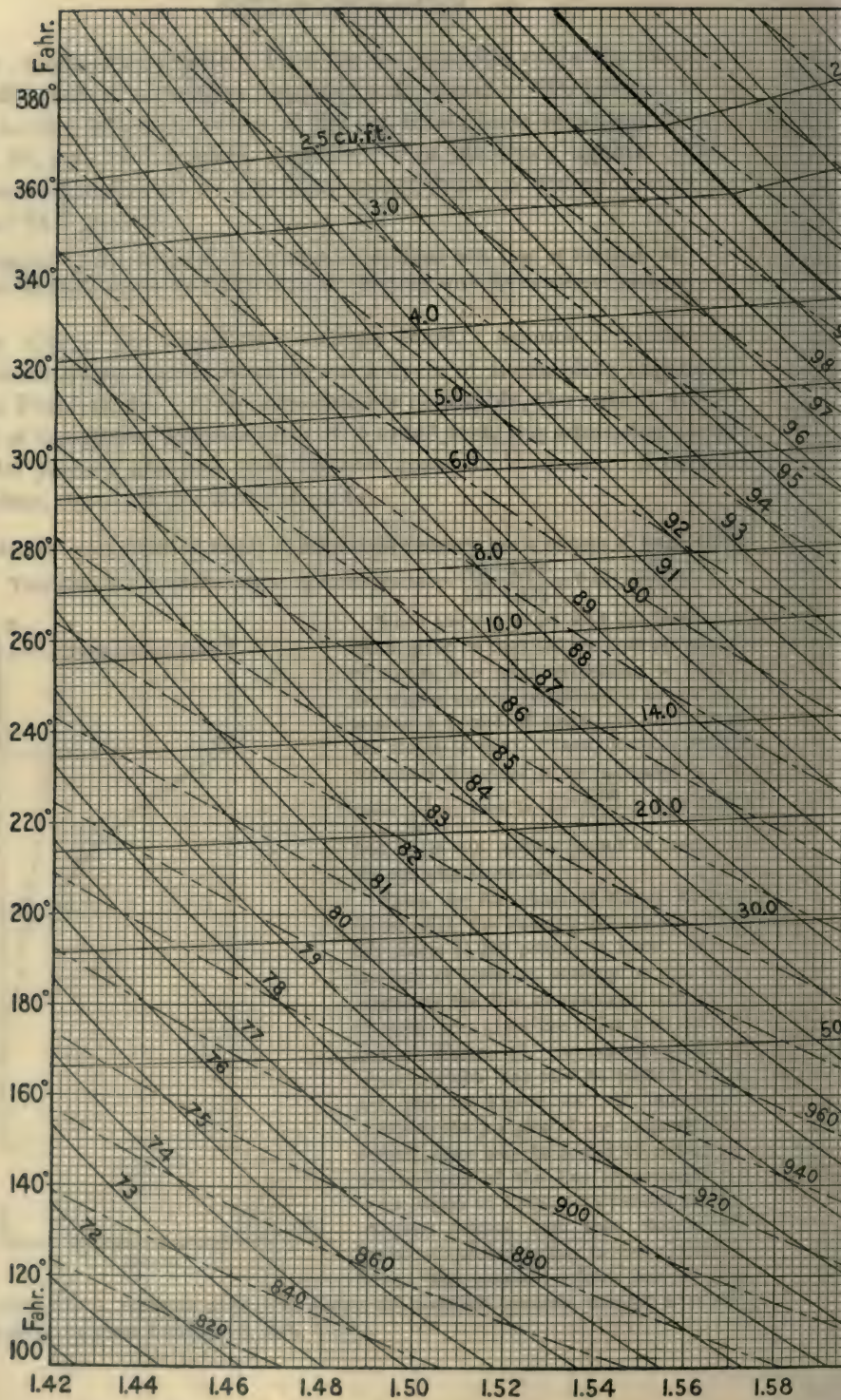
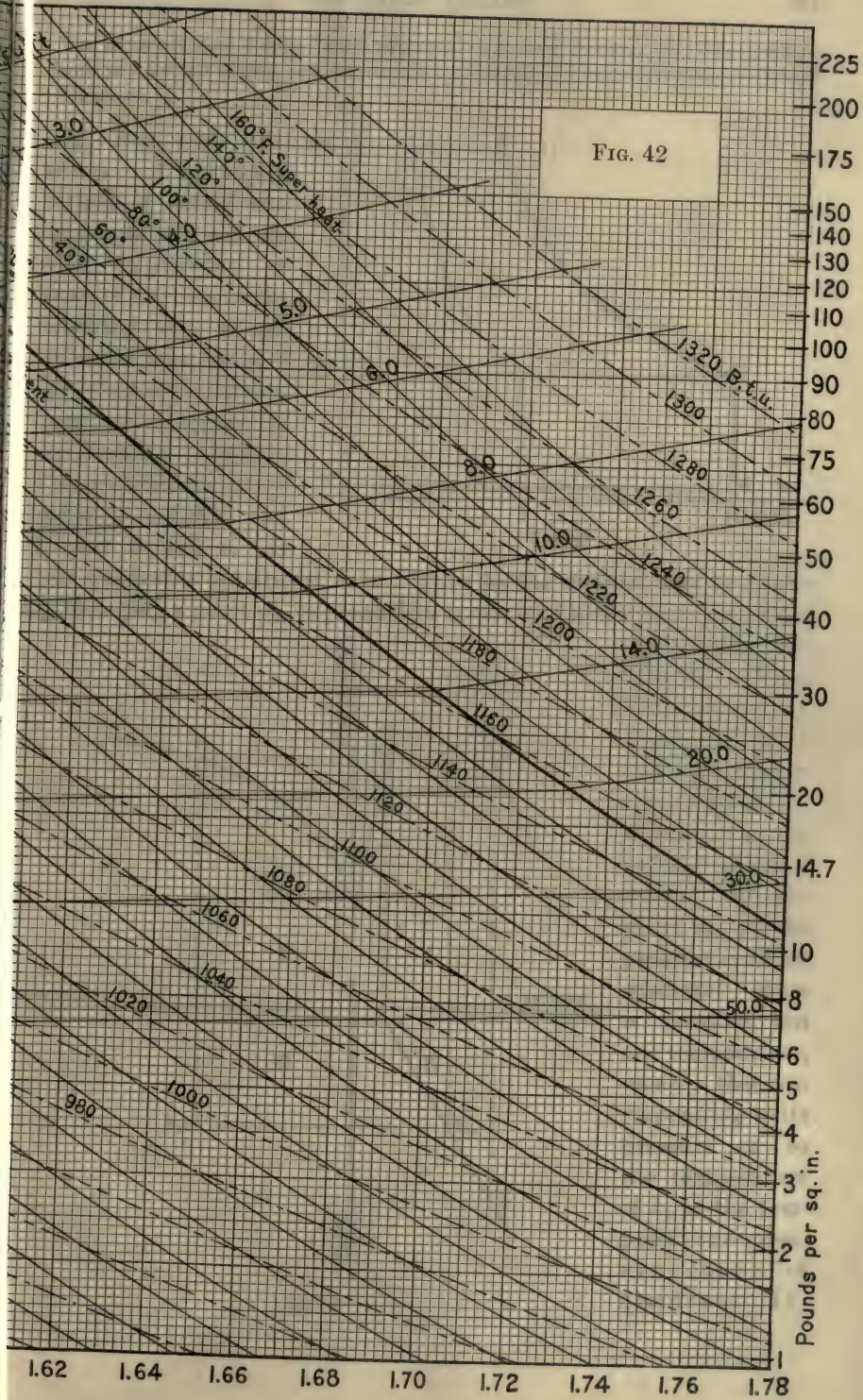
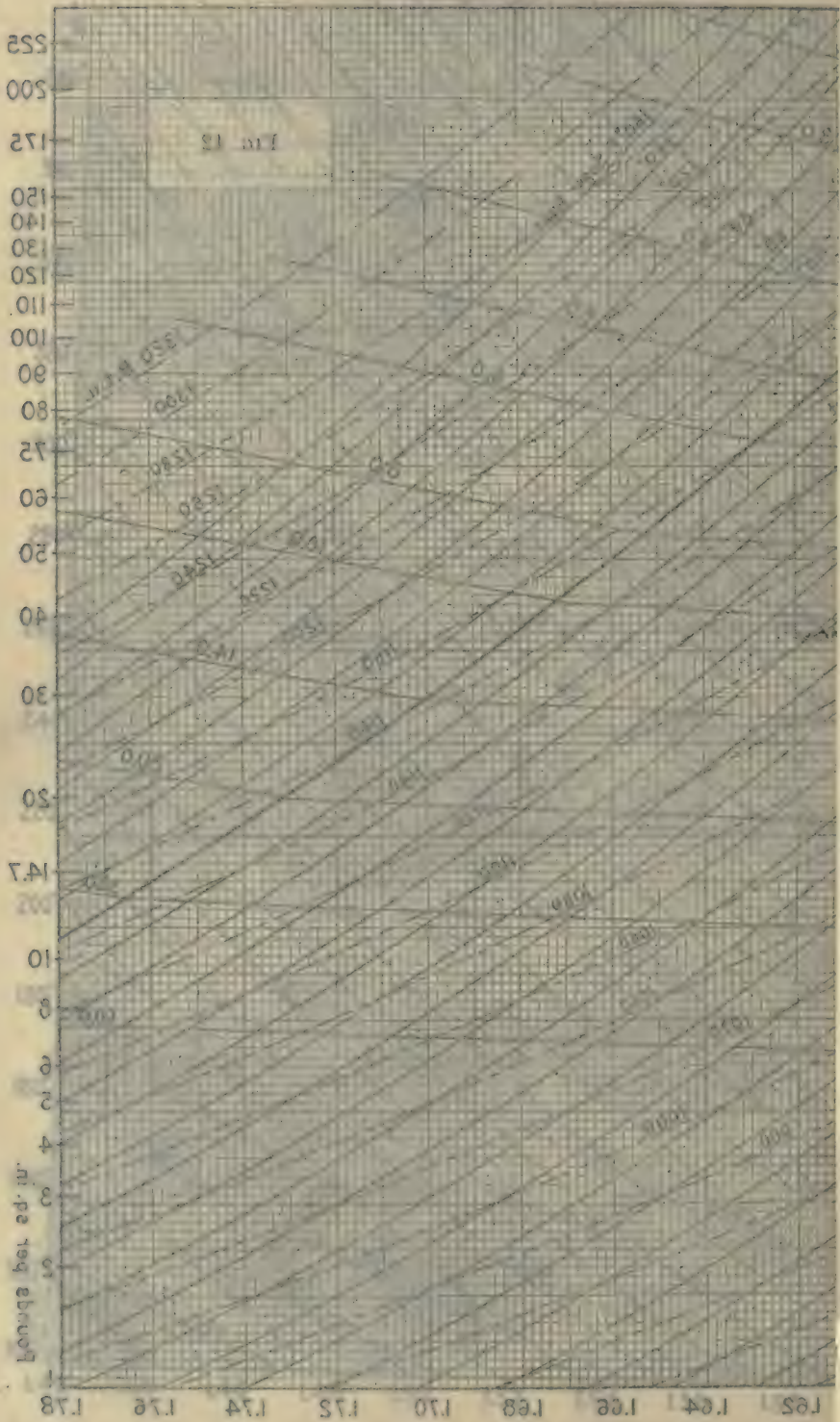


FIG. 42





The efficiency may now be found after the manner of Problem 27. The pressure volume plot for this engine with its piston displacement of 5 cubic feet is shown dotted in Figure 44.

31. *Boulvin diagram.* A simple method of transforming graphically a steam cycle expressed in terms of any one of the following sets of coördinates into terms of any other set is known from its proposer as the Boulvin diagram. The coördinates are p and v , p and θ , θ and ϕ , and ϕ and v . If

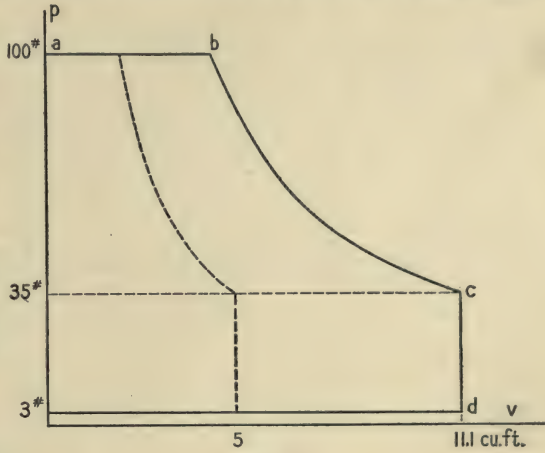


FIG. 44

four quadrants are drawn, as in Figure 45, the method may be shown by changing a point k_{iii} expressed in terms of p and v into each of the

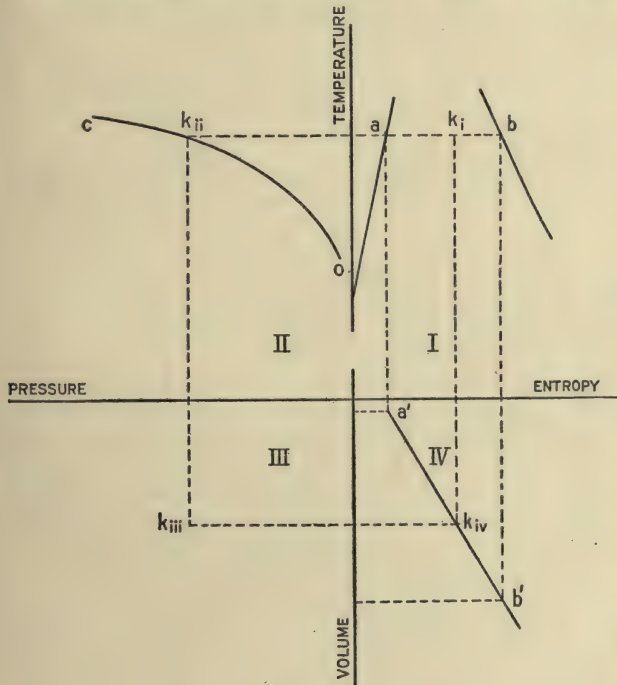


FIG. 45

other sets of coördinates. Thus, in quadrant II, draw the curve oc for the pressure-temperature relations for the boiling point of water. To any point k_{iii} in the pressure-volume quadrant there is possible but one temperature, namely, that corresponding to its pressure or to the temperature of k_{ii} on the temperature-pressure curve of quadrant II.

Now draw in quadrant I the θ - ϕ curve ab for this temperature. The line of constant temperature $a'b'$ in quadrant IV may then be drawn as described in Section 51. The projection of k_{iii} on the curve $a'b'$ at k_{iv} then gives the entropy corresponding to the pressure (that is, to the temperature) and volume of the point k_{iii} . The point k_i on the θ - ϕ plot is then found by projecting k_{iv} on ab .

Figure 46 shows the indicator card of an actual engine. The piston displacement was 1.037 cubic feet. Compression occurred at 3.8 per cent stroke. The amount of steam supplied per stroke to the end of the cylinder for which this card was obtained was 0.0781

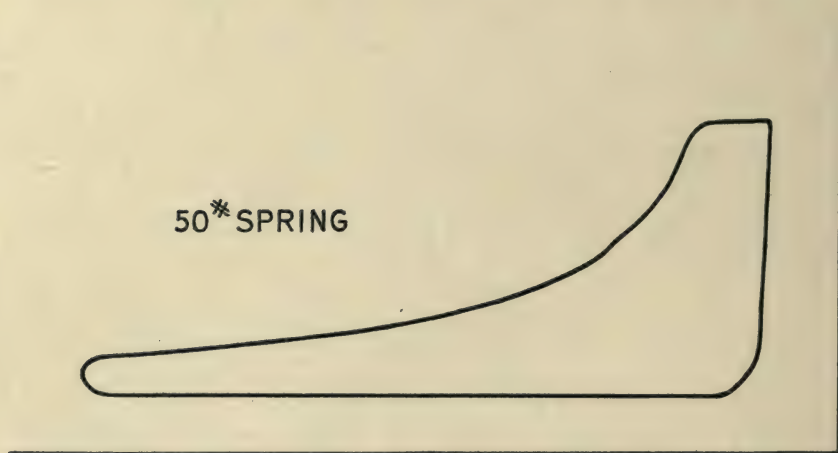


FIG. 46

pound. The clearance was 9.4 per cent of the piston displacement. The receiver back pressure was 14.7 pounds. Draw the θ - ϕ plot for the expansion and discuss the heat transfer indicated.

Solution. Assume the steam dry at compression. This is the ordinary assumption of steam engineering practice. All the water in the cylinder at release is supposed to reëvaporate during the back stroke of the piston and be expelled, except for that steam which remains in the clearance spaces.¹ The steam at compression then occupies $(9.4 + 3.8)$ per cent of 1.037 cubic foot at a pressure of 14.7 pounds. For this pressure the specific weight or density is found from Table 3 to be 0.0374. Hence the clearance steam was $\frac{13.2}{100}(1.037)(0.0374) = 0.0052$ pound. At cut-off there was then present in the cylinder $0.0781 + 0.0052 = 0.0833$ pound of steam.

Since the θ - ϕ diagram is most conveniently plotted for unit weight of steam, we

¹ For further discussion of the quality at compression see Reeve, "Thermodynamics of Heat Engines," p. 214 (Macmillan).

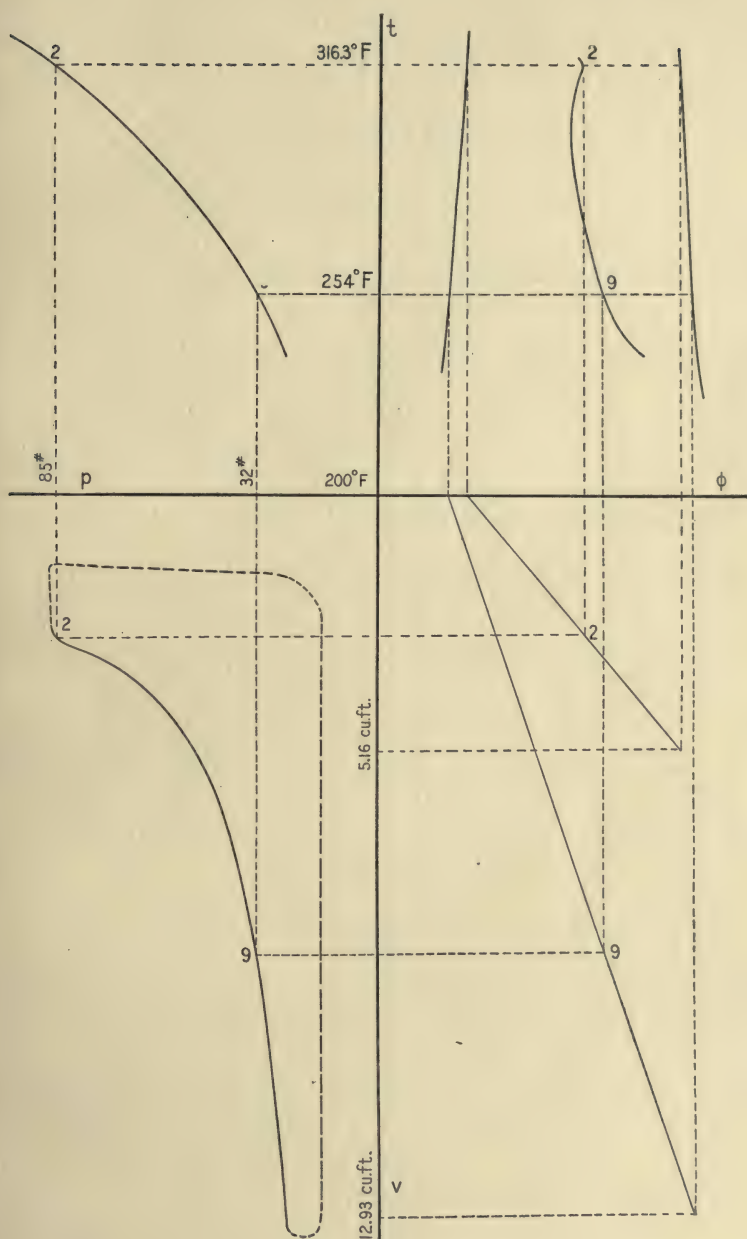


FIG. 47

follow the method of Problem 30 and plot in the p - v quadrant of the Boulvin diagram the changes in the pressure and volume per pound of mixture. That is, all the volumes of the actual problem are increased in the ratio of 1 to 0.0833. The p - v plot of quadrant III of Figure 47 shows the expansion line of the indicator card of Figure 46 so modified and plotted. The piston displacement has been increased to $\frac{1.037}{0.0833} = 12.45$ cubic feet.

Similarly, the clearance has become 9.4 per cent of 12.45, or 1.17 cubic feet.

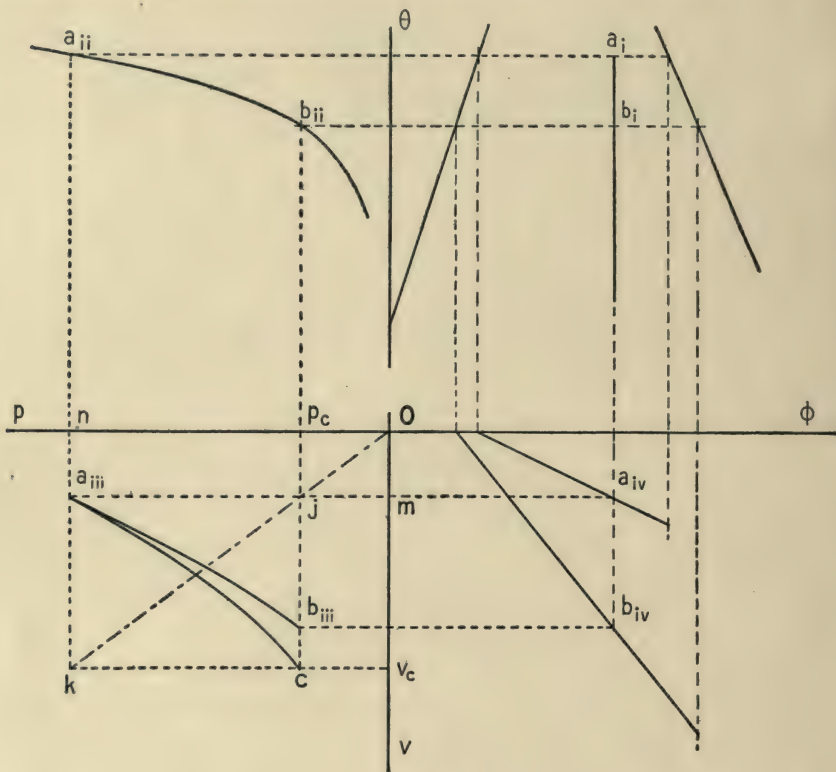


FIG. 48

The θ - ϕ plot of the expansion has been constructed by the method outlined above. The construction lines for two points which have been reproduced in the plate indicate sufficiently the method.

Discussion. The steam supplied to this engine was found by means of a throttling calorimeter to have a quality of 99 per cent when it entered the cylinder. Due to condensation the quality had fallen to 47 per cent at cut-off. Some further condensation took place during the earlier part of the expansion. For the latter and greater part of the expansion, however, heat was returned to the steam from the cylinder walls.¹ The quality therefore rose to 72 per cent at release.

¹ That this heat is at a lower availability as a result of the lower temperature is evident from Problem 50. The present problem serves as an introduction to the discussion of the gain due to superheating which is made in Section 62 of page 98.

32. *Hyperbolic expansion line.* Because the actual adiabatic line for a steam transformation is not to be drawn by any simple graphical method, it is the custom of steam engineers to judge of the actual operation of a given engine by drawing through the point of cut-off on the p - v plot (indicator card) of the engine a hyperbola. It is the object of this problem to compare the hyperbolic line with the true isentropic. By using the Boulvin diagram find the p - v plot of an isentropic line which has been assumed in the θ - ϕ quadrant. Then through the point of this adiabatic in the p - v quadrant which corresponds to the highest assumed pressure draw a rectangular hyperbola.

Solution. The construction of the adiabatic ab is evident from Figure 48. To draw the hyperbola proceed as follows. Draw $a_{iii}m$ perpendicular to Ov and $a_{iii}n$ perpendicular to Op . Through some point v_c on Ov draw v_ck perpendicular to Ov . At k the intersection of v_ck and $a_{iii}n$ draw kO to the origin. Through j , the intersection of kO and $a_{iii}m$, draw p_cj perpendicular to Op to meet v_ck in c , which is a second point on the hyperbola. In a similar manner other points may be obtained.

CHAPTER IV

SUPERHEATED STEAM

55. Three Molecular States. It is a well-known fact of physics that there are three molecular conditions in which substances may exist. These are commonly stated as solid, liquid, and aeriform. The aeriform bodies are commonly subdivided into two classes; namely, those which obey Boyle's Law at ordinary temperatures and pressures and are called gases, and those which at ordinary pressures and temperatures can exist in both a liquid and an aeriform state and are called vapors. The distinction between these molecular conditions is not one depending upon the substance considered, but is one dependent wholly upon its physical conditions of pressure and temperature.

Thus it is known that the so-called "permanent gases" like hydrogen and oxygen have been liquefied and even solidified at extremely low temperatures and high pressures. For any gas there is a certain critical temperature above which it cannot be liquefied. And at this critical temperature there is a certain minimum value of the pressure which must be applied to produce liquefaction. In the development of this portion of physics the term "permanent gases" was introduced and applied to those substances in an aeriform condition which resisted the efforts that had been made to liquefy them. The term to-day, since the work of Wroblewski and Olszewski, Pictet, Cailletet, Dewar, and Linde, is evidently a misnomer.

The same substance can exist in any one of the three principal molecular states. The commonest example of this is, of course, water, which at ordinary temperatures may coexist in all three states. For the ordinary temperatures of our atmosphere, however, the aeriform state which water assumes must be classed as a vapor state. It seems reasonable to consider that at higher temperatures it would more nearly approximate a true gaseous state.

56. Superheated Steam. Water vapor, as has already been noted in previous discussions, obeys the ordinary vapor laws by having the temperature and pressure of the liquid from which it is produced and with which it is in contact. But when removed from contact with

its liquid (whether actually or essentially as described in Section 44), its temperature may be raised above its former value. The vapor or steam is then said to be superheated. According to that portion of the molecular theory stated above it is to be expected, then, that the pressure, specific volume, and temperature of superheated steam would tend to follow an equation somewhat of the form of the characteristic equation of a perfect gas.

In accordance with this supposition two very satisfactory formulas have been proposed. They are both of the form $pv = R\theta$, but contain corrective terms. The simplest, and for ordinary purposes one of sufficient agreement with the actual experimental results, is that suggested by Tumlriz and based upon the experiments of Battelli. It will usually be referred to as the "equation of Tumlriz." Expressed in the units of the F. P. S. system it is

$$pv = 85.85 \theta - 0.256 p, \quad . \quad . \quad . \quad . \quad (57)$$

where θ is the absolute temperature in Fahrenheit degrees and p and v are specific pressure and volume.

The equation proposed by Knoblauch, Linde, and Klebe¹ is more satisfactory in its agreement, but the accuracy is attained at the expense of a tedious calculation. It is expressed in the same symbols as equation (57) as follows:

$$pv = 85.85 \theta - p(1 + 0.00000976 p) \left(\frac{150,300,000}{\theta^3} - 0.0833 \right). \quad (58)$$

57. Specific Heat of Superheated Steam. The specific heat of superheated steam has been determined for constant pressure conditions by Knoblauch and Jakob. The following table of their values is only partial:²

Pressure # per sq. in.	$p = 85.3$	$p = 113.8$	$p = 142.2$	$p = 170.6$	$p = 199.1$
Temperature ° F.	$t_0 = 316$	$t_0 = 336$	$t_0 = 354$	$t_0 = 368$	$t_0 = 381$
$t_1 = 392$	0.530	0.560	0.597	0.635	0.677
$t_1 = 482$	0.514	0.532	0.552	0.570	0.588
$t_1 = 572$	0.505	0.517	0.530	0.541	0.550

The table gives the average value of the specific heat (for the pressure indicated in the upper row) which should be used in calculating

¹ These equations expressed in F. P. S. units are taken from the "Steam and Entropy Tables" of Professor Peabody.

² A complete table in F. P. S. units is given in Professor Peabody's "Tables."

the heat required to superheat from the saturation temperature (corresponding to this pressure and given in the second row) to the temperature given in the left-hand column. Illustration of the use of this table will be given in Problem 35.

For all engineering purposes steam is superheated at constant pressure. The additional heat required to raise one pound of dry steam from the condition of saturated vapor at a pressure p_0 and a temperature t_0 (or θ_0 absolute) to a temperature t (or θ) will be denoted by the symbol h . That is,

$$h = c_p(\theta - \theta_0) = c_p(t - t_0), \quad \dots \quad (59)$$

where c_p represents the specific heat at constant pressure as found from the above table. The total heat contents of a pound of superheated steam at a temperature t is then

$$\lambda = q_0 + r_0 + h, \quad \dots \quad (60)$$

where r_0 and q_0 are the values for the saturation temperature t_0 which corresponds to the pressure p (equal to p_0 , since the superheating is at constant pressure).

For many engineering purposes the value of c_p may be taken as 0.48 with sufficient accuracy.

58. Entropy of Superheated Steam. The increase in entropy due to superheating, expressed by the symbol ϕ_s , is sufficiently well obtained for most purposes¹ by using the average value of the absolute temperature at which the superheating occurs and the specific heat c_p as 0.48. Thus,

$$\phi_s = \int_{\theta_0}^{\theta} \frac{c_p d\theta}{\theta} = \frac{h}{\frac{\theta + \theta_0}{2}} = \frac{0.48(\theta - \theta_0)}{\frac{\theta + \theta_0}{2}} = 0.96 \frac{\theta - \theta_0}{\theta + \theta_0} \quad \dots \quad (61)$$

The total entropy ϕ of a pound of superheated steam as measured from the assumed zero of entropy at 32° Fahrenheit may then be written

$$\phi = \phi_w + \phi_v + \phi_s. \quad \dots \quad (62)$$

59. Temperature-Entropy Chart for Superheated Steam. The θ - ϕ chart for steam may now be extended into the superheated region to the right of the curve of constant steam weight. The increase of entropy due to superheating at constant pressure is found by equation (61), and lines of constant pressure p_1 , p_2 , and so on, are plotted

¹ Stodola, in the 2d edition of "The Steam Turbine," uses 0.48. Cross interpolation in the table given on page 93 is usually unsatisfactory and not as accurate as a graphical process. The tables of Marks and Davis give values of v , λ , and ϕ for intervals of 10° F. for steam superheated at all the ordinary pressures of engineering.

as in Figure 49. These lines, representing the increase in entropy at constant pressure, are essentially straight for small ranges of temperature. For large ranges and for high temperatures of superheat, they should be plotted, using values of the entropy as determined more exactly by use of the specific heat values obtained by Knoblauch and Jakob.

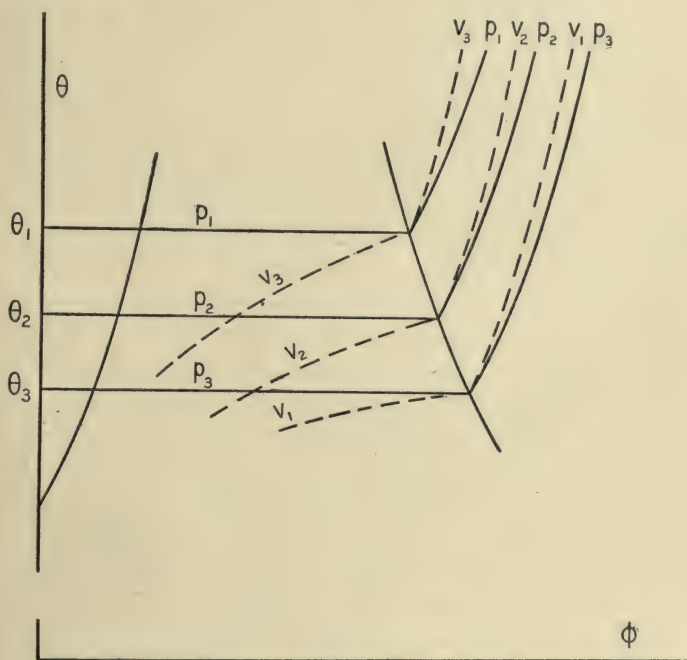


FIG. 49

Constant volume lines may also be extended into the superheated region by calculating for any assumed pressure and volume the corresponding temperature θ in accordance with either equation (57) or (58). The intersection of the line for constant pressure for the assumed value and the line of constant temperature for the calculated value θ gives a point on the desired constant volume line. The constant volume line might be drawn by a method similar to that employed in constructing the constant pressure line and using the increase in entropy for this condition calculated in a similar manner, were it not that the value of c_v as 0.346 is not reliable. Constant pressure and constant volume lines are shown in Figure 49.

In a temperature-entropy chart constructed as above it is evident from Figure 49 that both the lines of constant pressure and the

lines of constant volume are discontinuous at the curve of constant steam weight, since they are different functions of the temperature and the entropy in the wet steam and superheated steam regions. In the wet steam region the lines of constant pressure coincide with the lines of constant temperature. The chart is therefore equally well a θ - ϕ or a p - ϕ plot. In the superheated region it is merely a θ - ϕ plot. Because of the fact that steam is usually heated at constant pressure a p - ϕ plot is sometimes convenient in this region. A chart constructed in this way is shown in Figure 42 on page 86.¹ To indicate temperatures it is necessary to draw "quality" lines through those points representing steam of the same degree of superheat. Such lines would be everywhere equally distant from the curve of constant steam weight if the specific heat of superheated steam were constant. In the chart shown these lines are drawn for intervals of superheat equal to 20 degrees. The scale of temperatures given on the left-hand side obviously refers only to points within the wet steam region or on the dry steam line. The unequal scale of pressures on the right refers to any point of the plot. Such a plot is frequently spoken of as a temperature-entropy chart, but is one only for the region of wet steam and more rigorously should be called a pressure-entropy chart as stated above.

60. Intrinsic Energy of Superheated Steam. Let p_1 be the pressure upon a pound of superheated steam, t_1 (or θ_1 absolute) be its temperature, and v_1 its specific volume as calculated by either equation (57) or (58). This superheated steam may be considered as derived at constant pressure p_0 (equal to p_1) from a pound of dry saturated steam at a saturation temperature t_0 (or θ_0) corresponding to this pressure p_0 . During the formation of this pound of dry steam a quantity of intrinsic energy has been added to it, which measured from water at 32° Fahrenheit has been expressed in equation (50) as $\epsilon_0 = \frac{1}{A}(q_0 + \rho_0)$, where the subscript 0 indicates the value

of any magnitude to which it is attached corresponding to the saturation temperature t_0 . Since $\rho_0 = r_0 - Ap_0u_0$ by equation (45) and since $u_0 = s_0 - \sigma$ by equation (44), it follows that the intrinsic energy of the pound of dry saturated steam may be written as

$$\epsilon_0 = \frac{1}{A}(q_0 + r_0) - p_1(s_0 - \sigma). \quad . \quad . \quad . \quad . \quad (i)$$

¹ The entropy diagram constructed in connection with Problem 50 on page 120 is drawn both as a p - ϕ and as a θ - ϕ plot to illustrate this difference.

During superheating, at the pressure p_1 , to the temperature t_1 the volume has increased to a value v_1 , and there has been done an amount $p_1(v_1 - s_0)$ of external work. Since the total heat added has been h , the increase $\epsilon_1 - \epsilon_0$ in intrinsic energy due to superheating is

$$\epsilon_1 - \epsilon_0 = \frac{1}{A} h - p_1(v_1 - s_0). \quad \dots \quad (ii)$$

The total intrinsic energy ϵ_1 possessed by a pound of superheated steam in the condition indicated by the subscript 1 is then

$$A\epsilon_1 = q_0 + r_0 + h - Ap(v_1 - \sigma) = \lambda_1 - Ap(v_1 - \sigma). \quad (63)$$

61. Isentropic Transformation. From the θ - ϕ plot of Figure 50 it is evident that an isentropic or reversible adiabatic expansion of superheated steam may be considered in two parts; namely, an expansion entirely within the superheated region until the state k represented by the intersection of the isentropic line mn and the curve of constant steam weight, and a further expansion within the region of wet steam. During the first part of the expansion the quality, which may now be taken to mean degrees of superheat, that is $t - t_0$, decreases, and during the second part of the expansion the quality as represented by x decreases.

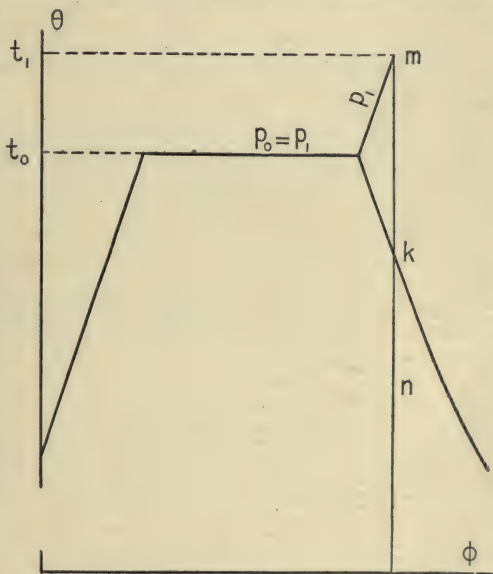


FIG. 50

If the expansion is wholly in the superheated region, the quality in the second state expressed in degrees superheat may most easily be found from a θ - ϕ chart. It may, however, be found analytically by the method illustrated in Problem 38.

If the expansion, as is more usual, extends into the region of wet steam, the quality may be found for the second state by the following relation for constant entropy:

$$\phi_1 = \phi_w + \phi_v + \phi_s = \phi_2 = \phi_{w_2} + x_2 \phi_{v_2}. \quad \dots \quad (64)$$

62. Superheating in Engineering Practice. The occasion for the use of superheated steam is evident in part from Problem 31 on page 87. The steam supplied to the engine there described had a quality of over 99 per cent as it entered the engine. Due to the fact that heat was lost to the walls of the engine cylinder its quality had dropped to 47 per cent at cut-off. During a portion of the expansion the transfer of heat to the cylinder walls continued, but during the latter part of the expansion some of this heat was returned to the mixture, reëvaporating some of the 53 per cent originally condensed during admission. During exhaust still further heat was abstracted from the cylinder walls to reëvaporate (at the condenser back pressure) the moisture left in the cylinder. As a result the temperature of the cylinder rises with the admission of steam and falls with its exhaust.

Consider for a moment that the entering steam is dry, if the quality at cut-off is x_1 , then the intrinsic energy of the mixture has been reduced from $q_1 + \rho_1$ to $q_1 + x_1\rho_1$. The difference $(1 - x_1)\rho_1$ then represents (neglecting all further changes due to heat transfers to the cylinder) a decrease in energy which should be available for the expansion. During the expansion still further heat energy is lost to the cylinder, and later returned at a lower availability. Of the heat lost to the walls during admission, part is returned at lower temperatures during expansion, and the rest in reëvaporation during exhaust. All of this energy of reëvaporation is of course rejected to the condenser.

It is evident, since a quantity of heat $q_2 + r_2$, where the subscript 2 represents condenser conditions, is always to be rejected to the condenser, that as much as possible of the difference $q_1 + r_1 - q_2 - r_2$ between the total heat supplied and rejected, must be converted into work. To accomplish this the mean temperature of the cylinder must be more nearly that of the entering steam. To prevent the loss at admission the steam may be supplied superheated by a number of degrees such that at cut-off the steam in the cylinder is dry. To prevent the loss during expansion the entering steam may be still further superheated so as to be dry at release. Then no heat will be abstracted from the cylinder walls to reëvaporate at the condenser back pressure. For any given engine it is of course a matter of experiment to determine how much superheat is required.

In general, it may be said, superheated steam reduces the amount of heat interchanged between the steam and the cylinder walls, and thereby reduces the weight of steam required per stroke of the

engine for the same work output. Quoting from Ripper:¹ "Superheating may be looked upon not as a means of obtaining a thermal efficiency with the engine in any way proportional to the temperature used in superheat, but as a device for realizing, or at least approaching, the full thermal efficiency of the saturated steam between the ranges of pressure used by the engine."

In the case of the steam turbine much of the advantage of dry steam is the consequent small friction in the turbine nozzle and passages. The presence of water materially increases this friction. Experiments by Stodola² show that the friction increases as the density of the steam increases. Thus, for example, interpreting some experiments performed by Lewicki, Stodola says, the work of friction of a wheel in saturated steam with equal specific weight (*i.e.* density γ), equal size of wheel, and equal velocity is 1.3 times that in the air. The work of friction for a wheel in superheated steam at atmospheric pressure and 572° Fahrenheit is the same as that for air.

63. Résumé of Equations for Superheated Steam.

Characteristic equations.

$$pv = 85.85 \theta - 0.256 p. \quad (57)$$

$$pv = 85.85 \theta - p \left(1 + 0.00000976 p \right) \left(\frac{150,300,000}{\theta^3} - 0.0833 \right). \quad (58)$$

Total heat contents.

$$\lambda = q_0 + r_0 + h = q_0 + r_0 + c_p(\theta - \theta_0). \quad (60)$$

Total intrinsic energy.

$$\epsilon_1 = \frac{1}{A} \lambda_1 - p(v_1 - \sigma). \quad (63)$$

Total entropy.

$$\phi = \phi_w + \phi_v + \phi_s = \phi_w + \phi_v + \frac{c_p(\theta - \theta_0)}{\frac{\theta + \theta_0}{2}}. \quad (62)$$

¹ Ripper, "Steam Engine, Theory and Practice," 4th ed., p. 156 (Macmillan).

² Lowenstein's translation, Stodola, "The Steam Turbine," 2d ed., pp. 135-140.

PROBLEMS AND SOLUTIONS: SUPERHEATED STEAM

33. *Specific volume.* Find from equations (57) and (58) the volume occupied by two pounds of steam superheated 75° Fahrenheit at a pressure of 135 pounds absolute.

Solution. From Table 3, for $p = 135$ we have $t_0 = 350^\circ \text{F.}$

The absolute temperature of the superheated steam is $\theta = 460 + 350 + 75 = 885$.

In eq. (58), substitution of this value of θ gives for the corrective term

$$\left(\frac{150,300,000}{\theta^3} - 0.0833 \right) = 0.134.$$

$$\text{Hence } v = \frac{(85.85)(885)}{(135)(144)} - [1 + 0.00000976(135)(144)](0.134),$$

$$\text{or } v = 3.91 - 0.16 = 3.75 \text{ cubic feet.}$$

Or the volume of two pounds of steam is 7.50 cubic feet.

$$\text{By eq. (57), } v = \frac{(85.85)(885)}{(135)(144)} - 0.256 = 3.91 - 0.26 = 3.65 \text{ cubic feet.}$$

The volume for two pounds is then 7.30 cubic feet.

The per cent of difference between the values calculated by equation (57) and (58)

$$\text{is then } \frac{3.75 - 3.65}{3.75} 100 = 2.7 \text{ per cent.}$$

34. *Temperature of superheat.* At 130.3 pounds gauge 3 pounds of steam occupy 10.5 cubic feet. Find the degrees of superheat.

Solution. The volume per pound is 3.5 cubic feet. The pressure is 145 pounds absolute. The equation of Knoblauch, Linde, and Klebe (58) is capable of solution only by means of tedious approximations. The equation of Tumlirz (57) is here used.

$$\theta = \frac{pv + 0.256p}{85.85} = \frac{(145)(144)(3.5 + 0.256)}{85.85} = 913^\circ \text{ Fahrenheit absolute.}$$

From Table 3, the saturation temperature corresponding to a pressure of 145 pounds is 355.8° F. or 815° F. absolute. The superheat is therefore $913 - 815 = 98^\circ \text{F.}$

If greater accuracy is desired and justified by the precision with which p and v are known, this value of $\theta = 913$ may be substituted in the corrective term of equation (58) and then that equation solved for θ , considering the corrective term to be constant. Such a substitution gives $\theta = 885$ and 70 degrees of superheat.

35. *Specific heat of superheated steam.* Find the heat required to superheat 2 pounds of steam 75 degrees at 135 pounds absolute.

Solution. For 135 pounds the saturation temperature is 350° Fahrenheit.

From Table 4, the specific heat is found to be approximately 0.558, by cross interpolation between the four values.

pounds	113.8	142.2
392° F.	0.560	0.597
482° F.	0.532	0.552

The heat required per pound is then $(0.558)(75) = 41.8$ or 42 B. t. u. For two pounds it would require 84 B. t. u.

36. Total heat. Find the total heat contents of two pounds of steam superheated 75° Fahrenheit at a pressure of 135 pounds absolute. How much heat is required to produce this steam from feed water at a temperature of 90° Fahrenheit?

Solution. The total heat contents λ_1 of one pound of steam under the above conditions is $\lambda_1 = q_1 + r_1 + h_1$.

From Table 3, $q_1 = 321.5$, $r_1 = 869.8$, and h_1 from Problem 35 is 42 B. t. u. Hence $\lambda_1 = 1233.3$ B. t. u.

For feed water at 90° Fahrenheit, $\lambda_2 = q_2 = 58.1$ B. t. u.

Hence per pound $\lambda_1 - \lambda_2 = 1175$ B. t. u. or 2350 B. t. u. for 2 pounds.

37. Entropy. Find the entropy for the superheated steam of Problem 36, and also the increase in entropy above that of the feed water.

Solution. The entropy of superheat is

$$\phi_{s_1} = \frac{h}{\frac{\theta_0 + \theta_1}{2}} = \frac{2(42)}{810 + 885} = 0.0495.$$

From Table 3,

$$\phi_{v_1} = 1.0741, \phi_{w_1} = 0.5037.$$

Hence

$$\phi_1 = 1.6266.$$

For the feed water

$$\phi_2 = 0.1117.$$

Hence

$$\phi_1 - \phi_2 = 1.5149 \text{ or } 3.030 \text{ for 2 pounds.}$$

38. Quality. If the entropy of steam at 135 pounds absolute is 1.60, find its quality approximately.

Solution. Using the values of Problem 37, if $\phi_1 = 1.60$, then

$$\phi_{s_1} = \phi_1 - \phi_{w_1} - \phi_{v_1} = 0.0251.$$

Assuming the specific heat as 0.48, we have $h_1 = 0.48(\theta_1 - \theta_0)$.

And

$$\phi_{s_1} = \frac{2(0.48)(\theta_1 - \theta_0)}{\theta_0 + \theta_1} = 0.0229.$$

Hence

$$0.96 \theta_1 - 0.96 \theta_0 = 0.023 \theta_0 + 0.023 \theta_1.$$

Substituting

$$\theta_0 = 350 + 460 = 810 \text{ as in Problem 37,}$$

$$\theta_1 = \frac{0.983(810)}{0.937} = 850, \text{ corresponding to } 40^\circ \text{ of superheat.}$$

39. Quality after adiabatic expansion. If the steam of Problem 36 expands adiabatically to 3 pounds back pressure (absolute), find the quality of the mixture.

Solution. From Problem 37,

$$\phi_1 = 1.627.$$

From Table 3, for a pressure of 3 pounds, $\phi_{w_3} = 0.201$,

and

$$\phi_{v_2} = 1.684.$$

By equation 64,

$$x_2 = \frac{\phi_1 - \phi_{w_2}}{\phi_{v_2}} = \frac{1.627 - 0.201}{1.684} = 0.846.$$

40. *Intrinsic energy.* Find the intrinsic energy of the steam of Problem 36.

Solution. By equation 63, the intrinsic energy ϵ_1 is

$$A\epsilon_1 = q_1 + r_1 + h_1 - Ap_1(v_1 - \sigma) = \lambda_1 - Ap_1(v_1 - \sigma).$$

From Problem 33, the specific volume is 3.75 cubic feet, and from Problem 36 the total heat is $\lambda_1 = 1233$.

Therefore

$$A\epsilon_1 = 1233 - \frac{(135)(144)}{778}(3.75 - \sigma) = 1139 \text{ B. t. u. per pound,}$$

or

$$\epsilon_1 = (778)(1139) = 885,000 \text{ foot pounds per pound of steam.}$$

For 2 pounds the energy is 1,770,000 foot pounds.

41. *Work of adiabatic expansion.* Find the work done in the adiabatic expansion of Problem 39.

Solution. For state 1, $p_1 = 135$, $t_1 = 425$, $\epsilon_1 = 885,000 \text{ ft. lb.}$

For state 2, $p_2 = 3$, $x_2 = 0.846$, $q_2 = 109.6$, $\rho_2 = 946.4$,

and hence $\epsilon_2 = \frac{1}{A}(q_2 + x_2\rho_2) = (778)(911) = 707,000 \text{ ft. lb.}$

The work of the adiabatic expansion is

$$w = \epsilon_1 - \epsilon_2 = 885,000 - 707,000 = 178,000 \text{ foot pounds per pound.}$$

For 2 pounds of steam the work is 356,000 foot pounds.

42. *Gain by superheating.* The steam at cut-off in a certain engine cylinder has a quality of 80 per cent. It is found that super-

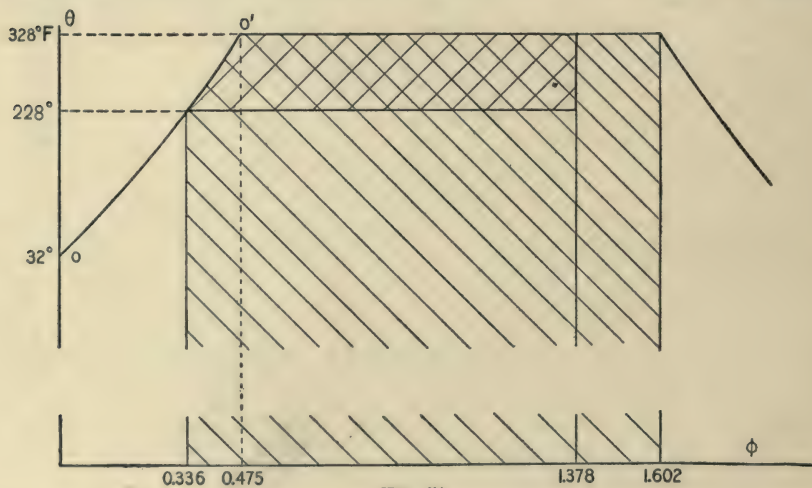


FIG. 51

heating the steam 120 degrees makes the quality unity at cut-off. The entering steam is at a pressure of 100 pounds absolute. The

back pressure is 20 pounds absolute. The specific heat may be taken at 0.52. What is the gain in efficiency due to superheating? Plot the temperature entropy relations for both cases. Assume adiabatic expansion.

Solution. The θ - ϕ plots for the condition of 80 per cent quality and 100 per cent quality are shown to scale in Figures 51 and 52 respectively. The useful work is repre-

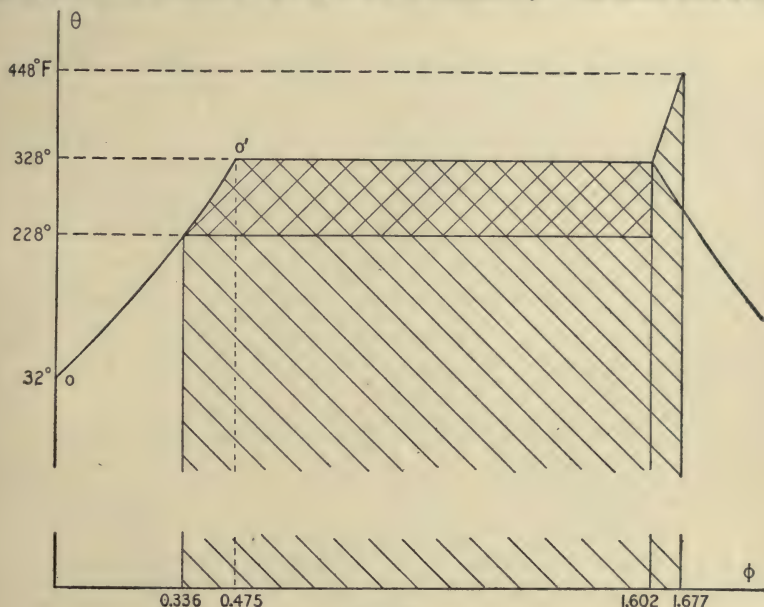


FIG. 52

sented in heat units by the doubly cross-hatched area in each case. The entire cross-hatched area represents the heat supplied.

The useful work $q_1 - q_2$ is found numerically by assuming the water line oo' a straight line and writing

$$q_1 - q_2 = \left[\frac{1}{2}(\phi_{w_1} - \phi_{w_2}) + x_1 \phi_{s_1} \right] (\theta_1 - \theta_2).$$

The useful work for 80 per cent quality is then

$$q_1 - q_2 = \left[\frac{1}{2}(0.4748 - 0.3362) + 0.80(1.1273) \right] (327.9 - 227.9) = 97.1 \text{ B. t. u.}$$

Similarly, for 100 per cent quality,

$$q_1' - q_2' = 119.7 \text{ B. t. u.}$$

The input is $q_1 = q_1 + r_1 - q_2 = 298.5 + 887.6 - 196.4 = 989.7$ B. t. u. for the first case. For the superheated steam $q_1' = q_1 + r_1 + h_1 - q_2$ where h_1 is approximately (0.52) (120) or 62.4 B. t. u. The input in the second case is therefore $1186 + 62 = 1052$ B. t. u.

The efficiency is then $\frac{97.1}{989.7} = 9.81$ per cent for wet steam and $\frac{119.7}{1052} = 9.60$ per cent for dry steam at cut-off.

For the conditions of the present problem the gain in efficiency due to superheating sufficiently to avoid cylinder condensation during admission is 1.54 per cent in 9.81 per cent or an increase of the efficiency of 15.7 per cent.

CHAPTER V

FLOW OF STEAM AND GASES

64. Formula of de Saint-Venant. Consider a compressible body B (Figure 53) acted upon by forces F_1 and F_1' . Let dz_1 represent the distance of motion of the point of application of F_1 , and similarly dz_1' represent the motion of the point of application of F_1' . The motion dz_1' is of course negative if there is expansion as shown in the figure. Then the gain in energy of the entire mass B is equal to the algebraic sum of the work of the acting forces, or $F_1 dz_1 - F_1' dz_1'$.

If we now consider two adjacent elements B and B' of an aeriform body, as shown in Figure 54, it is evident that the force F_1' , acting upon the element B , is equal and opposite to the force F_2' acting upon B' . Hence, for this case, the total work done upon the entire body, B plus B' , is given by the expression

$$dO = F_1 dz_1 - F_2 dz_2. \quad \dots \dots \dots (i)$$

This is the gain in energy of the entire mass, and will be represented in subsequent discussions by the symbol dO . The total change for an extended body will evidently be an expression of the same form as (i).

Consider a steady frictionless flow of steam in a non-conducting channel C , shown in Figure 55. The transformation of the steam will then be isentropic. Let the cross-section at A be denoted by a_1 and at B by a_2 , the corresponding pressures on these areas by p_1 and p_2 . Then, if F_1 and F_2 represent the forces acting upon a_1 and a_2 respectively,

$$F_1 = p_1 a_1 \quad \text{and} \quad F_2 = p_2 a_2. \quad \dots \dots \dots (ii)$$

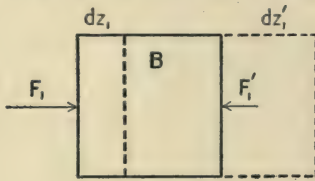


FIG. 53

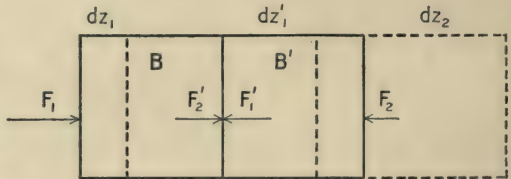


FIG. 54

Since the flow is steady, there is the same amount of steam in pounds passing *A* as is passing *B* in the same time. If *M* represents the weight of steam passing *A* or *B* in one second, and if the velocities at *A* and *B* are respectively v_1 and v_2 in feet per second, then this condition of steady flow may be expressed as

$$Mv_1 = a_1v_1$$

and $Mv_2 = a_2v_2$, . (iii)

where v_1 and v_2 are the specific volumes of the

steam in the states in which it passes *A* and *B* respectively. Equating *M* from the two equations of (iii) gives

$$M = \frac{a_1v_1}{v_1} = \frac{a_2v_2}{v_2}. \quad . \quad . \quad . \quad . \quad . \quad (65)$$

This equation is known as the "equation of continuity," and holds for any fluid in a condition of steady flow.

Substitution of this equation of continuity in the equation (i) for the increase of energy gives an equation expressing this increase *dO* in terms of the pressure and specific volumes at *A* and *B*. Thus, if we consider the flow for a small fraction of time *dt*, then

$$dz_1 = v_1dt \quad \text{and} \quad dz_2 = v_2dt. \quad . \quad . \quad . \quad . \quad . \quad (iv)$$

In the time *dt* the mass of steam passing *A* or *B* may be expressed as

$$dm = Mdt. \quad . \quad . \quad . \quad . \quad . \quad (v)$$

Hence, for this mass, *dm*, the equation of continuity (48) becomes

$$dm = \frac{a_1v_1dt}{v_1} = \frac{a_2v_2dt}{v_2}. \quad . \quad . \quad . \quad . \quad . \quad (vi)$$

Substitution from equations (ii), (iv), (v), and (vi) in equation (i) gives

$$dO = M(p_1v_1 - p_2v_2)dt. \quad . \quad . \quad . \quad . \quad . \quad (vii)$$

The total increase of energy expressed by equation (vii) is partly external kinetic energy of the entire mass of steam, *Mdt*, and partly

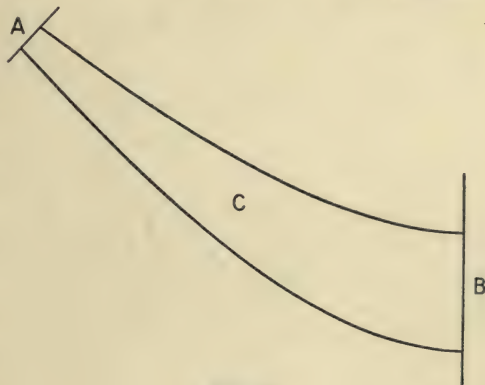


FIG. 55

intrinsic or molecular energy. Representing these components by dK and dE respectively, we have

$$dO = dK + dE. \quad \dots \quad (\text{viii})$$

The external kinetic energy expressed in terms of the gain in velocity of the entire mass of the steam is

$$dK = \frac{Mdt}{2g} (v_2^2 - v_1^2). \quad \dots \quad (\text{ix})$$

The increase of intrinsic energy dE is equal to the external work done in compressing the steam. If there is expansion, as in the case considered, the increase in intrinsic energy is negative, and is numerically equal to the work done by the gas in expanding. Thus,

$$dE = -Mdt \int_{v_1}^{v_2} p dv. \quad \dots \quad (\text{x})$$

Substitution from equations (viii), (ix), and (x) in equation (vii) gives

$$Mdt(p_1 v_1 - p_2 v_2) = Mdt \frac{v_2^2 - v_1^2}{2g} - Mdt \int_{v_1}^{v_2} p dv. \quad \dots \quad (\text{xi})$$

Or dividing by Mdt and transposing, we have

$$\frac{v_2^2 - v_1^2}{2g} = p_1 v_1 - p_2 v_2 + \int_{v_1}^{v_2} p dv. \quad \dots \quad (\text{xii})$$

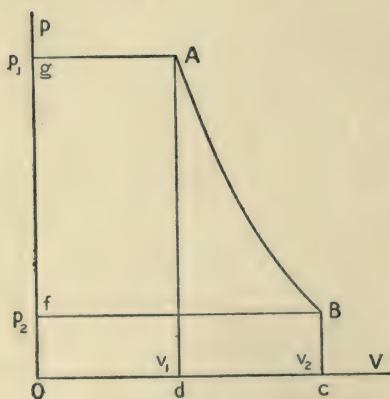


FIG. 56

From Figure 56, which represents the pressure-volume plot of the frictionless adiabatic expansion under consideration, it is evident that $p_1 v_1$ is equal to the area $Adog$, $p_2 v_2$ to the area $Bcof$, and $\int_{v_1}^{v_2} p dv$ to the area $ABcd$. Hence equation (xii) becomes

$$\frac{v_2^2 - v_1^2}{2g} = \text{area } ABfg = \int_{p_2}^{p_1} v dp. \quad (66)$$

This equation may be expressed in terms of the pressures, and specific volumes at A and B if the frictionless adiabatic expansion may be taken as represented by an equation of the form

$$pv^n = p_1 v_1^n = p_2 v_2^n. \quad \dots \quad (\text{xiii})$$

Then

$$\int_{p_2}^{p_1} v dp = \frac{n}{n-1} (p_1 v_1 - p_2 v_2). \quad \dots \quad (\text{xiv})$$

Therefore

$$\frac{v_2^2 - v_1^2}{2g} = \frac{n}{n-1} (p_1 v_1 - p_2 v_2). \quad \dots \quad (67)$$

In this equation, which is due to de Saint-Venant,¹ p and v are specific pressures and specific volumes in pounds per square foot and cubic feet per pound, respectively. The kinetic energy term represents the increase in external kinetic energy *per pound of steam*. The exponent n is given by Zeuner as $1.035 + 0.1x$ for wet steam of quality x ; that is, 1.135 for steam initially dry, and n is 1.3 for superheated steam.

65. Alternative Expression for the Formula of de Saint-Venant. If in equation (viii) of the preceding section we substitute for dK from equation (ix) of the same section, but for dE write $(\epsilon_2 - \epsilon_1) Mdt$, where ϵ_1 and ϵ_2 are the values of the intrinsic energy per pound of steam at A and at B respectively, we have

$$M(p_1 v_1 - p_2 v_2) dt = Mdt \frac{v_2^2 - v_1^2}{2g} + (\epsilon_2 - \epsilon_1) Mdt,$$

or dividing by Mdt and transposing, we obtain for a pound of a mixture undergoing a frictionless adiabatic expansion the relation

$$p_1 v_1 + \epsilon_1 + \frac{v_1^2}{2g} = p_2 v_2 + \epsilon_2 + \frac{v_2^2}{2g}. \quad \dots \quad (68)$$

Equations (66) and (67) of Section 64 and equation (68) given above as well as the equation of continuity (65) hold equally well for all fluids. Thus if the fluid is a gas, the value of the exponent n is of course $\kappa = \frac{c_p}{c_v} = 1.4$ for diatomic mixtures,

$$\frac{v_2^2 - v_1^2}{2g} = \frac{1.4}{0.4} (p_1 v_1 - p_2 v_2). \quad \dots \quad (69)$$

If the fluid is incompressible as water, then $v_1 = v_2$ and for an adiabatic change $\epsilon_1 = \epsilon_2$,

$$\text{whence} \quad \frac{v_2^2 - v_1^2}{2g} = (p_1 - p_2) \sigma. \quad \dots \quad (70)$$

¹ A more rigorous demonstration from which the above is derived is to be found in Lowenstein's translation of Stodola, "Steam Turbines," 2d ed., pp. 4-8.

For steam three cases may be distinguished; namely, (1) expansion through the channel from a condition of saturated steam of quality x_1 to a condition of quality x_2 , (2) expansion from a superheated condition to a saturated condition of quality x_2 , and (3) expansion entirely within the superheated region. For all these cases the expression for the gain in kinetic energy per pound of the steam mixture reduces to a simple form which will be derived in the following section.

66. Zeuner's Formula for Kinetic Energy. In equation (68) transposition gives

$$\frac{v_2^2 - v_1^2}{2g} = (p_1 v_1 + \epsilon_1) - (p_2 v_2 + \epsilon_2).$$

Considering an expansion which takes place entirely within the region of wet steam, we have

$$v_1 = x_1 u_1 + \sigma, \text{ and } v_2 = x_2 u_2 + \sigma. \quad \dots \quad (i)$$

Therefore, neglecting σ , which in general will make a difference of less than half of one per cent and substituting from equation (i) in equation (68), we may write

$$(\epsilon_1 + x_1 p_1 u_1) - (\epsilon_2 + x_2 p_2 u_2) = \frac{v_2^2 - v_1^2}{2g}. \quad \dots \quad (ii)$$

But

$$\epsilon_1 = \frac{1}{A} (q_1 + x_1 p_1) \text{ and } \epsilon_2 = \frac{1}{A} (q_2 + x_2 p_2). \quad \dots \quad (iii)$$

Substituting from equation (iii) in equation (ii) and remembering that

$$\lambda_1 = q_1 + x_1 p_1 + x_1 A p_1 u_1 \text{ and } \lambda_2 = q_2 + x_2 p_2 + x_2 A p_2 u_2, \quad \dots \quad (iv)$$

gives

$$A \frac{v_2^2 - v_1^2}{2g} = \lambda_1 - \lambda_2. \quad \dots \quad (71)$$

Similarly, consider the steam superheated in state 1 and wet in state 2. For superheated steam, equation (63) of Section 60 gives

$$A \epsilon_1 = q_1 + r_1 + c_p (\theta_1 - \theta_0) - A p_1 (v_1 - \sigma). \quad \dots \quad (v)$$

Or neglecting σ as before,

$$\epsilon_1 + p_1 v_1 = \frac{1}{A} [q_1 + r_1 + c_p (\theta_1 - \theta_0)] = \frac{\lambda_1}{A}. \quad \dots \quad (vi)$$

Substituting for $(\epsilon_1 + p_1 v_1)$ from equation (vi) and for $(\epsilon_2 + p_2 v_2)$ from equations (iii) and (iv) in equation (ii) gives

$$A \frac{v_2^2 - v_1^2}{2g} = \lambda_1 - \lambda_2. \quad (71)$$

For the case of an expansion entirely within the superheated region similar reduction leads to the same equation as (71).

67. Kinetic Energy of an Irreversible Adiabatic Expansion. If, however, the decrease in total heat contents is not entirely converted into external energy, but is partly expended in external work of amount w , while some heat of amount q is transferred from the substance to external bodies, we have to consider the general case of an expansion with heat changes and external work. Then equation (71) becomes

$$\lambda_1 - \lambda_2 = A \frac{v_2^2 - v_1^2}{2g} + q + Aw. \quad (72)$$

If there is work done against friction, there must be added to the right-hand side of equation (71) a term y representing the work per pound done against friction and expressed in heat units. If q is zero, then the expansion is adiabatic; if w is zero, then there is a free expansion; but if y is not zero, then there is an adiabatic expansion similar to that described in the case of the "porous plug experiment" of Section 35, which is irreversible and during which there is an increase in entropy. This case of the irreversible expansion of steam in a non-conducting channel will be discussed in the latter part of the following section.

68. Loss of Kinetic Energy Due to Friction. If there is friction during the expansion from a pressure of p_1 to a pressure of p_2 , the gain in kinetic energy will be less than it would have been for a similar frictionless expansion. Denote by v_2' the final velocity for a flow with friction for the same range of pressures as gives v_2 for a frictionless expansion. Then denoting by l the loss of kinetic energy per pound of the mixture expressed in thermal units, we have

$$l = A \frac{v_2^2 - v_1^2}{2g} - A \frac{v_1'^2 - v_1^2}{2g} = A \frac{v_2^2 - v_2'^2}{2g}. \quad (73)$$

Denoting by λ_2' the total heat contents, after flow with friction, it is clear that

$$l = (\lambda_1 - \lambda_2) - (\lambda_1 - \lambda_2') = \lambda_2' - \lambda_2, \quad (74)$$

where λ_1 and λ_2 have their former significance and refer to a frictionless flow.

As the expansion takes place, a portion of the kinetic energy of the mixture is converted into heat by friction and immediately afterwards taken up by the mixture at a lower temperature and availability and again converted into kinetic energy. Since the heat equivalent of the friction work is thus continually being returned to the mixture, the actual loss in external kinetic energy due to the friction is less than the total work done against frictional forces. The heat corresponding to this friction work has been represented in the previous section by y .

This heat y when returned to the steam or fluid in successive infinitesimal amounts dy produces changes in the intrinsic energy of the mixture of amount to be represented by $d\epsilon$ and also produces an amount of external work of value $p dv$. That is,

$$dy = A d\epsilon + A p dv. \quad (i)$$

But $A d(pv) = A p dv + A v dp. \quad (ii)$

Hence $dy = A d\epsilon + A d(pv) - A v dp. \quad (iii)$

In equation (iii) there may be substituted as in Section 66

$$A d\epsilon + A d(pv) = A(\epsilon_1 + p_1 v_1) - A(\epsilon_2 + p_2 v_2) = d\lambda. \quad (iv)$$

Hence $dy = d\lambda - A v dp. \quad (75)$

By integration from state 1 to state 2

$$y = \lambda_2 - \lambda_1 - A \int_{p_1}^{p_2} v dp, \quad (76)$$

where equation (76) is a general relation for the total amount of friction work.

If now the expansion is frictionless and takes place from state 1 to state 2, we have $y = 0$ and

$$\lambda_2 - \lambda_1 = A \int_{p_1}^{p_2} v dp. \quad (v)$$

For the expansion with friction between the states 1 and 2' at which the pressure is the same as for state 2 but at which the volume v' is different,

$$y = \lambda_2' - \lambda_1 - A \int_{p_1}^{p_2'} v' dp. \quad (vi)$$

Subtracting equations (v) and (vi) gives

$$y = \lambda_2' - \lambda_2 - A \left(\int_{p_1}^{p_2'} v' dp - \int_{p_1}^{p_2} v dp \right),$$

or,

$$y = \lambda_2' - \lambda_2 - A \int_{p_1}^{p_2} (v' - v) dp. \quad (vii)$$

In Figure 57 is shown the p - v plot for these two expansions. From this it is evident that

$$\int_{p_1}^{p_2} (v' - v) dp = \text{area } 122',$$

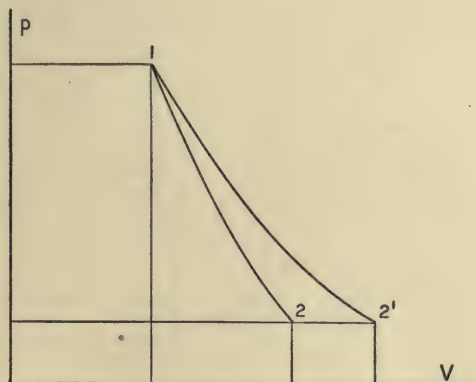


FIG. 57

or reversing the order of the integration and thus changing the algebraic sign,

$$-\int_{p_1}^{p_2} (v_1' - v) dp = + \int_{p_2}^{p_1} (v' - v) dp = \text{area } 12'2.$$

Therefore

$$y = \lambda_2' - \lambda_1 + A \times \text{area } 12'2. \quad \dots (77)$$

The total work done against friction is therefore larger than the loss of kinetic energy due to friction by the amount of work corresponding to the area on the p - v plot included by the line representing the actual expansion, the line representing an adiabatic expansion, and the line of constant final pressure.

The temperature entropy plot for these expansions is shown in Figure 58. For a frictionless adiabatic expansion the area $ab12$ represents in heat units the increase in kinetic energy per pound. If all the kinetic energy of the expansion were converted into friction and this friction into heat, the line 13 representing the curve of constant total heat contents through 1 would be the path followed by the state point of the steam mixture. The area $23fd$ would then be equal to the area $ab12$. The path followed by the state point during the actual irreversible expansion through a channel like a turbine nozzle is some line between the constant heat line 13 and the reversible adiabatic 12 , such as $12'$. For the path $12'$ the

friction work is $12'ed$, of which $22'ed$ represents the loss in kinetic energy l . The net increase of kinetic energy during an irreversible

adiabatic expansion is then expressed in heat units by the area $12ab$ diminished by the area $22'ed$.

The loss l is comparatively small. For nozzles such as are used in turbines which are not more than 2 inches long l is about 5 to 8 per cent of the energy which would be developed in a frictionless flow. For larger nozzles of length 4 to 6 inches and with diameters at the narrowest point of from one quarter to three eighths of an inch l is from 10 to 15 per cent.¹

69. Flow through an Orifice.

If a fluid flows from a reservoir of such size that v_1^2 may be neglected in comparison with v_2^2 (as is the case in the flow of steam from a boiler into a pipe), then equations (71) and (65) may be used to find the weight of fluid passing through the orifice if the second state be taken at the narrowest point or throat

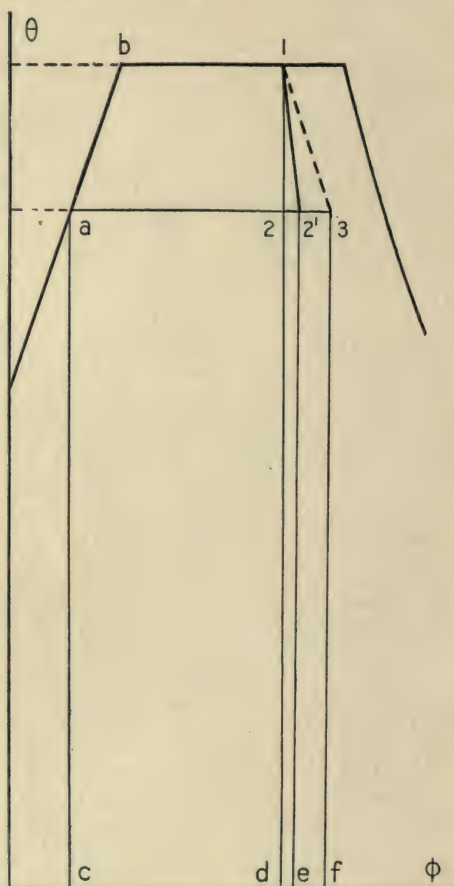


FIG. 58

of the short tube or orifice. The reason for specifying that the second state shall be that at the throat is evident from the following experimental fact and also may be seen from the curves of Problem 53.

Experiments of Napier show that the weight of steam discharged through an orifice increases for a constant pressure p_1 with a reduction of the back pressure p_2 until p_2 is $0.58 p_1$, after which further reduction does not increase the weight discharged. In other words,

¹ The discussion follows Stodola, "Steam Turbines," pp. 48-63.

the pressure in the orifice for steam is always such as to give a maximum value for the weight of steam discharged per second. For steam, therefore, the pressure at the orifice may be taken as $0.58 p_1$ provided that $p_1 > 1.73 p_2$.

Rankine therefore announced from his study of the experimental results obtained by Napier the following empirical formulas for calculating the flow of steam through an orifice. The symbols have the significance of the preceding discussion.

Where $p_1 \geq \frac{5}{3} p_2$,

$$M = a \frac{p_1}{70}; \quad (78)$$

and

where $p_1 < \frac{5}{3} p_2$,

$$M = a \frac{p_2}{42} \sqrt{\frac{3(p_1 - p_2)}{2 p_2}}. \quad (79)$$

It is to be noticed that these formulas of Napier and the assumption of the pressure in the throat as $0.58 p_1$ are only approximations. Since no flow is rigorously frictionless or adiabatic and the variations from these ideal conditions depend upon the orifice or tube employed when greater accuracy than about 2 per cent is required, the given nozzle should be calibrated by weighing the steam passing through it for various values of p_1 and p_2 .

70. Fliegner's Formulas for the Flow of Air. For the flow of air through a rounded orifice, where the pressures are p_1 and p_2 on the two sides of the orifice and the area is a , the experiments of Fliegner show that the weight M of air in pounds per second is given by the following formulas:

where $p_1 > 2 p_2$

$$M = 0.530 a \frac{p_1}{\sqrt{\theta_1}}; \quad (80)$$

where $p_1 < 2 p_2$,

$$M = 1.06 a \sqrt{\frac{p_2(p_1 - p_2)}{\theta_1}}. \quad (81)$$

In these formulas θ_1 is the absolute temperature Fahrenheit of the air in the reservoir. The pressure and the area are to be measured in any similar units, as either in pounds per square foot and in square feet, or the pressures in pounds per square inch and the area in square inches.

Solution. From Table 3, for 100 pounds, $q_1 = 298.5$, $r_1 = 887.6$, $\phi_{w_1} = 0.4748$, $\phi_{v_1} = 1.1273$, $s_1 = 4.432$; for $p_2 = 14.7$ pounds, $q_2 = 180.3$, $r_2 = 969.7$, $\phi_{w_2} = 0.3125$, $\phi_{v_2} = 1.4441$, $s_2 = 26.78$.

Hence the quality after adiabatic expansion without friction is

$$x_2 = \frac{\phi_{w_1} + \phi_{v_1} - \phi_{w_2}}{\phi_{v_2}} = \frac{0.4748 + 1.1273 - 0.3125}{1.4441} = 0.893.$$

(a) $v_1 = s_1 = 4.432$; $v_2 = x_2 s_2$, approximately, or $(0.893)(26.78) = 23.92$.

By equation (67),

$$\frac{v_2^2}{2g} = \frac{1.135}{0.135} [(100)(144)(4.432) - (14.7)(144)(23.92)] = 111,100 \text{ foot pounds.}$$

Therefore

$$v_2 = 2670 \text{ feet per second.}$$

(b) $\lambda_1 = 298.5 + 887.6 = 1186.1$, $\lambda_2 = 180.3 + x_2(969.7) = 1046.4$.

Hence, in equation (71), $J(\lambda_1 - \lambda_2) = 108,800$ foot pounds,

the term $\sigma(p_1 - p_2)$ is of value $(0.016)(144)(100 - 14.7) = 196$ foot pounds, and is negligible.

Therefore, neglecting this term, we have

$$\frac{v_2^2}{2g} = 108,800,$$

or

$$v_2 = 2650 \text{ feet per second.}$$

(c) The term neglected above evidently introduces an error of 0.2 per cent in the value of the kinetic energy, or 0.1 per cent in the value of the final velocity.

44. Kinetic energy of superheated steam. Given one pound of steam at 100 pounds absolute and 120° Fahrenheit of superheat, find the final velocity acquired in expanding along a reversible adiabatic to a back pressure of 14.7 pounds absolute, (a) by using equation (67), (b) by using equation (71), as in Problem 43.

Solution. From Table 3, the values of the constants for these pressures may be obtained, as given in the preceding problem. For 120° of superheat c_p may be taken as 0.53. Then

$$h_1 = c_p(\theta_1 - \theta_0) = 0.53(120) = 63.6 \text{ B. t. u.,}$$

and

$$\phi_{s1} = \frac{63.6}{\frac{1}{2}(788 + 908)} = 0.0750.$$

The specific volume is found by equation (58) to be 5.27 cubic feet.

The quality after expansion is, then,

$$x_2 = \frac{\phi_{w_1} + \phi_{v_1} + \phi_{s1} - \phi_{w_2}}{\phi_{v_2}} = \frac{0.4748 + 1.1273 + 0.0750 - 0.3125}{1.4441} = 0.945.$$

(a) Since the expansion extends from the superheated into the saturated region, it is necessary to divide the expression of equation (67) for the increase of kinetic energy into two parts, one representing the increase while the steam is superheated and calculated by using $n = 1.3$, and one representing the increase while the steam is saturated and calculated, as in Problem 43, by using $n = 1.135$. This demands a knowledge of the pressure and volume corresponding to the intersection of the isentropic line

with the curve of constant steam weight. This is most easily found from the θ - ϕ chart of page 86 to be 40 pounds for the pressure and 10.5 cubic feet for the specific volume. It may be found approximately by using the values for the entropy at 30 and 50 pounds, and assuming that the rate of change of the entropy is proportional to the rate of change of the pressure. Such an interpolation gives the pressure as 39 pounds. The more accurate value of 40 pounds will be used in this solution. Substitution of these values in equation (67) then gives

$$\begin{aligned} \frac{v_2^2}{2g} &= \frac{1.3}{0.3} [(100)(144)(5.27) - (40)(144)(10.5)] \\ &\quad + \frac{1.135}{0.135} [(40)(144)(10.5) - (14.7)(144)(25.3)] \\ &= 67,000 + 58,100 = 108,000 \text{ foot pounds.} \end{aligned}$$

Hence $v_2 = 2830$ feet per second.

(b) Solution by equation (71) is similar to that of Problem 43 except that λ_1 is greater by the superheat h_1 .

$$\lambda_1 = 1186.1 + 63.6 = 1249.7, \quad \lambda_2 = 180.3 + (0.945)(969.7) = 1096.5.$$

$$\frac{v_2^2}{2g} = J(\lambda_1 - \lambda_2) = 778(153.2) = 119,200 \text{ foot pounds.}$$

Hence $v_2 = 2770$ feet per second.

Discussion. The deviation between solutions (a) and (b) is due to the inaccuracy with which n is accompanied. The method of (b) is the more accurate and in general shorter of solution.

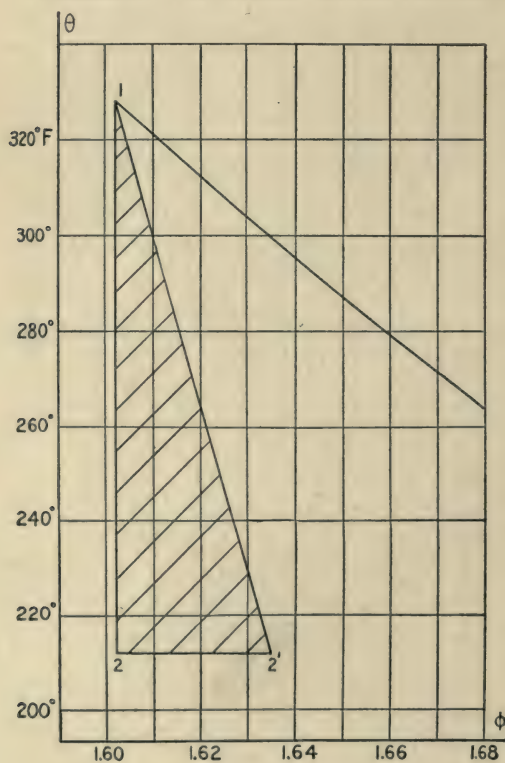


FIG. 59

45. *Loss due to friction.* In Problem 43 the velocity for a frictionless adiabatic expansion has already been found. Taking the same values for the pressures but assuming a loss of 16 per cent of the energy available for increasing the kinetic energy, find the final velocity. Sketch the θ - ϕ plot and find the total work done against friction. Assume the plot of the actual expansion line on the θ - ϕ diagram to be a straight line.

Solution. The increase in kinetic energy of the steam for

a reversible adiabatic has been determined as 108,800 foot pounds, and the final quality as 0.893. The loss, l , in kinetic energy is then $\frac{1}{10} (108,800) = 17,400$ foot pounds = 22 B. t. u. The absolute temperature corresponding to the back pressure is $212 + 460 = 672$ degrees. The increase in entropy due to the friction is then $\frac{22}{672} = 0.0328$ and is represented in Figure 59 by the line 22'.

The area 12'2 may now be found approximately by multiplying half this increase in entropy by the difference in temperature between points 1 and 2; that is,

$$\text{area } 12'2 = \frac{1}{2} (0.0328) (328 - 212) = 1.9, \text{ or approximately } 2 \text{ B. t. u.}$$

The work done against friction, y , is by equation (77) equal to the sum of this area and l . Hence $y = 22 + 2 = 24$ B. t. u., or 18,700 foot pounds.

The final velocity is of course one half of 16 per cent smaller than that for a frictionless flow, since the energy available is 16 per cent smaller. Hence $v_2 = 24400$ feet per second.

46. Flow of air. Find the weight of air passing per second through a rounded orifice of area 0.1 square inch, if the pressures on the two sides of the orifice are 45 and 14.7 pounds absolute, (a) by equation (67), (b) by equation (80). The temperature of the air within the reservoir is 80° Fahrenheit.

Solution. (a) The pressure p_t in the throat, where the area is given above, is to be taken according to Fliegner's experiments as 0.577 of p_1 , or for this case $p_t = (0.577)(45)$. The specific volume of the air in the reservoir is, by equation (13),

$$v_1 = \frac{80 + 460}{460} \frac{14.7}{45} 12.39 = 4.75 \text{ cubic feet.}$$

The specific volume after expansion is, by eq. (32),

$$v_2 = \left[\frac{45}{(0.577)45} \right]^{\frac{1}{1.40}} (4.75) = 7.02 \text{ cubic feet.}$$

Hence, by eq. (65),

$$M = \frac{0.1}{(144)(7.02)} \sqrt{\frac{1.40}{0.40} 2g[(45)(144)(4.75) - (0.577)(144)(7.02)]} = 0.100 \text{ pounds per second.}$$

(b) By equation (80), since $\theta_1 = 80 + 460 = 540$,

$$M = (0.530)(0.1) \frac{45}{\sqrt{540}} = 0.102 \text{ pound per second.}$$

47. Napier's formula. Given steam of 95 per cent quality expanding from 120 pounds absolute to atmospheric pressure of 14.7 pounds through an orifice of 0.1 square inch area. Assume the pressure in the throat to be $0.58 p_1$ or 70 pounds absolute. Find the weight of steam per second, (a) by equation (71), (b) by equation (78).

Solution. (a) From Table 3, for 120 pounds, $q_1 = 312.3$, $r_1 = 876.9$, $\phi_{w_1} = 0.4922$, $\phi_{v_1} = 1.0951$. For $p_t = 70$ pounds, $q_t = 272.9$, $r_t = 906.6$, $\phi_{w_t} = 0.4418$, $\phi_{v_t} = 1.1892$, $s_t = 6.199$.

Hence

$$x_t = \frac{0.4922 + (0.95)(1.0951) - 0.4418}{1.1892} = 0.917, \quad v_t = (0.917)(6.199) = 5.69.$$

And

$$\lambda_1 - \lambda_t = 312.3 + (0.95)(876.9) - 272.9 - (0.917)(906.6) = 41.1.$$

Then

$$M = \frac{a_t v_t}{v_t} = \frac{0.1 \sqrt{2g778(41.1)}}{(144)(5.69)} = 0.175 \text{ pound per second.}$$

$$(b) \quad M = \frac{(0.1)(120)}{70} = 0.172 \text{ pound per second.}$$

48. *Napier's formula.* Compare the weight of steam of quality 95 per cent passing through an orifice 0.1 square inch in area from an initial pressure of 20 pounds absolute to atmospheric pressure of 14.7 pounds, as calculated (a) by equation (71), (b) by equation (79).

Solution. (a) From Table 3, for $p_1 = 20$ pounds, $q_1 = 196.4$, $r_1 = 959.4$, $\phi_{w_1} = 0.3362$, $\phi_{v_1} = 1.3957$. For $p_2 = 14.7$ pounds, $q_2 = 180.3$, $r_2 = 969.7$, $\phi_{w_2} = 0.3125$, $\phi_{v_2} = 1.4441$, $s_2 = 26.78$.

Hence

$$x_2 = \frac{0.3362 + (0.95)(1.3957) - 0.3125}{1.4441} = 0.9345 \text{ or } 0.934.$$

$$v_2 = (0.934)(26.78) = 25.0.$$

$$\lambda_1 - \lambda_2 = 196.4 + (0.95)(959.4) - 180.3 - (0.934)(969.7) = 21.9 \text{ B. t. u.}$$

Therefore

$$M = \frac{0.1 \sqrt{2g778(21.9)}}{(144)(25.0)} = 0.0291 \text{ pound per second.}$$

$$(b) \quad M = \frac{0.1(14.7)}{42} \sqrt{\frac{3(20 - 14.7)}{2(14.7)}} = 0.0257 \text{ pound per second.}$$

49. *Injector (non-lifting).* An injector, as the name implies, is a device for injecting water into a boiler against the boiler steam pressure. This is accomplished by a mechanism the principles of which are evident from Figure 60. To start the injector, steam is

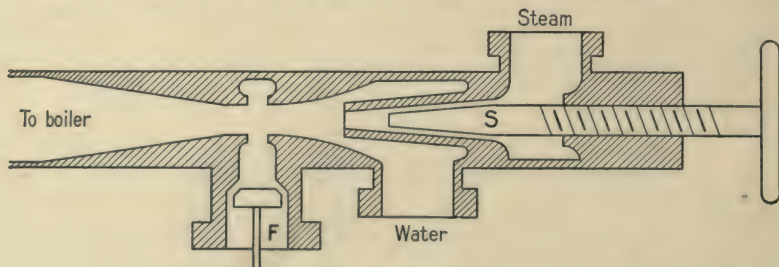


FIG. 60

allowed to flow from the steam delivery pipe through the injector and out of the overflow *F*, thus blowing out any condensed water

and insuring practically dry steam. The steam valve S is then turned off and the water valve turned on. When the water appears at the overflow the steam valve and the valve in the delivery tube to the boiler are both turned on. The entering steam is condensed by contact with the colder feed water and converts some of its heat energy into mechanical kinetic energy, forcing feed water and the condensed steam into the boiler. The partial vacuum formed by the steam and water rushing by the overflow valve draws it shut.

Calculate the number of pounds of water forced into a boiler against a pressure of 120 pounds absolute by one pound of steam from the same boiler. Neglect the velocity of the entering water. Assume the steam 98 per cent dry, the temperature of the water supplied to the injector as 70° Fahrenheit, and the temperature of the water leaving the injector as 170° Fahrenheit. Assume that the kinetic energy of the entire mass of water leaving the injector is one per cent of the total heat contents of one pound of steam.

Solution. Let there be m pounds of feed water entering the injector for every pound of steam. The total energy of the entering water plus that of the entering steam must equal the total energy of the two combined as they leave the injector. That is,

$$m\lambda_{\text{water}} + K. E._{\text{water}} + \lambda_{\text{steam}} + K. E._{\text{steam}} = (1 + m)\lambda_{(\text{water} + \text{steam})} + K. E._{(\text{water} + \text{steam})},$$

where

$$\lambda_{\text{water}} = 39 \text{ B. t. u.}; K. E._{\text{water}} \text{ is negligible}; \lambda_{\text{steam}} = q + xr = 1172 \text{ B. t. u. per pound.}$$

$$K. E._{(\text{water} + \text{steam})} = 12 \text{ B. t. u.}; \text{ and } \lambda_{(\text{water} + \text{steam})} = 138 \text{ B. t. u. per pound.}$$

$$\text{Hence } 39m + 1172 = (1 + m)138 + 12 \text{ or } m = 10 \text{ pounds approximately.}$$

For practical purpose it is sufficient to neglect all kinetic energy terms and write

$$m = \frac{(xr)s + q_s - q_{w+s}}{q_{w+s} - q_w} = \frac{1172 - 138}{138 - 39} = 10 \text{ pounds approximately.}$$

50. Peabody throttling calorimeter. The Peabody throttling calorimeter employs the phenomena of an irreversible adiabatic expansion in the determination of steam quality. It consists of a reservoir A shown in Figure 61 to which the steam of which the quality is desired is admitted by a small pipe b . A pressure gauge at f and a thermometer at e record the temperature and pressure of the steam after it has expanded to fill the reservoir. The valve d admits of the free passage of steam from the calorimeter. The entire

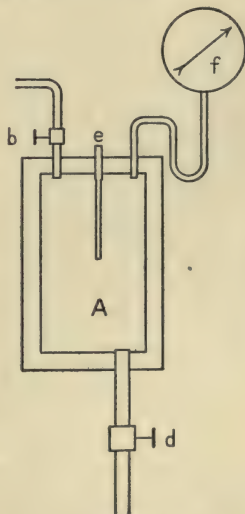


FIG. 61

reservoir is surrounded by a heat insulator. For a test, the valve b which connects the reservoir with the main steam pipe, in which it is desired to know the quality, is opened slightly, and thus admits steam throttled down to a pressure of 5 to 15 pounds above that of the atmosphere to which d is connected.

If the boiler pressure is $p_1 = 80.3$ pounds gauge, the pressure p_2 within the calorimeter is 5.3 pounds gauge, the atmospheric pressure is 14.7 pounds, and the temperature within the calorimeter is 278° Fahrenheit, find the quality of the steam. Sketch the temperature entropy diagram for this process.

Solution. There is assumed to be an irreversible adiabatic expansion, in which as fast as the heat of the entering steam is converted into kinetic energy of motion, this energy is transformed into work of friction and eddy currents, and is returned at a slightly lower temperature to the steam. The change thus takes place at constant heat contents. Hence $\lambda_1 = \lambda_2$. If the steam is nearly saturated, the constant heat line will take it into the superheated region for lower temperatures and pressures. Hence we have

$$\lambda_1 = q_1 + x_1 r_1 = \lambda_2 = q_2 + r_2 + c_p(t - t_0),$$

where t_0 is the saturation temperature corresponding to the pressure p_2 , and t is the temperature as recorded by the thermometer in the calorimeter.

Hence, using 0.48 for c_p and substituting the proper values for q and r from Table 3, we have

$$x_1 = \frac{q_2 + r_2 + c_p(t - t_0) - q_1}{r_1} = \frac{196.4 + 959.4 + 0.48(278 - 228) - 294.6}{890.5} = 0.994.$$

The temperature-entropy plot is shown in Figure 62, where are plotted to scale the corresponding values of the entropy as found from Table 3 for the constant steam weight line, and by calculation for the curve of constant heat contents equal to λ_1 .

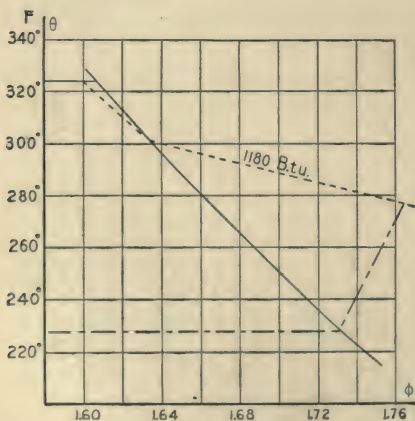


FIG. 62

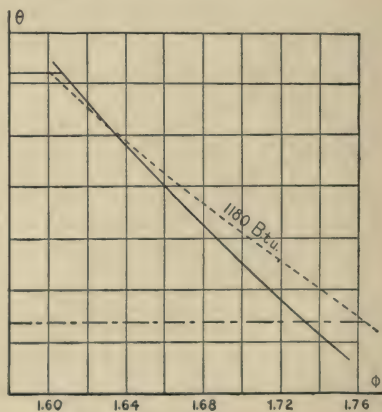


FIG. 63

To illustrate more fully the difference in appearance between the θ - ϕ and the p - ϕ diagrams discussed on page 96, there is shown in Figure 63 to the same scale as used above the p - ϕ plot for this process.

51. *Limits of throttling calorimeter.* If the valve d of the calorimeter described in Problem 50 is connected not to the atmosphere but to the exhaust pipe of the engine so that the back pressure is 3 pounds absolute, find the largest amount of priming (lowest value of x) that can be measured with the calorimeter for a boiler pressure of 85 pounds absolute. Neglect probable and observational errors in the thermometer.

Solution. At the back pressure of 3 pounds the steam must in the limiting case be just at the point of being superheated. Hence putting $c_p(t - t_x)$ equal to zero in the equation for x_1 derived in Problem 50 and substituting for q and r from Table 3, we have

$$x_1 = \frac{109.6 + 1012.2 - 286.5}{896.6} = 0.9315.$$

For this pressure the calorimeter can then be used to determine the steam quality down to 93.15 per cent or 6.85 per cent of priming. The range of the apparatus is evidently extended by connecting the outlet to the condenser and thus using a lower back pressure.

52. *Throttle control.* A small non-condensing engine uses steam at 50 pounds absolute, receiving it through a throttle valve from a header containing dry steam at 200 pounds absolute.¹ The atmospheric pressure is 14.7 pounds. Find the loss in availability of the steam in B.t.u. per pound due to throttling.

Solution. During throttling heat is converted into kinetic energy and then by friction into heat as in the calorimeter of Problem 50. The θ - ϕ diagram for throttling is similar to that of Problem 50.

If a quantity of heat $d\mathfrak{Q}$ is received from a source at a temperature of θ , and if the lowest available temperature is θ_0 , the Carnot efficiency between these two temperatures represents the fraction of $d\mathfrak{Q}$ which is available for useful work. This available portion of $d\mathfrak{Q}$, the heat received from the hot source, has been termed its "motivity" by Lord Kelvin.² Now the heat possessed by a mass of steam has been received at a series of temperatures ranging in the boiler from that of the feed water as it enters to its final temperature. At each of these temperatures the motivity or availability is given by an expression of the form $\frac{\theta - \theta_0}{\theta} d\mathfrak{Q}$. The availability of a given mass in any state, as 1, is then the sum of all the motivities due to the addition of the several infinitesimal amounts of heat $d\mathfrak{Q}$ at these various temperatures. That is, the motivity in state 1 is

$$\int_0 \frac{\theta - \theta_0}{\theta} d\mathfrak{Q} = \mathfrak{Q}_1 - \theta_0\phi_1.$$

Similarly, the motivity in any other state, as 2, is of the form $\mathfrak{Q}_2 - \theta_0\phi_2$. The difference in availability of the substance at these two states is then

$$\mathfrak{Q}_1 - \mathfrak{Q}_2 + \theta_0(\phi_2 - \phi_1),$$

¹ The conditions here chosen are not such as would ordinarily enter into good engineering practice. They are selected so that $\phi_2 - \phi_1$ is large as compared to the precision with which ϕ_1 or ϕ_2 is known.

² See Preston, "Heat," pp. 629, 630.

an expression which for the condition of constant heat contents required by this particular problem reduces to

$$\theta_0(\phi_2 - \phi_1).$$

The lowest temperature for this problem is that at which the steam is rejected, or 212° Fahrenheit. The absolute temperature θ_0 corresponding to this is 672. The entropy ϕ_1 is found directly from Table 3 to be 1.5459.

In order to find the entropy in state 2 it is necessary to make use of the condition that the heat contents remain constant and solve for the temperature as follows:

$$\lambda_1 = q_1 + r_1 = 354.3 + 843.5 = 1197.8,$$

$$\lambda_2 = q_2 + r_2 + c_p(t_2 - t_0) = 250.4 + 922.8 + c_p(t_2 - 281) = 1173.2 + h.$$

Therefore

$$h = 24.6; \quad t_2 - t_0 = 51.2. \quad \phi_2 = \frac{24.6}{460 + \frac{1}{2}(281 + 332)} = 0.0324.$$

The entropy in the saturated condition is found from the tables, hence the total entropy ϕ_2 is $1.6581 + 0.0324 = 1.6905$. The loss in availability per pound is then

$$672(1.6905 - 1.5459) = 97.2 \text{ B. t. u.}$$

53. Nozzle area. Plot to a scale of pressures for abscissas and ordinates of B. t. u. the energy $\lambda_1 - \lambda$ available for increasing the kinetic energy of a mixture of steam expanding adiabatically without friction from 150 pounds absolute and 75° Fahrenheit of superheat to a back pressure of 6 pounds absolute. Plot also the velocity during this expansion. Plot the area in square inches of a nozzle to discharge one pound per second.

Solution. From Table 3, for 150 pounds, $q_1 = 330.0$, $r_1 = 863.0$. For 75 degrees of superheat, use $c_p = 0.57$. Hence $h_1 = 43$. Therefore $\lambda_1 = 1236$. $\phi_{w_1} = 0.5142$, $\phi_v = 1.0551$, and $\phi_{s_1} = 0.0503$. Hence $\phi_1 = 1.6196$ or 1.62 approximately.

The values of the heat contents for other pressures may be found either by calculation or they may be read directly from the pressure-entropy chart of page 86. In this solution they are read directly from the chart,¹ and also the values of the specific volumes corresponding. The velocity is then found by equation (71) and the requisite area by equation (65), which may be written

$$\text{area in square inches} = \frac{(144)(\text{specific volume})}{\text{velocity}}.$$

The desired values are plotted in Figure 64.

54. Turbine nozzle. A 300 horse-power De Laval turbine uses 18 pounds of steam per horse-power hour. There are twelve nozzles supplying steam to the turbine wheel. The steam is at 150 pounds absolute and 75 degrees of superheat, and it expands to a back pressure of 2 pounds absolute. Find the diameters of the nozzle at the throat and at the exit end necessary to deliver the requisite amount

¹ For the purposes of this calculation the diagram proposed by Professor Mollier and named after him is most convenient. In it ordinates are λ 's and abscissas are ϕ 's. Lines of constant pressure and of quality are plotted. Such a diagram is contained in the Tables of Marks and Davis.

of steam: (a) for frictionless adiabatic expansion, (b) for 15 per cent energy loss by friction. For the throat pressure use $p_t = 0.58 p_1$.

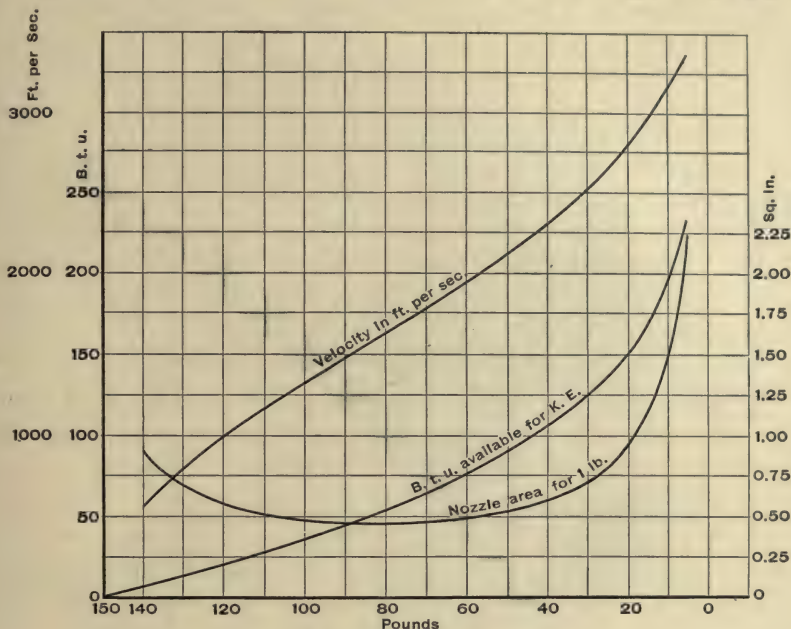


FIG. 64

Solution. (a) Each nozzle must deliver $\frac{(300)(18)}{(3600)(12)}$, or 0.125 pound per second.

The quality in the throat for adiabatic expansion is found as follows. The entropy is 1.6196 in the initial condition as found in Problem 53. For the throat pressure $(\phi_w)_t = 0.4617$ and $(\phi_v)_t = 1.1516$. Hence $(\phi_s)_t$ must equal 0.0063. Since the saturation temperature is 318° Fahrenheit, for the throat pressure of 87 pounds, the quality is 10 degrees of superheat (using $c_p = 0.48$).

By equation (57) the specific volume is

$$v_t = \frac{(85.85)(788)}{(87)(144)} - 0.256 = 5.13.$$

The total heat content at the throat is $(\lambda)_t = 288.2 + 895.3 + 4.8 = 1188.3$.

The total heat contents in the initial condition were found in Problem 53 to be $\lambda_1 = 1236$.

The velocity in the throat is then

$$v_t = \sqrt{2 g 778 (1236 - 1188.3)} = 1545 \text{ feet per second.}$$

The requisite area in the throat must then be

$$\frac{(144)(5.13)}{1545} = 0.479 \text{ square inch per pound of steam.}$$

But for the turbine nozzle of the problem only 0.125 pounds per second are to pass, hence the requisite area is $(0.125)(0.479) = 0.06$ square inches. The diameter at the throat is 0.276 inch.

Similar solution for the exit end gives $x_2 = 0.828$, $\lambda_2 = 940.0$, and the specific volume at the exit is 143.3 cubic feet. The velocity is therefore 3835 feet per second, the area of the turbine nozzle is 0.672 square inch, and the diameter is 0.924 inch.

(b) It is usual not to make allowance for the friction at the throat, since its effect would be small, but to make the nozzle at the throat according to the calculations for a frictionless adiabatic expansion as made above. It will be remembered in justification of this method that the steam discharged through an orifice is practically independent of the pressures, provided only that the exit pressure is less than 0.58 of the initial pressure. (Thus compare Problem 47.)

At the exit end the loss in energy is 15 per cent of $\lambda_1 - \lambda_2$ or 44 B. t. u. The quality at exit for a flow with friction is greater than that for a frictionless flow by an amount equal to the fraction of a pound of steam that this number of B. t. u. would evaporate. Since r_2 for 2 pounds is 1021.9, the increase in quality is $\frac{44}{1022} = 0.043$. The quality after flow with friction is then $0.828 + 0.043 = 0.871$. Hence the specific volume is greater and is $(0.871)(173.1) = 150.8$ cubic feet. In order that this turbine nozzle shall deliver the same amount of steam for the same pressure conditions the area at the exit end must be larger than that for the case of frictionless expansion in the ratio of the specific volumes in the two cases. Hence the requisite area is

$$\frac{150.8}{143.3} (0.672) = 0.706 \text{ square inch.}$$

And the diameter is 0.947 inch.

MISCELLANEOUS PROBLEMS

1. Determine four points on the pressure-volume curve and also four points on the temperature-entropy curve representing the transformation of a pound of dry air undergoing an increase of temperature from 100° to 200° Fahrenheit under the following conditions: (a) adiabatically from an initial pressure of 15 pounds, (b) at constant pressure of 15^1 pounds per square inch, (c) at constant volume of 11 cubic feet. Assume a zero of entropy for each initial condition.

2. Plot in the same manner as in Problem 1 the transformation at a constant temperature of one pound of air originally occupying 11 cubic feet at 35 pounds per square inch if the pressure decreases to 15 pounds.

3. Find the work done, the change in internal energy, the change in entropy, and the heat added during the expansion of 3.5 cubic feet of air from 200 to 90 pounds pressure. The temperature is maintained constant at 650° Fahrenheit absolute.

4. Find the external work done, the change in internal energy, the change in entropy, and the heat added if 3 cubic feet of air at 200° Fahrenheit expand adiabatically from a pressure of 85 pounds per square inch to a pressure of 15 pounds.

5. A quantity of gas occupying 9 cubic feet at a pressure of 42 pounds per square inch expands until the volume is 14 cubic feet at a pressure of 23 pounds. Find an equation representing a possible expansion between these two points. Find the work done. Find the change in intrinsic energy.

6. A steam engine working between 135 pounds and 2 pounds per square inch pressure requires 225 B. t. u. per indicated horse power per minute. What is the ratio between the thermal efficiency of this engine and that of a Carnot engine working between the same temperature limits?

7. Calculate the work done by an air compressor without clearance in compressing and delivering 10 cubic feet of dry air from an atmospheric pressure of 15 pounds to a gauge pressure of 75 pounds. Assume adiabatic compression.

¹ All pressures are absolute and not gauge unless otherwise stated.

8. What would have been the work in Problem 7 if the compression had been according to the relation $pV^{1.32} = \text{a constant}$?

9. If cooling water of temperature 65° Fahrenheit is available, to what pressure must dry atmospheric air be compressed in order that after cooling it may in an adiabatic expansion fall to 32° Fahrenheit? The initial temperature of the air is 70° Fahrenheit. Atmospheric pressure is 15 pounds per square inch.

10. Find from the entropy chart of page 86 what the pressure is upon a mixture of steam and water one pound of which occupies 6 cubic feet, if the quality is 95 per cent.

11. What is the quality of a mixture of steam and water of entropy 1.590 if one pound occupies 8 cubic feet?

12. What is the quality in Problem 11 if the mixture occupies 3 cubic feet?

13. One pound of saturated steam at a pressure of 140 pounds per square inch undergoes a constant heat content change until the final pressure is 25 pounds. Find the quality. Calculate the final intrinsic energy.

14. Steam at 140 pounds gauge and superheated 100° Fahrenheit expands adiabatically. What is the pressure when the steam is dry saturated steam? Atmospheric pressure is 15 pounds.

15. If the steam of Problem 14 expands in an engine cylinder to five times its original volume, what, approximately, are the final pressure and quality?

16. Steam of entropy 1.560 and heat contents 1200 B.t.u. expands adiabatically until 200 B.t.u. have been converted into external work. Find the final quality.

17. Find from the entropy chart of page 86 the limit of a throttling calorimeter when used to measure quality in steam of pressure 145 pounds if it is connected to the atmosphere of which the absolute pressure is 14.7 pounds.

18. Steam of quality 99 per cent and pressure 160 pounds passes through a reducing valve. To what pressure must it be throttled in order that the steam discharged shall be dry and saturated?

19. Steam from a boiler is passed into a barrel calorimeter in which there is 200 pounds of water at a temperature of 60° Fahrenheit. When 3 pounds have been condensed, the final temperature is 75° Fahrenheit. The boiler pressure is 145 pounds. Find the quality.

20. Feed water of temperature 80° Fahrenheit enters a boiler and passes from a boiler through a superheater. It emerges at 100 pounds pressure and a temperature of 600° Fahrenheit. If 8 pounds of water is thus converted into steam for each pound of coal burned, find the equivalent evaporation from and at 212° Fahrenheit.

21. Find from values taken from the entropy chart of page 86 the Rankine efficiency of an engine working between 140 pounds with a quality of 95 per cent and 10 pounds.

22. A four-stage impulse turbine is designed to develop equal velocity in each of the stages. If supplied with steam of 100 pounds pressure and 100° Fahrenheit superheat and expanding to a condenser pressure of 2 pounds, find the velocity developed in each stage.

23. What is the area at the throat of a nozzle which allows 50 pounds of steam to pass in a minute, if the entering steam is 150 pounds pressure and 100° Fahrenheit superheat, and the back pressure is less than 0.58 of the admission pressure?

24. What would be the area if the back pressure in Problem 23 were 100 pounds?

25. How much work is done in compressing adiabatically one pound of a mixture of steam and water of quality 80 per cent and pressure 14.7 pounds until the volume is 6 cubic feet?

26. Two pipes deliver into a third. One supplies 300 gallons per minute at a temperature of 80° Fahrenheit and the other supplies steam of quality 75 per cent and pressure 100 pounds per square inch at the rate of 5 pounds per minute. What is the resulting quality of the mixture in the third pipe?

27. A hot-water heater is constructed on the above principle (Problem 26). Ten gallons of water at 200° are to be obtained per minute in the third pipe. The entering water is at 60° Fahrenheit. The entering steam is at 50 pounds pressure and dry. What is the smallest amount of steam in pounds required per minute?

28. What approximately is the final temperature and pressure exerted by a mixture of steam and water in a closed tank if the tank originally contained 10 pounds at a pressure of 140 pounds per square inch and a quality of 98 per cent and 9000 B. t. u. are removed by cooling?

29. Find the increase in volume per pound caused by passing dry steam at a pressure of 130 pounds per square inch through a superheater which raises the temperature to 500° Fahrenheit.

30. A throttle valve reduces the pressure of dry steam from 135 pounds to 100 pounds. Find the loss in availability in B. t. u. per pound if the lowest available temperature is 141.5° Fahrenheit.

31. Plot to scale the indicator card for a theoretically perfect engine which has a stroke of 30 inches, a clearance of 5 per cent, cut-off at 30 per cent, compression at 5 per cent, release at the end of the stroke, a piston diameter of 16 inches, and makes 100 revolutions per minute. If the entering steam is 100 pounds pressure, the condenser pressure is 2 pounds, and the expansion and compression may be assumed to be adiabatic, find the work done during admission, expansion, exhaust, and compression. Find the indicated horse power. Find the mean effective pressure. The engine is of course double-acting, having valves at each end of the cylinder.

32. Find the horse power of a hot-air engine which works on a Sterling cycle from a maximum temperature of 700° Fahrenheit and a maximum pressure of 110 pounds absolute, if the volume of the working cylinder is 2 cubic feet and that of the displacement cylinder is 6.5 cubic feet. Neglect clearance. The lowest available temperature is 90° Fahrenheit. The engine makes 100 revolutions per minute.

33. Draw to scale the cross section of a nozzle which will discharge 0.50 pound of steam per second for frictionless adiabatic expansion from 120 pounds pressure and 120° Fahrenheit superheat to 2 pounds pressure per square inch.

34. If the friction loss in the nozzle of Problem 33 is to be taken as 12 per cent and the nozzle is drawn tapered from the throat to the exit end, how much larger must be the diameter of the exit end in order that the same amount of steam shall pass per second?

35. If the steam of Problem 33 had not been superheated but had been of quality 98 per cent, what would have been the velocity at the exit end of the nozzle?

36. If a steam turbine is to have four stages of equal velocity and to be supplied with steam under the conditions of Problem 33, what will be the pressures in the intermediate stages?

TABLES

TABLES

TABLE 1

GAS	EXPANSION COEFFICIENT α	PRESSURE COEFFICIENT β	OBSERVER	DATE
Hydrogen	0.0036600	0.0036626	P. Chappuis	1887
Hydrogen	0.003661	0.003668	Regnault	1840
Air	0.003670	0.003665	Regnault	1840
Air	————	0.003663	Kuenen and Randall	1895
Carbon dioxide (CO ₂) .	0.003710	0.003688	Regnault	1840
Nitrous oxide (N ₂ O) .	0.003719	0.003676	Regnault	1840
Oxygen	————	0.003674	von Jolly	1874
Nitrogen	————	0.003667	von Jolly	1874
Argon	————	0.003668	Kuenen and Randall	1895
Helium	————	0.0036627	Travers and Jaquered	1903

TABLE 2

SUBSTANCE	SPECIFIC VOLUME ¹ Cubic feet per pound	SPECIFIC HEAT Constant pressure	SPECIFIC HEAT Constant Volume	RATIO $k = \frac{c_p}{c_v}$
Hydrogen	178.2	3.409	2.406	1.408
Oxygen	12.21	0.2175	0.155	1.403
Air (dry)	12.39	0.2375	0.1689	1.401
Nitrogen	12.75	0.2438	0.173	1.409
Marsh gas (CH ₄) .	23.30	0.593	0.467	1.31
Carbon dioxide .	8.10	0.2025	0.171	1.30
Carbon monoxide .	12.80	0.2426	0.173	1.401
Ethylene (C ₂ H ₄) .	12.58	0.404	0.332	1.24

¹ For standard conditions of $p = 14.7$ pounds per square inch and $t = 32^\circ$ Fahrenheit.

TABLE 3¹

(F. P. S. Gravitational system; degrees Fahrenheit, British thermal units)

DRY SATURATED STEAM

TEMPERATURE, DEGREES FAHR.	PRESSURE, POUNDS PER SQUARE INCH	HEAT OF THE LIQUID	HEAT OF VAP- ORIZATION	HEAT CONTENTS	HEAT EQUIVALENT OF INTERNAL WORK	HEAT EQUIVALENT OF EXTERNAL WORK	SPECIFIC VOLUME	DENSITY, POUNDS PER CUBIC FOOT	ENTROPY OF THE LIQUID	ENTROPY OF VAPORIZATION	TOTAL ENTROPY
<i>t</i>	<i>p</i>	<i>q</i>	<i>r</i>	<i>λ</i>	<i>ρ</i>	<i>Apu</i>	<i>s</i>	$\gamma = \frac{1}{s}$	ϕ_w	ϕ_v	$\phi_w + \phi_v$
60	0.256	28.1	1057.0	1085.1	999.8	57.2	1207.	0.000828	0.0556	2.0347	2.0903
65	0.305	33.1	1054.4	1087.5	996.7	57.7	1021.	0.000979	0.0652	2.0103	2.0755
70	0.363	38.1	1051.8	1089.9	993.6	58.2	868.	0.001152	0.0747	1.9863	2.0610
75	0.429	43.1	1049.2	1092.3	990.5	58.7	739.	0.001353	0.0841	1.9629	2.0470
80	0.506	48.1	1046.5	1094.6	987.2	59.3	634.	0.001577	0.0934	1.9398	2.0332
85	0.594	53.1	1043.9	1097.0	984.1	59.8	544.	0.001838	0.1026	1.9171	2.0197
90	0.696	58.1	1041.2	1099.3	980.9	60.3	469.	0.002131	0.1117	1.8948	2.0065
100	0.946	68.0	1035.7	1103.7	974.4	61.3	351.0	0.002851	0.1297	1.8511	1.9808
101.8	1	69.8	1034.7	1104.5	973.1	61.6	333.0	0.00300	0.1329	1.8433	1.9702
110	1.271	78.0	1030.1	1108.1	967.7	62.4	265.0	0.003771	0.1473	1.8088	1.9561
120	1.689	88.0	1024.4	1112.4	961.0	63.4	203.0	0.004926	0.1647	1.7677	1.9324
126.1	2	94.2	1021.9	1116.1	957.8	64.1	173.0	0.005578	0.1753	1.7432	1.9185
141.5	3	109.6	1012.2	1121.8	946.4	65.8	118.4	0.00845	0.2011	1.6841	1.8852
153.0	4	121.0	1005.5	1126.5	938.6	66.9	90.4	0.01106	0.2200	1.6416	1.8616
162.3	5	130.3	1000.0	1130.3	932.1	67.9	73.3	0.01364	0.2351	1.6084	1.8435
170.1	6	138.1	995.5	1133.6	926.8	68.7	61.9	0.01616	0.2476	1.5812	1.8288
193.2	10	161.3	981.4	1142.7	910.4	71.0	38.37	0.02606	0.2838	1.5036	1.7874
212	14.7	180.3	969.7	1150.0	896.9	72.8	26.78	0.03734	0.3125	1.4441	1.7566
228.0	20	196.4	959.4	1155.8	885.1	74.3	20.09	0.04978	0.3362	1.3957	1.7319
250.3	30	219.1	944.4	1163.5	868.2	76.2	13.74	0.0728	0.3687	1.3305	1.6992
281.0	50	250.4	922.8	1173.2	844.1	78.7	8.507	0.1176	0.4119	1.2462	1.6581
303.0	70	272.9	906.6	1179.5	826.3	80.3	6.199	0.1613	0.4418	1.1892	1.6310
312.1	80	282.2	899.8	1182.0	818.9	80.9	5.466	0.1829	0.4540	1.1661	1.6201
316.3	85	286.5	896.6	1183.1	815.4	81.2	5.161	0.1938	0.4595	1.1557	1.6152
320.3	90	290.7	893.5	1184.2	812.1	81.4	4.886	0.2047	0.4649	1.1457	1.6106
324.2	95	294.6	890.5	1185.1	808.8	81.7	4.644	0.2153	0.4699	1.1363	1.6062
327.9	100	298.5	887.6	1186.1	805.7	81.9	4.432	0.2256	0.4748	1.1273	1.6021
331.4	105	302.1	884.8	1186.9	802.7	82.1	4.233	0.2362	0.4794	1.1187	1.5981
334.8	110	305.6	882.1	1187.7	799.7	82.4	4.047	0.2471	0.4838	1.1105	1.5943
338.1	115	309.0	879.5	1188.5	797.0	82.5	3.876	0.2580	0.4881	1.1026	1.5907
341.3	120	312.3	876.9	1189.2	794.2	82.7	3.723	0.2686	0.4922	1.0951	1.5873
344.4	125	315.5	874.5	1190.0	791.6	82.9	3.581	0.2793	0.4962	1.0878	1.5840
347.4	130	318.6	872.1	1190.7	789.0	83.1	3.451	0.2898	0.5000	1.0808	1.5808
350.3	135	321.5	869.8	1191.3	786.5	83.3	3.331	0.3002	0.5037	1.0741	1.5778
353.1	140	324.4	867.4	1191.8	784.0	83.4	3.220	0.3106	0.5073	1.0675	1.5748
355.8	145	327.3	865.2	1192.5	781.6	83.6	3.115	0.3210	0.5108	1.0612	1.5720
358.5	150	330.0	863.0	1193.0	779.3	83.7	3.014	0.3318	0.5142	1.0551	1.5693
363.6	160	335.3	858.8	1194.1	774.9	83.9	2.834	0.3528	0.5206	1.0434	1.5640
368.5	170	340.4	854.8	1195.2	770.6	84.2	2.673	0.3741	0.5268	1.0324	1.5592
373.2	180	345.2	850.9	1196.1	766.5	84.4	2.531	0.3951	0.5326	1.0219	1.5545
381.9	200	354.3	843.5	1197.8	758.8	84.7	2.288	0.4371	0.5434	1.0025	1.5450
401.1	250	374.2	826.9	1201.1	741.5	85.4	1.845	0.542	0.5669	0.9609	1.5278

¹ These values are taken from the 1909 edition of Professor C. H. Peabody's "Tables of the Properties of Steam," with the permission of the author and the publishers, Wiley and Sons.

LOGARITHMS

	0	1	2	3	4	5	6	7	8	9	1 2 3	4 5 6	7 8 9
10	0000	0043	0086	0128	0170	0212	0253	0294	0334	0374	4 8 12	17 21 25	29 33 37
11	0414	0453	0492	0531	0569	0607	0645	0682	0719	0755	4 8 11	15 19 23	26 30 34
12	0792	0828	0864	0899	0934	0969	1004	1038	1072	1106	3 7 10	14 17 21	24 28 31
13	1139	1173	1206	1239	1271	1303	1335	1367	1399	1430	3 6 10	13 16 19	23 26 29
14	1461	1492	1523	1553	1584	1614	1644	1673	1703	1732	3 6 9	12 15 18	21 24 27
15	1761	1790	1818	1847	1875	1903	1931	1959	1987	2014	3 6 8	11 14 17	20 22 25
16	2041	2068	2095	2122	2148	2175	2201	2227	2253	2279	3 5 8	11 13 16	18 21 24
17	2304	2330	2355	2380	2405	2430	2455	2480	2504	2529	2 5 7	10 12 15	17 20 22
18	2553	2577	2601	2625	2648	2672	2695	2718	2742	2765	2 5 7	9 12 14	16 19 21
19	2788	2810	2833	2856	2878	2900	2923	2945	2967	2989	2 4 7	9 11 13	16 18 20
20	3010	3032	3054	3075	3096	3118	3139	3160	3181	3201	2 4 6	8 11 13	15 17 19
21	3222	3243	3263	3284	3304	3324	3345	3365	3385	3404	2 4 6	8 10 12	14 16 18
22	3424	3444	3464	3483	3502	3522	3541	3560	3579	3598	2 4 6	8 10 12	14 15 17
23	3617	3636	3655	3674	3692	3711	3729	3747	3766	3784	2 4 6	7 9 11	13 15 17
24	3802	3820	3838	3856	3874	3892	3909	3927	3945	3962	2 4 5	7 9 11	12 14 16
25	3979	3997	4014	4031	4048	4065	4082	4099	4116	4133	2 3 5	7 9 10	12 14 15
26	4150	4166	4183	4200	4216	4232	4249	4265	4281	4298	2 3 5	7 8 10	11 13 15
27	4314	4330	4346	4362	4378	4393	4409	4425	4440	4456	2 3 5	6 8 9	11 13 14
28	4472	4487	4502	4518	4533	4548	4564	4579	4594	4609	2 3 5	6 8 9	11 12 14
29	4624	4639	4654	4669	4683	4698	4713	4728	4742	4757	1 3 4	6 7 9	10 12 13
30	4771	4786	4800	4814	4829	4843	4857	4871	4886	4900	1 3 4	6 7 9	10 11 13
31	4914	4928	4942	4955	4969	4983	4997	5011	5024	5038	1 3 4	6 7 8	10 11 12
32	5051	5065	5079	5092	5105	5119	5132	5145	5159	5172	1 3 4	5 7 8	9 11 12
33	5185	5198	5211	5224	5237	5250	5263	5276	5289	5302	1 3 4	5 6 8	9 10 12
34	5315	5328	5340	5353	5366	5378	5391	5403	5416	5428	1 3 4	5 6 8	9 10 11
35	5441	5453	5465	5478	5490	5502	5514	5527	5539	5551	1 2 4	5 6 7	9 10 11
36	5563	5575	5587	5599	5611	5623	5635	5647	5658	5670	1 2 4	5 6 7	8 10 11
37	5682	5694	5705	5717	5729	5740	5752	5763	5775	5786	1 2 3	5 6 7	8 9 10
38	5798	5809	5821	5832	5843	5855	5866	5877	5888	5899	1 2 3	5 6 7	8 9 10
39	5911	5922	5933	5944	5955	5966	5977	5988	5999	6010	1 2 3	4 5 7	8 9 10
40	6021	6031	6042	6053	6064	6075	6085	6096	6107	6117	1 2 3	4 5 6	8 9 10
41	6128	6138	6149	6160	6170	6180	6191	6201	6212	6222	1 2 3	4 5 6	7 8 9
42	6232	6243	6253	6263	6274	6284	6294	6304	6314	6325	1 2 3	4 5 6	7 8 9
43	6335	6345	6355	6365	6375	6385	6395	6405	6415	6425	1 2 3	4 5 6	7 8 9
44	6435	6444	6454	6464	6474	6484	6493	6503	6513	6522	1 2 3	4 5 6	7 8 9
45	6532	6542	6551	6561	6571	6580	6590	6599	6609	6618	1 2 3	4 5 6	7 8 9
46	6628	6637	6646	6656	6665	6675	6684	6693	6702	6712	1 2 3	4 5 6	7 7 8
47	6721	6730	6739	6749	6758	6767	6776	6785	6794	6803	1 2 3	4 5 5	6 7 8
48	6812	6821	6830	6839	6848	6857	6866	6875	6884	6893	1 2 3	4 4 5	6 7 8
49	6902	6911	6920	6928	6937	6946	6955	6964	6972	6981	1 2 3	4 4 5	6 7 8
50	6990	6998	7007	7016	7024	7033	7042	7050	7059	7067	1 2 3	3 4 5	6 7 8
51	7076	7084	7093	7101	7110	7118	7126	7135	7143	7152	1 2 3	3 4 5	6 7 8
52	7160	7168	7177	7185	7193	7202	7210	7218	7226	7235	1 2 2	3 4 5	6 7 7
53	7243	7251	7259	7267	7275	7284	7292	7300	7308	7316	1 2 2	3 4 5	6 6 7
54	7324	7332	7340	7348	7356	7364	7372	7380	7388	7396	1 2 2	3 4 5	6 6 7

LOGARITHMS

	0	1	2	3	4	5	6	7	8	9	1 2 3	4 5 6	7 8 9
55	7404	7412	7419	7427	7435	7443	7451	7459	7466	7474	1 2 2	3 4 5	5 6 7
56	7482	7490	7497	7505	7513	7520	7528	7536	7543	7551	1 2 2	3 4 5	5 6 7
57	7559	7566	7574	7582	7589	7597	7604	7612	7619	7627	1 2 2	3 4 5	5 6 7
58	7634	7642	7649	7657	7664	7672	7679	7686	7694	7701	1 1 2	3 4 4	5 6 7
59	7709	7716	7723	7731	7738	7745	7752	7760	7767	7774	1 1 2	3 4 4	5 6 7
60	7782	7789	7796	7803	7810	7818	7825	7832	7839	7846	1 1 2	3 4 4	5 6 6
61	7853	7860	7868	7875	7882	7889	7896	7903	7910	7917	1 1 2	3 4 4	5 6 6
62	7924	7931	7938	7945	7952	7959	7966	7973	7980	7987	1 1 2	3 3 4	5 6 6
63	7993	8000	8007	8014	8021	8028	8035	8041	8048	8055	1 1 2	3 3 4	5 5 6
64	8062	8069	8075	8082	8089	8096	8102	8109	8116	8122	1 1 2	3 3 4	5 5 6
65	8129	8136	8142	8149	8156	8162	8169	8176	8182	8189	1 1 2	3 3 4	5 5 6
66	8195	8202	8209	8215	8222	8228	8235	8241	8248	8254	1 1 2	3 3 4	5 5 6
67	8261	8267	8274	8280	8287	8293	8299	8306	8312	8319	1 1 2	3 3 4	5 5 6
68	8325	8331	8338	8344	8351	8357	8363	8370	8376	8382	1 1 2	3 3 4	4 5 6
69	8388	8395	8401	8407	8414	8420	8426	8432	8439	8445	1 1 2	2 3 4	4 5 6
70	8451	8457	8463	8470	8476	8482	8488	8494	8500	8506	1 1 2	2 3 4	4 5 6
71	8513	8519	8525	8531	8537	8543	8549	8555	8561	8567	1 1 2	2 3 4	4 5 5
72	8573	8579	8585	8591	8597	8603	8609	8615	8621	8627	1 1 2	2 3 4	4 5 5
73	8633	8639	8645	8651	8657	8663	8669	8675	8681	8686	1 1 2	2 3 4	4 5 5
74	8692	8698	8704	8710	8716	8722	8727	8733	8739	8745	1 1 2	2 3 4	4 5 5
75	8751	8756	8762	8768	8774	8779	8785	8791	8797	8802	1 1 2	2 3 3	4 5 5
76	8808	8814	8820	8825	8831	8837	8842	8848	8854	8859	1 1 2	2 3 3	4 5 5
77	8865	8871	8876	8882	8887	8893	8899	8904	8910	8915	1 1 2	2 3 3	4 4 5
78	8921	8927	8932	8938	8943	8949	8954	8960	8965	8971	1 1 2	2 3 3	4 4 5
79	8976	8982	8987	8993	8998	9004	9009	9015	9020	9025	1 1 2	2 3 3	4 4 5
80	9031	9036	9042	9047	9053	9058	9063	9069	9074	9079	1 1 2	2 3 3	4 4 5
81	9085	9090	9096	9101	9106	9112	9117	9122	9128	9133	1 1 2	2 3 3	4 4 5
82	9138	9143	9149	9154	9159	9165	9170	9175	9180	9186	1 1 2	2 3 3	4 4 5
83	9191	9196	9201	9206	9212	9217	9222	9227	9232	9238	1 1 2	2 3 3	4 4 5
84	9243	9248	9253	9258	9263	9269	9274	9279	9284	9289	1 1 2	2 3 3	4 4 5
85	9294	9299	9304	9309	9315	9320	9325	9330	9335	9340	1 1 2	2 3 3	4 4 5
86	9345	9350	9355	9360	9365	9370	9375	9380	9385	9390	1 1 2	2 3 3	4 4 5
87	9395	9400	9405	9410	9415	9420	9425	9430	9435	9440	0 1 1	2 2 3	3 4 4
88	9445	9450	9455	9460	9465	9469	9474	9479	9484	9489	0 1 1	2 2 3	3 4 4
89	9494	9499	9504	9509	9513	9518	9523	9528	9533	9538	0 1 1	2 2 3	3 4 4
90	9542	9547	9552	9557	9562	9566	9571	9576	9581	9586	0 1 1	2 2 3	3 4 4
91	9590	9595	9600	9605	9609	9614	9619	9624	9628	9633	0 1 1	2 2 3	3 4 4
92	9638	9643	9647	9652	9657	9661	9666	9671	9675	9680	0 1 1	2 2 3	3 4 4
93	9685	9689	9694	9699	9703	9708	9713	9717	9722	9727	0 1 1	2 2 3	3 4 4
94	9731	9736	9741	9745	9750	9754	9759	9763	9768	9773	0 1 1	2 2 3	3 4 4
95	9777	9782	9786	9791	9795	9800	9805	9809	9814	9818	0 1 1	2 2 3	3 4 4
96	9823	9827	9832	9836	9841	9845	9850	9854	9859	9863	0 1 1	2 2 3	3 4 4
97	9868	9872	9877	9881	9886	9890	9894	9899	9903	9908	0 1 1	2 2 3	3 4 4
98	9912	9917	9921	9926	9930	9934	9939	9943	9948	9952	0 1 1	2 2 3	3 4 4
99	9956	9961	9965	9969	9974	9978	9983	9987	9991	9996	0 1 1	2 2 3	3 3 4



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